Solutions Manual for Thermodynamics: An Engineering Approach Seventh Edition Yunus A. Cengel, Michael A. Boles McGraw-Hill, 2011

Chapter 5 MASS AND ENERGY ANALYSIS OF CONTROL VOLUMES

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Conservation of Mass

5-1C Mass flow rate is the amount of mass flowing through a cross-section per unit time whereas the volume flow rate is the amount of volume flowing through a cross-section per unit time.

5-2C Flow through a control volume is steady when it involves no changes with time at any specified position.

5-3C The amount of mass or energy entering a control volume does not have to be equal to the amount of mass or energy leaving during an unsteady-flow process.

5-4C No, a flow with the same volume flow rate at the inlet and the exit is not necessarily steady (unless the density is constant). To be steady, the mass flow rate through the device must remain constant.

5-5E A pneumatic accumulator arranged to maintain a constant pressure as air enters or leaves is considered. The amount of air added is to be determined.

Assumptions 1 Air is an ideal gas.

Properties The gas constant of air is R = 0.3704 psia·ft³/lbm·R (Table A-1E).

Analysis At the beginning of the filling, the mass of the air in the container is

$$m_1 = \frac{P_1 V_1}{RT_1} = \frac{(200 \text{ psia})(0.2 \text{ ft}^3)}{(0.3704 \text{ psia} \cdot \text{ft}^3/\text{lbm} \cdot \text{R})(80 + 460 \text{ R})} = 0.200 \text{ lbm}$$

During the process both pressure and temperature remain constant while volume increases by 5 times. Thus,

$$m_2 = \frac{P_2 V_2}{RT_2} = 5m_1 = 5(0.200) = 1.00 \text{ lbm}$$

The amount of air added to the container is then

$$\Delta m = m_2 - m_1 = 1.00 - 0.200 = 0.8 \text{ lbm}$$

5-6E Helium at a specified state is compressed to another specified state. The mass flow rate and the inlet area are to be determined.

Assumptions Flow through the compressor is steady.

Properties The gas cosntant of helium is $R = 2.6809 \text{ psia} \cdot \text{ft}^3/\text{lbm} \cdot \text{R}$ (Table A-1E)

Analysis The mass flow rate is determined from

$$\dot{m} = \frac{A_2 V_2}{v_2} = \frac{A_2 V_2 P_2}{RT_2} = \frac{(0.01 \,\text{ft}^2)(100 \,\text{ft/s})(200 \,\text{psia})}{(2.6809 \,\text{psia} \cdot \text{ft}^3/\text{lbm} \cdot \text{R})(1060 \,\text{R})} = 0.07038 \,\text{lbm/s}$$

The inlet area is determined from

$$A_{1} = \frac{\dot{m}\boldsymbol{v}_{1}}{V_{1}} = \frac{\dot{m}RT_{1}}{V_{1}P_{1}} = \frac{(0.07038 \,\text{lbm/s})(2.6809 \,\text{psia} \cdot \text{ft}^{3}/\text{lbm} \cdot \text{R})(530 \,\text{R})}{(50 \,\text{ft/s})(15 \,\text{psia})} = \mathbf{0.1333 \,\text{ft}^{2}}$$

200 psia 600°F 0.01 ft² Compressor 15 psia 70°F 50 ft/s

5-7 Air is accelerated in a nozzle. The mass flow rate and the exit area of the nozzle are to be determined.

Assumptions Flow through the nozzle is steady.

Properties The density of air is given to be 2.21 kg/m³ at the inlet, and 0.762 kg/m^3 at the exit.

Analysis (a) The mass flow rate of air is determined from the inlet conditions to be

$$\dot{m} = \rho_1 A_1 V_1 = (2.21 \text{ kg/m}^3)(0.009 \text{ m}^2)(40 \text{ m/s}) = 0.796 \text{ kg/s}$$

(b) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. Then the exit area of the nozzle is determined to be

$$\dot{m} = \rho_2 A_2 V_2 \longrightarrow A_2 = \frac{\dot{m}}{\rho_2 V_2} = \frac{0.796 \text{ kg/s}}{(0.762 \text{ kg/m}^3)(180 \text{ m/s})} = 0.0058 \text{ m}^2 = 58 \text{ cm}^2$$





5-8 Water flows through the tubes of a boiler. The velocity and volume flow rate of the water at the inlet are to be determined.

Assumptions Flow through the boiler is steady.

Properties The specific volumes of water at the inlet and exit are (Tables A-6 and A-7)

$$\begin{array}{c} P_{1} = 7 \text{ MPa} \\ T_{1} = 65^{\circ}\text{C} \end{array} \end{array} v_{1} = 0.001017 \text{ m}^{3}/\text{kg} \\ P_{2} = 6 \text{ MPa} \\ T_{2} = 450^{\circ}\text{C} \end{array} \end{aligned} v_{2} = 0.05217 \text{ m}^{3}/\text{kg} \qquad \begin{array}{c} 7 \text{ MPa} \\ 65^{\circ}\text{C} \end{array} \xrightarrow{\text{Steam}} 6 \text{ MPa}, 450^{\circ}\text{C} \\ 80 \text{ m/s} \end{array}$$

Analysis The cross-sectional area of the tube is

$$A_c = \frac{\pi D^2}{4} = \frac{\pi (0.13 \text{ m})^2}{4} = 0.01327 \text{ m}^2$$

The mass flow rate through the tube is same at the inlet and exit. It may be determined from exit data to be

$$\dot{m} = \frac{A_c V_2}{v_2} = \frac{(0.01327 \text{ m}^2)(80 \text{ m/s})}{0.05217 \text{ m}^3/\text{kg}} = 20.35 \text{ kg/s}$$

The water velocity at the inlet is then

$$V_1 = \frac{\dot{m}v_1}{A_c} = \frac{(20.35 \text{ kg/s})(0.001017 \text{ m}^3/\text{kg})}{0.01327 \text{ m}^2} = 1.560 \text{ m/s}$$

The volumetric flow rate at the inlet is

$$\dot{V}_1 = A_c V_1 = (0.01327 \text{ m}^2)(1.560 \text{ m/s}) = 0.0207 \text{ m}^3/\text{s}$$

5-9 Air is expanded and is accelerated as it is heated by a hair dryer of constant diameter. The percent increase in the velocity of air as it flows through the drier is to be determined.

Assumptions Flow through the nozzle is steady.

Properties The density of air is given to be 1.20 kg/m^3 at the inlet, and 1.05 kg/m^3 at the exit.

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. Then,

$$\dot{m}_1 = \dot{m}_2$$

$$\rho_1 A V_1 = \rho_2 A V_2$$

$$\frac{V_2}{V_1} = \frac{\rho_1}{\rho_2} = \frac{1.20 \text{ kg/m}^3}{0.95 \text{ kg/m}^3} = 1.263 \quad \text{(or, and increase of 26.3%)}$$





5-10 A rigid tank initially contains air at atmospheric conditions. The tank is connected to a supply line, and air is allowed to enter the tank until the density rises to a specified level. The mass of air that entered the tank is to be determined.

Properties The density of air is given to be 1.18 kg/m^3 at the beginning, and 7.20 kg/m^3 at the end.

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. The mass balance for this system can be expressed as

Mass balance:

$$m_{in} - m_{out} = \Delta m_{system} \rightarrow m_i = m_2 - m_1 = \rho_2 V - \rho_1 V$$

Substituting,

$$m_i = (\rho_2 - \rho_1) \mathbf{V} = [(7.20 - 1.18) \text{ kg/m}^3](1 \text{ m}^3) = 6.02 \text{ kg}$$

Therefore, 6.02 kg of mass entered the tank.



5-11 A cyclone separator is used to remove fine solid particles that are suspended in a gas stream. The mass flow rates at the two outlets and the amount of fly ash collected per year are to be determined.

Assumptions Flow through the separator is steady.

Analysis Since the ash particles cannot be converted into the gas and vice-versa, the mass flow rate of ash into the control volume must equal that going out, and the mass flow rate of flue gas into the control volume must equal that going out. Hence, the mass flow rate of ash leaving is

$$\dot{m}_{ash} = y_{ash}\dot{m}_{in} = (0.001)(10 \text{ kg/s}) = 0.01 \text{ kg/s}$$

The mass flow rate of flue gas leaving the separator is then

$$\dot{m}_{\rm flue\,gas} = \dot{m}_{\rm in} - \dot{m}_{\rm ash} = 10 - 0.01 = 9.99 \, \text{kg/s}$$

The amount of fly ash collected per year is

 $m_{\rm ash} = \dot{m}_{\rm ash} \Delta t = (0.01 \, \text{kg/s})(365 \times 24 \times 3600 \, \text{s/year}) = 315,400 \, \text{kg/year}$

5-12 Air flows through an aircraft engine. The volume flow rate at the inlet and the mass flow rate at the exit are to be determined.

Assumptions 1 Air is an ideal gas. 2 The flow is steady.

Properties The gas constant of air is R = 0.287 kPa·m³/kg·K (Table A-1).

Analysis The inlet volume flow rate is

$$\dot{V}_1 = A_1 V_1 = (1 \text{ m}^2)(180 \text{ m/s}) = 180 \text{ m}^3/\text{s}$$

The specific volume at the inlet is

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$$v_1 = \frac{RT_1}{P_1} = \frac{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(20 + 273 \text{ K})}{100 \text{ kPa}} = 0.8409 \text{ m}^3/\text{kg}$$

Since the flow is steady, the mass flow rate remains constant during the flow. Then,

$$\dot{m} = \frac{\dot{\nu}_1}{\nu_1} = \frac{180 \text{ m}^3/\text{s}}{0.8409 \text{ m}^3/\text{kg}} = 214.1 \text{ kg/s}$$

5-13 A spherical hot-air balloon is considered. The time it takes to inflate the balloon is to be determined.

Assumptions 1 Air is an ideal gas.

Properties The gas constant of air is $R = 0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K}$ (Table A-1).

Analysis The specific volume of air entering the balloon is

$$\mathbf{v} = \frac{RT}{P} = \frac{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(20 + 273 \text{ K})}{120 \text{ kPa}} = 0.7008 \text{ m}^3/\text{kg}$$

The mass flow rate at this entrance is

$$\dot{m} = \frac{A_c V}{v} = \frac{\pi D^2}{4} \frac{V}{v} = \frac{\pi (1.0 \text{ m})^2}{4} \frac{3 \text{ m/s}}{0.7008 \text{ m}^3/\text{kg}} = 3.362 \text{ kg/s}$$

The initial mass of the air in the balloon is

$$m_i = \frac{V_i}{v} = \frac{\pi D^3}{6v} = \frac{\pi (5 \text{ m})^3}{6(0.7008 \text{ m}^3/\text{kg})} = 93.39 \text{ kg}$$

Similarly, the final mass of air in the balloon is

$$m_f = \frac{V_f}{v} = \frac{\pi D^3}{6v} = \frac{\pi (15 \text{ m})^3}{6(0.7008 \text{ m}^3/\text{kg})} = 2522 \text{ kg}$$

The time it takes to inflate the balloon is determined from

$$\Delta t = \frac{m_f - m_i}{\dot{m}} = \frac{(2522 - 93.39) \text{ kg}}{3.362 \text{ kg/s}} = 722 \text{ s} = 12.0 \text{ min}$$

5-14 A water pump increases water pressure. The diameters of the inlet and exit openings are given. The velocity of the water at the inlet and outlet are to be determined.

Assumptions 1 Flow through the pump is steady. 2 The specific volume remains constant.

Properties The inlet state of water is compressed liquid. We approximate it as a saturated liquid at the given temperature. Then, at 15°C and 40°C, we have (Table A-4)

Analysis The velocity of the water at the inlet is

$$V_1 = \frac{\dot{m} v_1}{A_1} = \frac{4 \dot{m} v_1}{\pi D_1^2} = \frac{4(0.5 \text{ kg/s})(0.001001 \text{ m}^3/\text{kg})}{\pi (0.01 \text{ m})^2} = 6.37 \text{ m/s}$$

Since the mass flow rate and the specific volume remains constant, the velocity at the pump exit is

$$V_2 = V_1 \frac{A_1}{A_2} = V_1 \left(\frac{D_1}{D_2}\right)^2 = (6.37 \text{ m/s}) \left(\frac{0.01 \text{ m}}{0.015 \text{ m}}\right)^2 = 2.83 \text{ m/s}$$

Using the specific volume at 40°C, the water velocity at the inlet becomes

$$V_1 = \frac{\dot{m}\boldsymbol{v}_1}{A_1} = \frac{4\dot{m}\boldsymbol{v}_1}{\pi D_1^2} = \frac{4(0.5 \text{ kg/s})(0.001008 \text{ m}^3/\text{kg})}{\pi (0.01 \text{ m})^2} = \mathbf{6.42 \text{ m/s}}$$

which is a 0.8% increase in velocity.



5-15 Refrigerant-134a flows through a pipe. Heat is supplied to R-134a. The volume flow rates of air at the inlet and exit, the mass flow rate, and the velocity at the exit are to be determined.

Properties The specific volumes of R-134a at the inlet and exit are (Table A-13)

$$P_{1} = 200 \text{ kPa} T_{1} = 20^{\circ}\text{C}$$
$$v_{1} = 0.1142 \text{ m}^{3}/\text{kg}$$
$$P_{1} = 180 \text{ kPa} T_{1} = 40^{\circ}\text{C}$$
$$v_{2} = 0.1374 \text{ m}^{3}/\text{kg}$$

Analysis



(a) (b) The volume flow rate at the inlet and the mass flow rate are

$$\dot{V}_{1} = A_{c}V_{1} = \frac{\pi D^{2}}{4}V_{1} = \frac{\pi (0.28 \text{ m})^{2}}{4}(5 \text{ m/s}) = 0.3079 \text{ m}^{3}/\text{s}$$
$$\dot{m} = \frac{1}{\nu_{1}}A_{c}V_{1} = \frac{1}{\nu_{1}}\frac{\pi D^{2}}{4}V_{1} = \frac{1}{0.1142 \text{ m}^{3}/\text{kg}}\frac{\pi (0.28 \text{ m})^{2}}{4}(5 \text{ m/s}) = 2.696 \text{ kg/s}$$

(c) Noting that mass flow rate is constant, the volume flow rate and the velocity at the exit of the pipe are determined from

$$\dot{\mathbf{V}}_2 = \dot{m}\mathbf{v}_2 = (2.696 \text{ kg/s})(0.1374 \text{ m}^3/\text{kg}) = 0.3705 \text{ m}^3/\text{s}$$

 $V_2 = \frac{\dot{\mathbf{V}}_2}{A_c} = \frac{0.3705 \text{ m}^3/\text{s}}{\frac{\pi (0.28 \text{ m})^2}{4}} = 6.02 \text{ m/s}$

5-16 A smoking lounge that can accommodate 15 smokers is considered. The required minimum flow rate of air that needs to be supplied to the lounge and the diameter of the duct are to be determined.

Assumptions Infiltration of air into the smoking lounge is negligible.

Properties The minimum fresh air requirements for a smoking lounge is given to be 30 L/s per person.

Analysis The required minimum flow rate of air that needs to be supplied to the lounge is determined directly from

$$\dot{\boldsymbol{V}}_{air} = \dot{\boldsymbol{V}}_{air \text{ per person}} (\text{No. of persons})$$

$$= (30 \text{ L/s} \cdot \text{person})(15 \text{ persons}) = 450 \text{ L/s} = 0.45 \text{ m}^3/\text{s}$$

The volume flow rate of fresh air can be expressed as

$$\dot{\boldsymbol{V}} = VA = V(\pi D^2 / 4)$$

Solving for the diameter D and substituting,

$$D = \sqrt{\frac{4\dot{\nu}}{\pi V}} = \sqrt{\frac{4(0.45 \text{ m}^3/\text{s})}{\pi (8 \text{ m/s})}} = 0.268 \text{ m}$$

Therefore, the diameter of the fresh air duct should be at least 26.8 cm if the velocity of air is not to exceed 8 m/s.

Smoking Lounge 15 smokers 5-17 The minimum fresh air requirements of a residential building is specified to be 0.35 air changes per hour. The size of the fan that needs to be installed and the diameter of the duct are to be determined.

Analysis The volume of the building and the required minimum volume flow rate of fresh air are

2

$$V_{\text{room}} = (3.0 \text{ m})(200 \text{ m}^2) = 600 \text{ m}^3$$

$$\dot{V} = V_{\text{room}} \times \text{ACH} = (600 \text{ m}^3)(0.35/\text{h}) = 210 \text{ m}^3 / \text{h} = 210,000 \text{ L/h} = 3500 \text{ L/min}$$

The volume flow rate of fresh air can be expressed as

$$\dot{V} = VA = V(\pi D^2 / 4)$$

Solving for the diameter *D* and substituting,
0.35 ACH 200 m²

$$D = \sqrt{\frac{4\dot{\nu}}{\pi V}} = \sqrt{\frac{4(210/3600 \text{ m}^3/\text{s})}{\pi (4 \text{ m/s})}} = 0.136 \text{ m}$$

1

Therefore, the diameter of the fresh air duct should be at least 13.6 cm if the velocity of air is not to exceed 4 m/s.

Flow Work and Energy Transfer by Mass

5-18C Energy can be transferred to or from a control volume as heat, various forms of work, and by mass.

5-19C Flow energy or flow work is the energy needed to push a fluid into or out of a control volume. Fluids at rest do not possess any flow energy.

5-20C Flowing fluids possess flow energy in addition to the forms of energy a fluid at rest possesses. The total energy of a fluid at rest consists of internal, kinetic, and potential energies. The total energy of a flowing fluid consists of internal, kinetic, potential, and flow energies.

5-21E A water pump increases water pressure. The flow work required by the pump is to be determined.

Assumptions 1 Flow through the pump is steady. 2 The state of water at the pump inlet is saturated liquid. 3 The specific volume remains constant.

Properties The specific volume of saturated liquid water at 10 psia is

$$v = v_{f@10 \text{ psia}} = 0.01659 \text{ ft}^3/\text{lbm}$$
 (Table A-5E)

Then the flow work relation gives

$$w_{\text{flow}} = P_2 v_2 - P_1 v_1 = v(P_2 - P_1)$$

= (0.01659 ft³/lbm)(50-10)psia $\left(\frac{1 \text{Btu}}{5.404 \text{ psia} \cdot \text{ft}^3}\right)$
= **0.1228 Btu/lbm**



5-22 An air compressor compresses air. The flow work required by the compressor is to be determined.

Assumptions 1 Flow through the compressor is steady. 2 Air is an ideal gas.

Properties The gas constant of air is $R = 0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K}$ (Table A-1).

Analysis Combining the flow work expression with the ideal gas equation of state gives

$$w_{\text{flow}} = P_2 v_2 - P_1 v_1$$

= $R(T_2 - T_1)$
= (0.287 kJ/kg · K)(400 - 20)K
= **109 kJ/kg**



5-23E Steam is leaving a pressure cooker at a specified pressure. The velocity, flow rate, the total and flow energies, and the rate of energy transfer by mass are to be determined.

Assumptions **1** The flow is steady, and the initial start-up period is disregarded. **2** The kinetic and potential energies are negligible, and thus they are not considered. **3** Saturation conditions exist within the cooker at all times so that steam leaves the cooker as a saturated vapor at 20 psia.

Properties The properties of saturated liquid water and water vapor at 20 psia are $v_f = 0.01683$ ft³/lbm, $v_g = 20.093$ ft³/lbm, $u_g = 1081.8$ Btu/lbm, and $h_g = 1156.2$ Btu/lbm (Table A-5E).

Analysis (*a*) Saturation conditions exist in a pressure cooker at all times after the steady operating conditions are established. Therefore, the liquid has the properties of saturated liquid and the exiting steam has the properties of saturated vapor at the operating pressure. The amount of liquid that has evaporated, the mass flow rate of the exiting steam, and the exit velocity are

$$m = \frac{\Delta \mathbf{V}_{\text{liquid}}}{\mathbf{v}_f} = \frac{0.6 \text{ gal}}{0.01683 \text{ ft}^3/\text{lbm}} \left(\frac{0.13368 \text{ ft}^3}{1 \text{ gal}}\right) = 4.766 \text{ lbm}$$

$$\dot{m} = \frac{m}{\Delta t} = \frac{4.766 \text{ lbm}}{45 \text{ min}} = 0.1059 \text{ lbm/min} = \mathbf{1.765} \times \mathbf{10^{-3}} \text{ lbm/s}$$

$$V = \frac{\dot{m}}{\rho_g A_c} = \frac{\dot{m} \mathbf{v}_g}{A_c} = \frac{(1.765 \times 10^{-3} \text{ lbm/s})(20.093 \text{ ft}^3/\text{lbm})}{0.15 \text{ in}^2} \left(\frac{144 \text{ in}^2}{1 \text{ ft}^2}\right) = \mathbf{34.1 \text{ ft/s}}$$

(b) Noting that h = u + Pv and that the kinetic and potential energies are disregarded, the flow and total energies of the exiting steam are

$$e_{\text{flow}} = P \mathbf{v} = h - u = 1156.2 - 1081.8 = 74.4$$
 Btu/lbm
 $\theta = h + ke + pe \cong h = 1156.2$ Btu/lbm

Note that the kinetic energy in this case is $ke = V^2/2 = (34.1 \text{ ft/s})^2/2 = 581 \text{ ft}^2/\text{s}^2 = 0.0232 \text{ Btu/lbm}$, which is very small compared to enthalpy.

(c) The rate at which energy is leaving the cooker by mass is simply the product of the mass flow rate and the total energy of the exiting steam per unit mass,

$$\dot{E}_{\text{mass}} = \dot{m}\theta = (1.765 \times 10^{-3} \text{ lbm/s})(1156.2 \text{ Btu/lbm}) = 2.04 \text{ Btu/s}$$

Discussion The numerical value of the energy leaving the cooker with steam alone does not mean much since this value depends on the reference point selected for enthalpy (it could even be negative). The significant quantity is the difference between the enthalpies of the exiting vapor and the liquid inside (which is h_{fg}) since it relates directly to the amount of energy supplied to the cooker.

Properties The properties of air are R = 0.287 kJ/kg.K and $c_p = 1.008$ kJ/kg.K (at 350 K from Table A-2b)

Analysis (a) The diameter is determined as follows

$$v = \frac{RT}{P} = \frac{(0.287 \text{ kJ/kg.K})(77 + 273 \text{ K})}{(300 \text{ kPa})} = 0.3349 \text{ m}^3/\text{kg}$$

$$A = \frac{\dot{m}v}{V} = \frac{(18/60 \text{ kg/s})(0.3349 \text{ m}^3/\text{kg})}{25 \text{ m/s}} = 0.004018 \text{ m}^2$$
$$= \sqrt{44} \sqrt{4(0.004018 \text{ m}^2)} = 0.004018 \text{ m}^2$$

 $D = \sqrt{\frac{4A}{\pi}} = \sqrt{\frac{4(0.004018 \,\mathrm{m}^2)}{\pi}} = 0.0715 \,\mathrm{m}$

(b) The rate of flow energy is determined from

$$\dot{W}_{\text{flow}} = \dot{m}P \boldsymbol{v} = (18/60 \text{ kg/s})(300 \text{ kPa})(0.3349 \text{ m}^3/\text{kg}) = 30.14 \text{ kW}$$

(c) The rate of energy transport by mass is

$$\dot{E}_{\text{mass}} = \dot{m}(h + ke) = \dot{m} \left(c_p T + \frac{1}{2} V^2 \right)$$

= (18/60 kg/s) $\left[(1.008 \text{ kJ/kg.K})(77 + 273 \text{ K}) + \frac{1}{2} (25 \text{ m/s})^2 \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) \right]$
= **105.94 kW**

(d) If we neglect kinetic energy in the calculation of energy transport by mass

$$\dot{E}_{\text{mass}} = \dot{m}h = \dot{m}c_p T = (18/60 \text{ kg/s})(1.005 \text{ kJ/kg.K})(77 + 273 \text{ K}) = 105.84 \text{ kW}$$

Therefore, the error involved if neglect the kinetic energy is only 0.09%.

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Steady Flow Energy Balance: Nozzles and Diffusers

5-25C No.

5-26C It is mostly converted to internal energy as shown by a rise in the fluid temperature.

5-27C The kinetic energy of a fluid increases at the expense of the internal energy as evidenced by a decrease in the fluid temperature.

5-28C Heat transfer to the fluid as it flows through a nozzle is desirable since it will probably increase the kinetic energy of the fluid. Heat transfer from the fluid will decrease the exit velocity.

5-29 Air is decelerated in a diffuser from 230 m/s to 30 m/s. The exit temperature of air and the exit area of the diffuser are to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Air is an ideal gas with variable specific heats. **3** Potential energy changes are negligible. **4** The device is adiabatic and thus heat transfer is negligible. **5** There are no work interactions.

Properties The gas constant of air is 0.287 kPa.m³/kg.K (Table A-1). The enthalpy of air at the inlet temperature of 400 K is $h_1 = 400.98$ kJ/kg (Table A-17).

Analysis (a) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take diffuser as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{Bate of net energy transfer} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\frac{\dot{E}_{in} - \dot{E}_{out}}{\dot{E}_{in} = \dot{E}_{out}} = \dot{h}_{0} + \frac{V_{2}^{2}}{2} + \frac{V_{2}^{2}}{2} \quad (since \dot{Q} \cong \dot{W} \cong \Delta pe \cong 0)$$

$$0 = h_{2} - h_{1} + \frac{V_{2}^{2} - V_{1}^{2}}{2} , \quad (since \dot{Q} \cong \dot{W} \cong \Delta pe \cong 0)$$

or,

$$h_2 = h_1 - \frac{V_2^2 - V_1^2}{2} = 400.98 \text{ kJ/kg} - \frac{(30 \text{ m/s})^2 - (230 \text{ m/s})^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2}\right) = 426.98 \text{ kJ/kg}$$

From Table A-17,

$$T_2 = 425.6 \text{ K}$$

(b) The specific volume of air at the diffuser exit is

$$v_2 = \frac{RT_2}{P_2} = \frac{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(425.6 \text{ K})}{(100 \text{ kPa})} = 1.221 \text{ m}^3/\text{kg}$$

From conservation of mass,

$$\dot{m} = \frac{1}{v_2} A_2 V_2 \longrightarrow A_2 = \frac{\dot{m} v_2}{V_2} = \frac{(6000/3600 \text{ kg/s})(1.221 \text{ m}^3/\text{kg})}{30 \text{ m/s}} = 0.0678 \text{ m}^2$$

5-30 Air is accelerated in a nozzle from 45 m/s to 180 m/s. The mass flow rate, the exit temperature, and the exit area of the nozzle are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Air is an ideal gas with constant specific heats. 3 Potential energy changes are negligible. 4 The device is adiabatic and thus heat transfer is negligible. 5 There are no work interactions.

Properties The gas constant of air is 0.287 kPa.m³/kg.K (Table A-1). The specific heat of air at the anticipated average temperature of 450 K is $c_p = 1.02$ kJ/kg.°C (Table A-2).

Analysis (*a*) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. Using the ideal gas relation, the specific volume and the mass flow rate of air are determined to be

$$\boldsymbol{\nu}_{1} = \frac{RT_{1}}{P_{1}} = \frac{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(473 \text{ K})}{300 \text{ kPa}} = 0.4525 \text{ m}^{3}/\text{kg}$$
$$\dot{m} = \frac{1}{\boldsymbol{\nu}_{1}} A_{1} \boldsymbol{\nu}_{1} = \frac{1}{0.4525 \text{ m}^{3}/\text{kg}} (0.0110 \text{ m}^{2})(45 \text{ m/s}) = \mathbf{1.094 \text{ kg/s}}$$



(b) We take nozzle as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{70 \text{ (steady)}} = 0$$

$$\underbrace{\dot{E}_{\text{in}} = \dot{E}_{\text{out}}}_{\text{internal}, \text{ kinetic, energies}} = \dot{D}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = \dot{D}_{\text{internal}, \text{ kinetic, energies}} =$$

Substituting,

$$0 = (1.02 \text{ kJ/kg} \cdot \text{K})(T_2 - 200^{\circ} \text{C}) + \frac{(180 \text{ m/s})^2 - (45 \text{ m/s})^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2}\right)$$

It yields

$T_2 = 185.2^{\circ}C$

(c) The specific volume of air at the nozzle exit is

$$\boldsymbol{v}_2 = \frac{RT_2}{P_2} = \frac{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(185.2 + 273 \text{ K})}{100 \text{ kPa}} = 1.315 \text{ m}^3/\text{kg}$$
$$\dot{m} = \frac{1}{\boldsymbol{v}_2} A_2 \boldsymbol{V}_2 \longrightarrow 1.094 \text{ kg/s} = \frac{1}{1.315 \text{ m}^3/\text{kg}} A_2 (180 \text{ m/s}) \longrightarrow A_2 = 0.00799 \text{ m}^2 = \mathbf{79.9 \text{ cm}^2}$$

5-31 Problem 5-30 is reconsidered. The effect of the inlet area on the mass flow rate, exit velocity, and the exit area as the inlet area varies from 50 cm² to 150 cm² is to be investigated, and the final results are to be plotted against the inlet area.

Analysis The problem is solved using EES, and the solution is given below.

```
Function HCal(WorkFluid$, Tx, Px)

"Function to calculate the enthalpy of an ideal gas or real gas"

If 'Air' = WorkFluid$ then

HCal:=ENTHALPY(Air,T=Tx) "Ideal gas equ."

else

HCal:=ENTHALPY(WorkFluid$,T=Tx, P=Px)"Real gas equ."

endif

end HCal
```

```
"System: control volume for the nozzle"
"Property relation: Air is an ideal gas"
"Process: Steady state, steady flow, adiabatic, no work"
"Knowns - obtain from the input diagram"
WorkFluid$ = 'Air'
T[1] = 200 [C]
P[1] = 300 [kPa]
Vel[1] = 45 [m/s]
P[2] = 100 [kPa]
Vel[2] = 180 [m/s]
A[1]=110 [cm^2]
Am[1]=A[1]*convert(cm^2,m^2)
```

"Property Data - since the Enthalpy function has different parameters for ideal gas and real fluids, a function was used to determine h." h[1]=HCal(WorkFluid\$,T[1],P[1]) h[2]=HCal(WorkFluid\$,T[2],P[2])

```
"The Volume function has the same form for an ideal gas as for a real fluid." v[1]=volume(workFluid$,T=T[1],p=P[1]) v[2]=volume(WorkFluid$,T=T[2],p=P[2])
```

```
"Conservation of mass: "
m_dot[1]= m_dot[2]
"Mass flow rate"
m_dot[1]=Am[1]*Vel[1]/v[1]
m_dot[2]= Am[2]*Vel[2]/v[2]
```

"Conservation of Energy - SSSF energy balance" h[1]+Vel[1]^2/(2*1000) = h[2]+Vel[2]^2/(2*1000)

```
"Definition"
A_ratio=A[1]/A[2]
A[2]=Am[2]*convert(m^2,cm^2)
```

A ₁	A ₂	m ₁	T_2
[cm ²]	[cm ²]	[kg/s]	[C]
50	36.32	0.497	185.2
60	43.59	0.5964	185.2
70	50.85	0.6958	185.2
80	58.12	0.7952	185.2
90	65.38	0.8946	185.2
100	72.65	0.9941	185.2
110	79.91	1.093	185.2
120	87.18	1.193	185.2
130	94.44	1.292	185.2
140	101.7	1.392	185.2
150	109	1.491	185.2



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5-32E Air is accelerated in an adiabatic nozzle. The velocity at the exit is to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Air is an ideal gas with constant specific heats. **3** Potential energy changes are negligible. **4** There are no work interactions. **5** The nozzle is adiabatic.

Properties The specific heat of air at the average temperature of (700+645)/2=672.5°F is $c_p = 0.253$ Btu/lbm·R (Table A-2Eb).

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take nozzle as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as



$$V_{2} = \left[V_{1}^{2} + 2(h_{1} - h_{2})\right]^{0.5} = \left[V_{1}^{2} + 2c_{p}(T_{1} - T_{2})\right]^{0.5}$$
$$= \left[(80 \text{ ft/s})^{2} + 2(0.253 \text{ Btu/lbm} \cdot \text{R})(700 - 645)\text{R}\left(\frac{25,037 \text{ ft}^{2}/\text{s}^{2}}{1 \text{ Btu/lbm}}\right)\right]^{0.5}$$
$$= 838.6 \text{ ft/s}$$

5-33 Air is decelerated in an adiabatic diffuser. The velocity at the exit is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Air is an ideal gas with constant specific heats. 3 Potential energy changes are negligible. 4 There are no work interactions. 5 The diffuser is adiabatic.

Properties The specific heat of air at the average temperature of $(20+90)/2=55^{\circ}C=328$ K is $c_p = 1.007$ kJ/kg·K (Table A-2b).

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take diffuser as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}^{\pi 0 \text{ (steady)}}_{\text{net energy transfer}} = 0$$

$$\underbrace{\dot{E}_{\text{in}} = \dot{E}_{\text{out}}}_{\text{by heat, work, and mass}} = \dot{E}_{\text{out}} = \underbrace{\dot{E}_{\text{out}}}_{0 \text{ (steady)}} = 0$$

$$\frac{\dot{E}_{\text{in}} = \dot{E}_{\text{out}}}{\dot{E}_{\text{in}} = \dot{E}_{\text{out}}} = \underbrace{\dot{h}_{0} + V_2^2/2}_{0 \text{ (h}_1 + V_1^2/2) = \dot{m}(h_2 + V_2^2/2)}_{h_1 + V_1^2/2 = h_2 + V_2^2/2} = \underbrace{\dot{h}_2 + V_2^2/2}_{0 \text{ (steady)}} = 0$$

$$100 \text{ kPa}_{20^\circ\text{C}} = \underbrace{\dot{h}_{0} + V_2^2/2}_{0 \text{ (steady)}} = 0$$

$$100 \text{ kPa}_{20^\circ\text{C}} = \underbrace{\dot{h}_{0} + V_2^2/2}_{0 \text{ (steady)}} = \underbrace{\dot{h}_{0} + V_2^2/2}_{0 \text{ (steady)}}$$

Solving for exit velocity,

$$V_{2} = \left[V_{1}^{2} + 2(h_{1} - h_{2})\right]^{0.5} = \left[V_{1}^{2} + 2c_{p}(T_{1} - T_{2})\right]^{0.5}$$
$$= \left[(500 \text{ m/s})^{2} + 2(1.007 \text{ kJ/kg} \cdot \text{K})(20 - 90)\text{K}\left(\frac{1000 \text{ m}^{2}/\text{s}^{2}}{1 \text{ kJ/kg}}\right)\right]^{0.5}$$
$$= 330.2 \text{ m/s}$$

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5-34 Heat is lost from the steam flowing in a nozzle. The velocity and the volume flow rate at the nozzle exit are to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Potential energy change is negligible. **3** There are no work interactions.

Analysis We take the steam as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as



$$\frac{\dot{E}_{\rm in} - \dot{E}_{\rm out}}{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\rm system}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{70 \text{ (steady)}} = 0$$

$$\frac{\dot{E}_{\rm in}}{\dot{E}_{\rm in}} = \dot{E}_{\rm out}$$

$$\dot{m} \left(h_1 + \frac{V_1^2}{2} \right) = \dot{m} \left(h_2 + \frac{V_2^2}{2} \right) + \dot{Q}_{\rm out} \quad \text{since } \dot{W} \cong \Delta \text{pe} \cong 0 \text{)}$$

$$h_1 + \frac{V_1^2}{2} = h_2 + \frac{V_2^2}{2} + \frac{\dot{Q}_{\rm out}}{\dot{m}}$$

or

The properties of steam at the inlet and exit are (Table A-6)

$$P_{1} = 800 \text{ kPa} \left\{ \boldsymbol{\nu}_{1} = 0.38429 \text{ m}^{3}/\text{kg} \right.$$
$$T_{1} = 400^{\circ}\text{C} \quad \int h_{1} = 3267.7 \text{ kJ/kg}$$
$$P_{2} = 200 \text{ kPa} \left\{ \boldsymbol{\nu}_{2} = 1.31623 \text{ m}^{3}/\text{kg} \right.$$
$$T_{1} = 300^{\circ}\text{C} \quad \left\{ h_{2} = 3072.1 \text{ kJ/kg} \right\}$$

.

The mass flow rate of the steam is

$$\dot{m} = \frac{1}{\nu_1} A_1 V_1 = \frac{1}{0.38429 \text{ m}^3/\text{s}} (0.08 \text{ m}^2)(10 \text{ m/s}) = 2.082 \text{ kg/s}$$

Substituting,

$$3267.7 \text{ kJ/kg} + \frac{(10 \text{ m/s})^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) = 3072.1 \text{ kJ/kg} + \frac{V_2^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) + \frac{25 \text{ kJ/s}}{2.082 \text{ kg/s}}$$
$$\longrightarrow V_2 = 606 \text{ m/s}$$

The volume flow rate at the exit of the nozzle is

$$\dot{\mathbf{V}}_2 = \dot{m}\mathbf{v}_2 = (2.082 \text{ kg/s})(1.31623 \text{ m}^3/\text{kg}) = 2.74 \text{ m}^3/\text{s}$$



5-35 Steam is accelerated in a nozzle from a velocity of 40 m/s to 300 m/s. The exit temperature and the ratio of the inlet-to-exit area of the nozzle are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Potential energy changes are negligible. 3 There are no work interactions. 4 The device is adiabatic and thus heat transfer is negligible.

Properties From the steam tables (Table A-6),

$$P_1 = 3 \text{ MPa}$$
 $v_1 = 0.09938 \text{ m}^3/\text{kg}$
 $T_1 = 400^{\circ}\text{C}$ $h_1 = 3231.7 \text{ kJ/kg}$

Analysis (a) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take nozzle as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as



or,

$$h_2 = h_1 - \frac{V_2^2 - V_1^2}{2} = 3231.7 \text{ kJ/kg} - \frac{(300 \text{ m/s})^2 - (40 \text{ m/s})^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2}\right) = 3187.5 \text{ kJ/kg}$$

Thus,

$$P_2 = 2.5 \text{ MPa}$$
 $T_2 = 376.6 ^{\circ} \text{C}$
 $h_2 = 3187.5 \text{ kJ/kg}$ $v_2 = 0.11533 \text{ m}^3/\text{kg}$

(b) The ratio of the inlet to exit area is determined from the conservation of mass relation,

$$\frac{1}{\boldsymbol{v}_2} A_2 V_2 = \frac{1}{\boldsymbol{v}_1} A_1 V_1 \longrightarrow \frac{A_1}{A_2} = \frac{\boldsymbol{v}_1}{\boldsymbol{v}_2} \frac{V_2}{V_1} = \frac{(0.09938 \text{ m}^3/\text{kg})(300 \text{ m/s})}{(0.11533 \text{ m}^3/\text{kg})(40 \text{ m/s})} = \mathbf{6.46}$$

5-36E Air is decelerated in a diffuser from 600 ft/s to a low velocity. The exit temperature and the exit velocity of air are to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Air is an ideal gas with variable specific heats. **3** Potential energy changes are negligible. **4** The device is adiabatic and thus heat transfer is negligible. **5** There are no work interactions.

Properties The enthalpy of air at the inlet temperature of 50°F is $h_1 = 121.88$ Btu/lbm (Table A-17E).

Analysis (*a*) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take diffuser as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as



or,

$$h_2 = h_1 - \frac{V_2^2 - V_1^2}{2} = 121.88 \text{ Btu/lbm} - \frac{0 - (600 \text{ ft/s})^2}{2} \left(\frac{1 \text{ Btu/lbm}}{25,037 \text{ ft}^2/\text{s}^2}\right) = 129.07 \text{ Btu/lbm}$$

From Table A-17E,

 $T_2 = 540 \text{ R}$

(b) The exit velocity of air is determined from the conservation of mass relation,

$$\frac{1}{v_2} A_2 V_2 = \frac{1}{v_1} A_1 V_1 \longrightarrow \frac{1}{RT_2 / P_2} A_2 V_2 = \frac{1}{RT_1 / P_1} A_1 V_1$$

Thus,

$$V_2 = \frac{A_1 T_2 P_1}{A_2 T_1 P_2} V_1 = \frac{1}{4} \frac{(540 \text{ R})(13 \text{ psia})}{(510 \text{ R})(14.5 \text{ psia})} (600 \text{ ft/s}) = 142 \text{ ft/s}$$

5-37 CO_2 gas is accelerated in a nozzle to 450 m/s. The inlet velocity and the exit temperature are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 CO_2 is an ideal gas with variable specific heats. **3** Potential energy changes are negligible. **4** The device is adiabatic and thus heat transfer is negligible. **5** There are no work interactions.

Properties The gas constant and molar mass of CO₂ are 0.1889 kPa.m³/kg.K and 44 kg/kmol (Table A-1). The enthalpy of CO₂ at 500°C is $\overline{h}_1 = 30,797$ kJ/kmol (Table A-20).

Analysis (a) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. Using the ideal gas relation, the specific volume is determined to be

$$v_1 = \frac{RT_1}{P_1} = \frac{(0.1889 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(773 \text{ K})}{1000 \text{ kPa}} = 0.146 \text{ m}^3/\text{kg}$$

Thus,

$$\dot{m} = \frac{1}{\nu_1} A_1 V_1 \longrightarrow V_1 = \frac{\dot{m} \nu_1}{A_1} = \frac{(6000/3600 \text{ kg/s})(0.146 \text{ m}^3/\text{kg})}{40 \times 10^{-4} \text{ m}^2} = 60.8 \text{ m/s}$$

(b) We take nozzle as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Rate of net energy transfer}} = \Delta \underbrace{\dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0$$

$$\frac{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}{\dot{E}_{\text{in}} = \dot{E}_{\text{out}}} = \dot{E}_{\text{out}}$$

$$\dot{m}(h_1 + V_1^2 / 2) = \dot{m}(h_2 + V_2^2 / 2) \quad (\text{since } \dot{Q} \cong \dot{W} \cong \Delta \text{pe} \cong 0)$$

$$0 = h_2 - h_1 + \frac{V_2^2 - V_1^2}{2}$$

Substituting,

$$\overline{h}_{2} = \overline{h}_{1} - \frac{V_{2}^{2} - V_{1}^{2}}{2} M$$

= 30,797 kJ/kmol - $\frac{(450 \text{ m/s})^{2} - (60.8 \text{ m/s})^{2}}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^{2}/\text{s}^{2}}\right) (44 \text{ kg/kmol})$
= 26,423 kJ/kmol

Then the exit temperature of CO₂ from Table A-20 is obtained to be $T_2 = 685.8 \text{ K}$



5-38 R-134a is accelerated in a nozzle from a velocity of 20 m/s. The exit velocity of the refrigerant and the ratio of the inlet-to-exit area of the nozzle are to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Potential energy changes are negligible. **3** There are no work interactions. **4** The device is adiabatic and thus heat transfer is negligible.

Properties From the refrigerant tables (Table A-13)

$$P_1 = 700 \text{ kPa}$$
 $v_1 = 0.043358 \text{ m}^3/\text{kg}$
 $T_1 = 120^{\circ}\text{C}$ $h_1 = 358.90 \text{ kJ/kg}$

and

$$P_2 = 400 \text{ kPa}$$
 $v_2 = 0.056796 \text{ m}^3/\text{kg}$
 $T_2 = 30^{\circ}\text{C}$ $h_2 = 275.07 \text{ kJ/kg}$



Analysis (*a*) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take nozzle as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0$$

$$\frac{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}{\dot{E}_{\text{in}} = \dot{E}_{\text{out}}} = \dot{h}_{(h_1} + V_1^2/2) = \dot{m}(h_2 + V_2^2/2) \quad (\text{since } \dot{Q} \cong \dot{W} \cong \Delta \text{pe} \cong 0)$$

$$0 = h_2 - h_1 + \frac{V_2^2 - V_1^2}{2}$$

Substituting,

$$0 = (275.07 - 358.90) \text{kJ/kg} + \frac{V_2^2 - (20 \text{ m/s})^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2}\right)$$

It yields

$$V_2 = 409.9 \text{ m/s}$$

(b) The ratio of the inlet to exit area is determined from the conservation of mass relation,

$$\frac{1}{\boldsymbol{v}_2} A_2 V_2 = \frac{1}{\boldsymbol{v}_1} A_1 V_1 \longrightarrow \frac{A_1}{A_2} = \frac{\boldsymbol{v}_1}{\boldsymbol{v}_2} \frac{V_2}{V_1} = \frac{(0.043358 \text{ m}^3/\text{kg})(409.9 \text{ m/s})}{(0.056796 \text{ m}^3/\text{kg})(20 \text{ m/s})} = \mathbf{15.65}$$

1

\

5-39 Nitrogen is decelerated in a diffuser from 275 m/s to a lower velocity. The exit velocity of nitrogen and the ratio of the inlet-to-exit area are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Nitrogen is an ideal gas with variable specific heats. 3 Potential energy changes are negligible. 4 The device is adiabatic and thus heat transfer is negligible. 5 There are no work interactions.

Properties The molar mass of nitrogen is M = 28 kg/kmol (Table A-1). The enthalpies are (Table A-18)

$$T_1 = 7^{\circ}\text{C} = 280 \text{ K} \rightarrow h_1 = 8141 \text{ kJ/kmol}$$
$$T_2 = 27^{\circ}\text{C} = 300 \text{ K} \rightarrow \overline{h_2} = 8723 \text{ kJ/kmol}$$

Analysis (*a*) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take diffuser as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{70 \text{ (steady)}} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{m}(h_1 + V_1^2 / 2) = \dot{m}(h_2 + V_2^2 / 2)} \text{ (since } \dot{Q} \cong \dot{W} \cong \Delta \text{pe} \cong 0)$$

$$0 = h_2 - h_1 + \frac{V_2^2 - V_1^2}{2} = \frac{\bar{h}_2 - \bar{h}_1}{M} + \frac{V_2^2 - V_1^2}{2}$$



Substituting,

$$0 = \frac{(8723 - 8141) \text{ kJ/kmol}}{28 \text{ kg/kmol}} + \frac{V_2^2 - (275 \text{ m/s})^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2}\right)$$

It yields

 $V_2 = 185 \text{ m/s}$

(b) The ratio of the inlet to exit area is determined from the conservation of mass relation,

$$\frac{1}{v_2}A_2V_2 = \frac{1}{v_1}A_1V_1 \longrightarrow \frac{A_1}{A_2} = \frac{v_1}{v_2}\frac{V_2}{V_1} = \left(\frac{RT_1/P_1}{RT_2/P_2}\right)\frac{V_2}{V_1}$$

or,

$$\frac{A_1}{A_2} = \left(\frac{T_1 / P_1}{T_2 / P_2}\right) \frac{V_2}{V_1} = \frac{(280 \text{ K}/60 \text{ kPa})(185 \text{ m/s})}{(300 \text{ K}/85 \text{ kPa})(200 \text{ m/s})} = 0.887$$

5-40 Problem 5-39 is reconsidered. The effect of the inlet velocity on the exit velocity and the ratio of the inlet-toexit area as the inlet velocity varies from 210 m/s to 350 m/s is to be investigated. The final results are to be plotted against the inlet velocity.

Analysis The problem is solved using EES, and the solution is given below.

```
Function HCal(WorkFluid$, Tx, Px)

"Function to calculate the enthalpy of an ideal gas or real gas"

If 'N2' = WorkFluid$ then

HCal:=ENTHALPY(WorkFluid$,T=Tx) "Ideal gas equ."

else

HCal:=ENTHALPY(WorkFluid$,T=Tx, P=Px)"Real gas equ."

endif

end HCal
```

"System: control volume for the nozzle" "Property relation: Nitrogen is an ideal gas" "Process: Steady state, steady flow, adiabatic, no work"

"Knowns" WorkFluid\$ = 'N2' T[1] = 7 [C] P[1] = 60 [kPa] "Vel[1] = 275 [m/s]" P[2] = 85 [kPa] T[2] = 27 [C]

"Property Data - since the Enthalpy function has different parameters for ideal gas and real fluids, a function was used to determine h." h[1]=HCal(WorkFluid\$,T[1],P[1]) h[2]=HCal(WorkFluid\$,T[2],P[2])

"The Volume function has the same form for an ideal gas as for a real fluid." v[1]=volume(workFluid\$,T=T[1],p=P[1]) v[2]=volume(WorkFluid\$,T=T[2],p=P[2])

"From the definition of mass flow rate, m_dot = A*Vel/v and conservation of mass the area ratio A_Ratio = A_1/A_2 is:" A_Ratio*Vel[1]/v[1] =Vel[2]/v[2] "Conservation of Energy - SSSF energy balance" h[1]+Vel[1]^2/(2*1000) = h[2]+Vel[2]^2/(2*1000)

Vel ₁	Vel ₂	A _{Ratio}
[m/s]	[m/s]	
210	50.01	0.3149
224	92.61	0.5467
238	122.7	0.6815
252	148	0.7766
266	170.8	0.8488
280	191.8	0.9059
294	211.7	0.9523
308	230.8	0.9908
322	249.2	1.023
336	267	1.051
350	284.4	1.075





5-41 R-134a is decelerated in a diffuser from a velocity of 120 m/s. The exit velocity of R-134a and the mass flow rate of the R-134a are to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Potential energy changes are negligible. **3** There are no work interactions.

Properties From the R-134a tables (Tables A-11 through A-13)

$$P_1 = 800 \text{ kPa}$$
 $v_1 = 0.025621 \text{ m}^3/\text{kg}$
sat.vapor $h_1 = 267.29 \text{ kJ/kg}$

and

$$P_2 = 900 \text{ kPa} \left\{ \boldsymbol{v}_2 = 0.023375 \text{ m}^3/\text{kg} \right. \\ T_2 = 40^{\circ}\text{C} \left. \right\} h_2 = 274.17 \text{ kJ/kg}$$



Analysis (*a*) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. Then the exit velocity of R-134a is determined from the steady-flow mass balance to be

$$\frac{1}{\nu_2}A_2V_2 = \frac{1}{\nu_1}A_1V_1 \longrightarrow V_2 = \frac{\nu_2}{\nu_1}\frac{A_1}{A_2}V_1 = \frac{1}{1.8}\frac{(0.023375 \text{ m}^3/\text{kg})}{(0.025621 \text{ m}^3/\text{kg})}(120 \text{ m/s}) = 60.8 \text{ m/s}$$

(b) We take diffuser as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{Rate of net energy transfer} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\frac{\dot{Q}_{in} + \dot{m}(h_1 + V_1^2 / 2) = \dot{m}(h_2 + V_2^2 / 2) \quad (since \dot{W} \cong \Delta pe \cong 0)$$

$$\dot{Q}_{in} = \dot{m} \left(h_2 - h_1 + \frac{V_2^2 - V_1^2}{2} \right)$$

Substituting, the mass flow rate of the refrigerant is determined to be

$$2 \text{ kJ/s} = \dot{m} \left((274.17 - 267.29) \text{kJ/kg} + \frac{(60.8 \text{ m/s})^2 - (120 \text{ m/s})^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) \right)$$

It yields

 $\dot{m} =$ 1.308 kg/s

5-42 Steam is accelerated in a nozzle from a velocity of 60 m/s. The mass flow rate, the exit velocity, and the exit area of the nozzle are to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Potential energy changes are negligible. **3** There are no work interactions.

Properties From the steam tables (Table A-6)

$$P_1 = 4 \text{ MPa}$$
 $\mathbf{v}_1 = 0.07343 \text{ m}^3/\text{kg}$
 $T_1 = 400^{\circ}\text{C}$ $\mathbf{h}_1 = 3214.5 \text{ kJ/kg}$

and

$$P_2 = 2 \text{ MPa} \left\{ \boldsymbol{v}_2 = 0.12551 \text{ m}^3/\text{kg} \right\}$$

 $T_2 = 300^{\circ}\text{C} \left\{ \begin{array}{l} h_2 = 3024.2 \text{ kJ/kg} \end{array} \right\}$



Analysis (a) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. The mass flow rate of steam is

$$\dot{m} = \frac{1}{\nu_1} V_1 A_1 = \frac{1}{0.07343 \text{ m}^3/\text{kg}} (60 \text{ m/s})(50 \times 10^{-4} \text{ m}^2) = 4.085 \text{ kg/s}$$

(*b*) We take nozzle as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Bate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0$$

$$\frac{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}{\dot{E}_{\text{in}} = \dot{E}_{\text{out}}}$$

$$\dot{m}(h_1 + V_1^2 / 2) = \dot{Q}_{\text{out}} + \dot{m}(h_2 + V_2^2 / 2) \quad (\text{since } \dot{W} \cong \Delta \text{pe} \cong 0)$$

$$- \dot{Q}_{\text{out}} = \dot{m} \left(h_2 - h_1 + \frac{V_2^2 - V_1^2}{2} \right)$$

Substituting, the exit velocity of the steam is determined to be

$$-75 \text{ kJ/s} = (4.085 \text{ kg/s}) \left(3024.2 - 3214.5 + \frac{V_2^2 - (60 \text{ m/s})^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) \right)$$

It yields

$V_2 = 589.5 \text{ m/s}$

(c) The exit area of the nozzle is determined from

$$\dot{m} = \frac{1}{v_2} V_2 A_2 \longrightarrow A_2 = \frac{\dot{m} v_2}{V_2} = \frac{(4.085 \text{ kg/s})(0.12551 \text{ m}^3/\text{kg})}{589.5 \text{ m/s}} = 8.70 \times 10^{-4} \text{m}^2$$

Turbines and Compressors

5-43C Yes.

5-44C The volume flow rate at the compressor inlet will be greater than that at the compressor exit.

5-45C Yes. Because energy (in the form of shaft work) is being added to the air.

5-46C No.

5-47 R-134a at a given state is compressed to a specified state. The mass flow rate and the power input are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible.

Analysis We take the compressor as the system, which is a control volume since mass crosses the boundary. Noting that one fluid stream enters and leaves the compressor, the energy balance for this steady-flow system can be expressed in the rate form as

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{Bate of net energy transfer} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\vec{E}_{in} = \dot{E}_{out}$$

$$\dot{W}_{in} + \dot{m}h_1 = \dot{m}h_2 \quad (since \Delta ke \cong \Delta pe \cong 0)$$

$$\dot{W}_{in} = \dot{m}(h_2 - h_2)$$

From R134a tables (Tables A-11, A-12, A-13)

$$P_{1} = 100 \text{ kPa} \ h_{1} = 236.33 \text{ kJ/kg}$$

$$T_{1} = -24^{\circ}\text{C} \ v_{1} = 0.1947 \text{ m}^{3}/\text{kg}$$

$$P_{2} = 800 \text{ kPa} \ T_{2} = 60^{\circ}\text{C} \ h_{2} = 296.81 \text{ kJ/kg}$$

The mass flow rate is

$$\dot{m} = \frac{\dot{V}_1}{v_1} = \frac{(1.35/60) \text{ m}^3/\text{s}}{0.1947 \text{ m}^3/\text{kg}} = 0.1155 \text{ kg/s}$$

Substituting,

$$\dot{W}_{in} = \dot{m}(h_2 - h_1) = (0.1155 \text{ kg/s})(296.81 - 236.33)\text{kJ/kg} = 6.99 \text{ kW}$$



5-48 Saturated R-134a vapor is compressed to a specified state. The power input is given. The exit temperature is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Heat transfer with the surroundings is negligible.

Analysis We take the compressor as the system, which is a control volume since mass crosses the boundary. Noting that one fluid stream enters and leaves the compressor, the energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{W}_{in} + \dot{m}h_1 = \dot{m}h_2} \quad (\text{since } \Delta ke \cong \Delta pe \cong 0)$$

$$\dot{W}_{in} = \dot{m}(h_2 - h_2)$$

From R134a tables (Table A-12)

$$P_1 = 180 \text{ kPa}$$
 $h_1 = 242.86 \text{ kJ/kg}$
 $x_1 = 0$ $v_1 = 0.1104 \text{ m}^3/\text{kg}$

The mass flow rate is

$$\dot{m} = \frac{\dot{V}_1}{v_1} = \frac{(0.35/60) \text{ m}^3/\text{s}}{0.1104 \text{ m}^3/\text{kg}} = 0.05283 \text{ kg/s}$$

Substituting for the exit enthalpy,

$$\dot{W}_{in} = \dot{m}(h_2 - h_1)$$

2.35 kW = (0.05283 kg/s)($h_2 - 242.86$)kJ/kg \longrightarrow $h_2 = 287.34$ kJ/kg

From Table A-13,

$$P_{2} = 700 \text{ kPa} \\ h_{2} = 287.34 \text{ kJ/kg}$$



5-49 Steam expands in a turbine. The change in kinetic energy, the power output, and the turbine inlet area are to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Potential energy changes are negligible. **3** The device is adiabatic and thus heat transfer is negligible.

Properties From the steam tables (Tables A-4 through 6)

$$P_1 = 6 \text{ MPa}$$
 $v_1 = 0.047420 \text{ m}^3/\text{kg}$
 $T_1 = 400^{\circ}\text{C}$ $h_1 = 3178.3 \text{ kJ/kg}$

and

$$P_{2} = 40 \text{ kPa} \\ x_{2} = 0.92$$
 $h_{2} = h_{f} + x_{2}h_{fg} = 317.62 + 0.92 \times 2392.1 = 2318.5 \text{ kJ/kg}$

Analysis (a) The change in kinetic energy is determined from

$$\Delta ke = \frac{V_2^2 - V_1^2}{2} = \frac{(50 \text{ m/s})^2 - (80 \text{ m/s})^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2}\right) = -1.95 \text{ kJ/kg}$$

 $P_{1} = 6 \text{ MPa}$ $T_{1} = 400^{\circ}\text{C}$ $V_{1} = 80 \text{ m/s}$ M M = 12 kg/s W $P_{2} = 40 \text{ kPa}$ $x_{2} = 0.92$ $V_{2} = 50 \text{ m/s}$

(b) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the turbine as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Bate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0$$

$$\underbrace{\dot{E}_{\text{in}} = \dot{E}_{\text{out}}}_{\text{in}} = \dot{E}_{\text{out}}$$

$$\dot{m}(h_1 + V_1^2 / 2) = \dot{W}_{\text{out}} + \dot{m}(h_2 + V_2^2 / 2) \quad (\text{since } \dot{Q} \cong \Delta \text{pe} \cong 0)$$

$$\dot{W}_{\text{out}} = -\dot{m} \left(h_2 - h_1 + \frac{V_2^2 - V_1^2}{2} \right)$$

Then the power output of the turbine is determined by substitution to be

$$\dot{W}_{out} = -(20 \text{ kg/s})(2318.5 - 3178.3 - 1.95)\text{kJ/kg} = 14,590 \text{ kW} = 14.6 \text{ MW}$$

(c) The inlet area of the turbine is determined from the mass flow rate relation,

$$\dot{m} = \frac{1}{\nu_1} A_1 V_1 \longrightarrow A_1 = \frac{\dot{m} \nu_1}{V_1} = \frac{(20 \text{ kg/s})(0.047420 \text{ m}^3/\text{kg})}{80 \text{ m/s}} = 0.0119 \text{ m}^2$$

5-50 Problem 5-49 is reconsidered. The effect of the turbine exit pressure on the power output of the turbine as the exit pressure varies from 10 kPa to 200 kPa is to be investigated. The power output is to be plotted against the exit pressure.

Analysis The problem is solved using EES, and the solution is given below.

"Knowns "

T[1] = 450 [C] P[1] = 6000 [kPa] Vel[1] = 80 [m/s] P[2] = 40 [kPa] X_2=0.92 Vel[2] = 50 [m/s] m_dot[1]=12 [kg/s] Fluid\$='Steam IAPWS'

"Property Data"

 $\begin{array}{l} h[1]= enthalpy(Fluid\$, T=T[1], P=P[1]) \\ h[2]= enthalpy(Fluid\$, P=P[2], x=x_2) \\ T[2]= temperature(Fluid\$, P=P[2], x=x_2) \\ v[1]= volume(Fluid\$, T=T[1], p=P[1]) \\ v[2]= volume(Fluid\$, P=P[2], x=x_2) \end{array}$

"Conservation of mass: " m_dot[1]= m_dot[2]

"Mass flow rate" m_dot[1]=A[1]*Vel[1]/v[1] m_dot[2]= A[2]*Vel[2]/v[2]



"Conservation of Energy - Steady Flow energy balance"

m_dot[1]*(h[1]+Vel[1]^2/2*Convert(m^2/s^2, kJ/kg)) = m_dot[2]*(h[2]+Vel[2]^2/2*Convert(m^2/s^2, kJ/kg))+W_dot_turb*convert(MW,kJ/s)

DELTAke=Vel[2]^2/2*Convert(m^2/s^2, kJ/kg)-Vel[1]^2/2*Convert(m^2/s^2, kJ/kg)

P ₂	W _{turb}	T ₂
[kPa]	[MW]	[C]
10	10.95	45.81
31.11	10.39	69.93
52.22	10.1	82.4
73.33	9.909	91.16
94.44	9.76	98.02
115.6	9.638	103.7
136.7	9.535	108.6
157.8	9.446	112.9
178.9	9.367	116.7
200	9.297	120.2



5-51 Steam expands in a turbine. The mass flow rate of steam for a power output of 5 MW is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 The device is adiabatic and thus heat transfer is negligible.

Properties From the steam tables (Tables A-4 through 6)

$$\begin{array}{c} P_{1} = 10 \text{ MPa} \\ T_{1} = 500^{\circ}\text{C} \end{array} \right\} h_{1} = 3375.1 \text{ kJ/kg} \\ P_{2} = 10 \text{ kPa} \\ x_{2} = 0.90 \end{array} \right\} h_{2} = h_{f} + x_{2}h_{fg} = 191.81 + 0.90 \times 2392.1 = 2344.7 \text{ kJ/kg}$$

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the turbine as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}^{70 \text{ (steady)}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0$$

$$\frac{\dot{E}_{in} = \dot{E}_{out}}{\dot{E}_{in} = \dot{E}_{out}} = \dot{E}_{out}$$

$$\frac{\dot{m}h_1 = \dot{W}_{out} + \dot{m}h_2}{\dot{W}_{out} = -\dot{m}(h_2 - h_1)} \text{ (since } \dot{Q} \cong \Delta ke \cong \Delta pe \cong 0)$$

Substituting, the required mass flow rate of the steam is determined to be

5000 kJ/s = $-\dot{m}(2344.7 - 3375.1)$ kJ/kg $\longrightarrow \dot{m} = 4.852$ kg/s

1

H₂O

2

5-33

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible.

Properties From the steam tables (Tables A-4E through 6E)

$$P_{1} = 1000 \text{ psia}$$

$$T_{1} = 900^{\circ}\text{F}$$

$$P_{2} = 5 \text{ psia}$$

$$sat.vapor$$

$$h_{2} = 1130.7 \text{ Btu/lbm}$$

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the turbine as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{Bate of net energy transfer} = \Delta \underline{\dot{E}_{system}}^{70 \text{ (steady)}} = 0$$
Rate of net energy transfer by heat, work, and mass $\dot{E}_{in} = \dot{E}_{out}$

$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{m}h_1 = \dot{Q}_{out} + \dot{W}_{out} + \dot{m}h_2 \quad (\text{since } \Delta ke \cong \Delta pe \cong 0)$$

$$\dot{Q}_{out} = -\dot{m}(h_2 - h_1) - \dot{W}_{out}$$

Substituting,

$$\dot{Q}_{\text{out}} = -(45000/3600 \text{ lbm/s})(1130.7 - 1448.6)\text{Btu/lbm} - 4000 \text{ kJ/s} \left(\frac{1 \text{ Btu}}{1.055 \text{ kJ}}\right) = 182.0 \text{ Btu/s}$$

5-53 Air is compressed at a rate of 10 L/s by a compressor. The work required per unit mass and the power required are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Air is an ideal gas with constant specific heats.

Properties The constant pressure specific heat of air at the average temperature of $(20+300)/2=160^{\circ}C=433$ K is $c_p = 1.018$ kJ/kg·K (Table A-2b). The gas constant of air is R = 0.287 kPa·m³/kg·K (Table A-1).

Analysis (*a*) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the compressor as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

10 L/s



Thus,

$$w_{\rm in} = c_p (T_2 - T_1) = (1.018 \,\text{kJ/kg} \cdot \text{K})(300 - 20)\text{K} = 285.0 \,\text{kJ/kg}$$

(b) The specific volume of air at the inlet and the mass flow rate are

$$\boldsymbol{\nu}_{1} = \frac{RT_{1}}{P_{1}} = \frac{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(20 + 273 \text{ K})}{120 \text{ kPa}} = 0.7008 \text{ m}^{3}/\text{kg}$$
$$\dot{\boldsymbol{m}} = \frac{\boldsymbol{\dot{\nu}}_{1}}{\boldsymbol{\nu}_{1}} = \frac{0.010 \text{ m}^{3}/\text{s}}{0.7008 \text{ m}^{3}/\text{kg}} = 0.01427 \text{ kg/s}$$

Then the power input is determined from the energy balance equation to be

$$\dot{W}_{in} = \dot{m}c_p (T_2 - T_1) = (0.01427 \text{ kg/s})(1.018 \text{ kJ/kg} \cdot \text{K})(300 - 20)\text{K} = 4.068 \text{ kW}$$

5-54 Argon gas expands in a turbine. The exit temperature of the argon for a power output of 190 kW is to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Potential energy changes are negligible. **3** The device is adiabatic and thus heat transfer is negligible. **4** Argon is an ideal gas with constant specific heats.

Properties The gas constant of Ar is R = 0.2081 kPa.m³/kg.K. The constant pressure specific heat of Ar is $c_p = 0.5203$ kJ/kg.°C (Table A-2a)

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. The inlet specific volume of argon and its mass flow rate are

$$v_1 = \frac{RT_1}{P_1} = \frac{(0.2081 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(723 \text{ K})}{1600 \text{ kPa}} = 0.09404 \text{ m}^3/\text{kg}$$

Thus,

$$\dot{m} = \frac{1}{\nu_1} A_1 V_1 = \frac{1}{0.09404 \text{ m}^3/\text{kg}} (0.006 \text{ m}^2) (55 \text{ m/s}) = 3.509 \text{ kg/s}$$

We take the turbine as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{\text{by heat, work, and mass}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0$$

$$\dot{m}(h_1 + V_1^2/2) = \dot{W}_{out} + \dot{m}(h_2 + V_2^2/2) \quad (\text{since } \dot{Q} \cong \Delta \text{pe} \cong 0)$$

 $\dot{W}_{out} = -\dot{m}\left(h_2 - h_1 + \frac{V_2^2 - V_1^2}{2}\right)$

 $A_1 = 60 \text{ cm}^2$ $P_1 = 1600 \text{ kPa}$ $T_1 = 450^{\circ}\text{C}$ $V_1 = 55 \text{ m/s}$ ARGON $P_2 = 150 \text{ kPa}$ $V_2 = 150 \text{ m/s}$

Substituting,

$$190 \text{ kJ/s} = -(3.509 \text{ kg/s}) \left[(0.5203 \text{ kJ/kg} \cdot ^{\circ}\text{C})(T_2 - 450^{\circ}\text{C}) + \frac{(150 \text{ m/s})^2 - (55 \text{ m/s})^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) \right]$$

It yields

$$T_2 = 327^{\circ}C$$
5-55 Helium is compressed by a compressor. For a mass flow rate of 90 kg/min, the power input required is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Helium is an ideal gas with constant specific heats.

Properties The constant pressure specific heat of helium is $c_p = 5.1926 \text{ kJ/kg·K}$ (Table A-2a).

= 0

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the

 $\dot{W}_{in} + \dot{m}h_1 = \dot{Q}_{out} + \dot{m}h_2$ (since $\Delta ke \cong \Delta pe \cong 0$)

 $\dot{W}_{in} - \dot{Q}_{out} = \dot{m}(h_2 - h_1) = \dot{m}c_p(T_2 - T_1)$

 $\underline{\dot{E}_{in} - \dot{E}_{out}}_{by heat, work, and mass} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies}^{70 (steady)}$

 $\dot{E}_{in} = \dot{E}_{out}$

compressor as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

 $P_{2} = 700 \text{ kPa}$ $T_{2} = 430 \text{ K}$ $P_{1} = 120 \text{ kPa}$ $T_{1} = 310 \text{ K}$

Thus,

$$\dot{W}_{in} = \dot{Q}_{out} + \dot{m}c_p(T_2 - T_1)$$

= (90/60 kg/s)(20 kJ/kg) + (90/60 kg/s)(5.1926 kJ/kg · K)(430 - 310)K
= **965 kW**

5-56 CO_2 is compressed by a compressor. The volume flow rate of CO_2 at the compressor inlet and the power input to the compressor are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Helium is an ideal gas with variable specific heats. 4 The device is adiabatic and thus heat transfer is negligible.

Properties The gas constant of CO₂ is R = 0.1889 kPa.m³/kg.K, and its molar mass is M = 44 kg/kmol (Table A-1). The inlet and exit enthalpies of CO₂ are (Table A-20)

$$T_1 = 300 \text{ K} \rightarrow h_1 = 9,431 \text{ kJ/kmol}$$
$$T_2 = 450 \text{ K} \rightarrow \overline{h_2} = 15,483 \text{ kJ/kmol}$$

Analysis (a) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. The inlet specific volume of air and its volume flow rate are

$$\boldsymbol{v}_{1} = \frac{RT_{1}}{P_{1}} = \frac{\left(0.1889 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K}\right)(300 \text{ K})}{100 \text{ kPa}} = 0.5667 \text{ m}^{3}/\text{kg}$$
$$\dot{\boldsymbol{V}} = \dot{\boldsymbol{m}}\boldsymbol{v}_{1} = (0.5 \text{ kg/s})(0.5667 \text{ m}^{3}/\text{kg}) = 0.283 \text{ m}^{3}/\text{s}$$

(b) We take the compressor as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \Delta \dot{E}_{system}^{70 \text{ (steady)}} = 0$$
Rate of net energy transfer by heat, work, and mass $\dot{E}_{in} = \dot{E}_{out}$

$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{W}_{in} + \dot{m}h_1 = \dot{m}h_2 \quad (\text{since } \dot{Q} \cong \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

$$\dot{W}_{in} = \dot{m}(h_2 - h_1) = \dot{m}(\bar{h}_2 - \bar{h}_1)/M$$

Substituting

by l

$$\dot{W}_{in} = \frac{(0.5 \text{ kg/s})(15,483 - 9,431 \text{ kJ/kmol})}{44 \text{ kg/kmol}} = 68.8 \text{ kW}$$



Assumptions 1 This is a steady-flow process since there is no change with time. 2 The turbine is well-insulated, and thus there is no heat transfer. 3 Air is an ideal gas with constant specific heats.

Properties The constant pressure specific heat of air at the average temperature of $(500+127)/2=314^{\circ}C=587$ K is $c_p = 1.048$ kJ/kg·K (Table A-2b). The gas constant of air is R = 0.287 kPa·m³/kg·K (Table A-1).

Analysis (*a*) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the turbine as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\frac{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{70 \text{ (steady)}} = 0$$

$$\frac{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}{\dot{E}_{\text{in}} = \dot{E}_{\text{out}}} = \dot{E}_{\text{out}}$$

$$\dot{m} \left(h_1 + \frac{V_1^2}{2} \right) = \dot{m} \left(h_2 + \frac{V_2^2}{2} \right) + \dot{W}_{\text{out}}$$

$$\dot{W}_{\text{out}} = \dot{m} \left(h_1 - h_2 + \frac{V_1^2 - V_2^2}{2} \right) = \dot{m} \left(c_p (T_1 - T_2) + \frac{V_1^2 - V_2^2}{2} \right)$$

$$\frac{100 \text{ kPa}}{127^{\circ}\text{C}}$$

The specific volume of air at the inlet and the mass flow rate are

$$\boldsymbol{\nu}_{1} = \frac{RT_{1}}{P_{1}} = \frac{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(500 + 273 \text{ K})}{1300 \text{ kPa}} = 0.1707 \text{ m}^{3}/\text{kg}$$
$$\dot{\boldsymbol{m}} = \frac{A_{1}V_{1}}{\boldsymbol{\nu}_{1}} = \frac{(0.2 \text{ m}^{2})(40 \text{ m/s})}{0.1707 \text{ m}^{3}/\text{kg}} = \mathbf{46.88 \text{ kg/s}}$$

Similarly at the outlet,

$$v_2 = \frac{RT_2}{P_2} = \frac{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(127 + 273 \text{ K})}{100 \text{ kPa}} = 1.148 \text{ m}^3/\text{kg}$$

 $V_2 = \frac{\dot{m}v_2}{A_2} = \frac{(46.88 \text{ kg/s})(1.148 \text{ m}^3/\text{kg})}{1 \text{ m}^2} = 53.82 \text{ m/s}$

(b) Substituting into the energy balance equation gives

$$\dot{W}_{\text{out}} = \dot{m} \left(c_p (T_1 - T_2) + \frac{V_1^2 - V_2^2}{2} \right)$$

= (46.88 kg/s) $\left[(1.048 \text{ kJ/kg} \cdot \text{K})(500 - 127)\text{K} + \frac{(40 \text{ m/s})^2 - (53.82 \text{ m/s})^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) \right]$
= **18.300 kW**

Assumptions 1 This is a steady-flow process since there is no change with time. 2 The turbine is well-insulated, and thus there is no heat transfer. 3 Air is an ideal gas with constant specific heats.

Properties The constant pressure specific heat of air at the average temperature of $(800+250)/2=525^{\circ}F$ is $c_p = 0.2485$ Btu/lbm·R (Table A-2Eb). The gas constant of air is R = 0.3704 psia·ft³/lbm·R (Table A-1E).

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the turbine as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\frac{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{70 \text{ (steady)}} = 0$$

$$\frac{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}{\dot{E}_{\text{in}} = \dot{E}_{\text{out}}} = \dot{h} \left(h_1 + \frac{V_1^2}{2} \right) = \dot{m} \left(h_2 + \frac{V_2^2}{2} \right) + \dot{W}_{\text{out}}$$

$$\frac{\dot{W}_{\text{out}}}{\dot{W}_{\text{out}}} = \dot{m} \left(h_1 - h_2 + \frac{V_1^2 - V_2^2}{2} \right) = \dot{m} \left(c_p (T_1 - T_2) + \frac{V_1^2 - V_2^2}{2} \right)$$

$$500 \text{ psia} = 500 \text{ psia$$

The specific volume of air at the exit and the mass flow rate are

$$\boldsymbol{v}_{2} = \frac{RT_{2}}{P_{2}} = \frac{(0.3704 \text{ psia} \cdot \text{ft}^{3}/\text{lbm} \cdot \text{R})(250 + 460 \text{ R})}{60 \text{ psia}} = 4.383 \text{ ft}^{3}/\text{lbm}$$
$$\dot{\boldsymbol{m}} = \frac{\dot{\boldsymbol{V}}_{2}}{\boldsymbol{v}_{2}} = \frac{50 \text{ ft}^{3}/\text{s}}{4.383 \text{ ft}^{3}/\text{lbm}} = 11.41 \text{ kg/s}$$
$$V_{2} = \frac{\dot{\boldsymbol{m}}\boldsymbol{v}_{2}}{A_{2}} = \frac{(11.41 \text{ lbm/s})(4.383 \text{ ft}^{3}/\text{lbm})}{1.2 \text{ ft}^{2}} = 41.68 \text{ ft/s}$$

Similarly at the inlet,

$$\boldsymbol{v}_{1} = \frac{RT_{1}}{P_{1}} = \frac{(0.3704 \text{ psia} \cdot \text{ft}^{3}/\text{lbm} \cdot \text{R})(800 + 460 \text{ R})}{500 \text{ psia}} = 0.9334 \text{ ft}^{3}/\text{lbm}$$
$$V_{1} = \frac{\dot{m}\boldsymbol{v}_{1}}{A_{1}} = \frac{(11.41 \text{ lbm/s})(0.9334 \text{ ft}^{3}/\text{lbm})}{0.6 \text{ ft}^{2}} = 17.75 \text{ ft/s}$$

Substituting into the energy balance equation gives

$$\dot{W}_{\text{out}} = \dot{m} \left(c_p \left(T_1 - T_2 \right) + \frac{V_1^2 - V_2^2}{2} \right)$$

= (11.41 lbm/s) $\left[(0.2485 \text{ Btu/lbm} \cdot \text{R})(800 - 250)\text{R} + \frac{(17.75 \text{ ft/s})^2 - (41.68 \text{ m/s})^2}{2} \left(\frac{1 \text{ Btu/lbm}}{25,037 \text{ ft}^2/\text{s}^2} \right) \right]$
= **1559 Btu/s = 1645 kW**

5-59 Steam expands in a two-stage adiabatic turbine from a specified state to another state. Some steam is extracted at the end of the first stage. The power output of the turbine is to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Kinetic and potential energy changes are negligible. **3** The turbine is adiabatic and thus heat transfer is negligible.

Properties From the steam tables (Tables A-5 and A-6)

$$P_{1} = 5 \text{ MPa}
T_{1} = 600^{\circ}\text{C}
h_{1} = 3666.9 \text{ kJ/kg}
P_{2} = 0.5 \text{ MPa}
x_{2} = 1
h_{2} = 2748.1 \text{ kJ/kg}
P_{3} = 10 \text{ kPa}
h_{3} = h_{f} + xh_{fg}
x_{2} = 0.85
= 191.81 + (0.85)(2392.1) = 2225.1 \text{ kJ/kg}$$

Analysis We take the entire turbine, including the connection part between the two stages, as the system, which is a control volume since mass crosses the boundary. Noting that one fluid stream enters the turbine and two fluid streams leave, the energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{by heat, work, and mass} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{m}_1 h_1 = \dot{m}_2 h_2 + \dot{m}_3 h_3 + \dot{W}_{out}}_{\dot{W}_{out} = \dot{m}_1 (h_1 - 0.1h_2 - 0.9h_3)}$$

Substituting, the power output of the turbine is

$$\dot{W}_{out} = \dot{m}_1 (h_1 - 0.1h_2 - 0.9h_3)$$

= (20 kg/s)(3666.9 - 0.1 × 2748.1 - 0.9 × 2225.1) kJ/kg
= **27,790 kW**



Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible.

Properties From the steam tables (Table A-4, A5, A-6)

$$\begin{array}{c} P_{1} = 6 \text{ MPa} \\ T_{1} = 600^{\circ}\text{C} \end{array} \right\} h_{1} = 3658.8 \text{ kJ/kg} \\ P_{2} = 0.5 \text{ MPa} \\ T_{2} = 200^{\circ}\text{C} \end{array} \right\} h_{2} = 2855.8 \text{ kJ/kg}$$

Analysis We take the turbine as the system, which is a control volume since mass crosses the boundary. Noting that one fluid stream enters and leaves the compressor, the energy balance for this steady-flow system can be expressed in the rate form as



Substituting,

$$\dot{Q}_{\text{out}} = -\dot{W}_{\text{out}} + \dot{m} \left(h_1 - h_2 + \frac{V_1^2 - V_2^2}{2} \right)$$

= 20,000 kW + (26 kg/s) $\left[(3658.8 - 2855.8) \text{kJ/kg} + \frac{(0 - 180 \text{ m/s})^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) \right]$
= **455 kW**

5-61 Helium at a specified state is compressed to another specified state. The power input is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Helium is an ideal gas.

Properties The properties of helium are $c_p = 5.1926 \text{ kJ/kg} \cdot \text{K}$ and $R = 2.0769 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K}$ (Table A-2*a*).

Analysis (a) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the compressor as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{70 \text{ (steady)}} = 0$$

$$\underbrace{Aate of change in internal, kinetic, potential, etc. energies}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{200 \text{ vC}} \underbrace{200^{\circ}\text{C}}_{\text{Compressor}}^{200^{\circ}\text{C}}_{\text{in}} = \dot{E}_{out}$$

$$\underbrace{\dot{W}_{in} + \dot{m}h_1 = \dot{m}h_2}_{\dot{m}_1} \text{ (since } \Delta \text{ke} \cong \Delta \text{pe} \cong 0)}_{\dot{W}_{in}} = \dot{m}(h_2 - h_1) = \dot{m}c_p(T_2 - T_1)}$$

The mass flow rate is determined from

$$\dot{m} = \frac{A_1 V_1}{\nu_1} = \frac{A_1 V_1 P_1}{RT_1} = \frac{(0.1 \text{ m}^2)(15 \text{ m/s})(150 \text{ kPa})}{(2.0769 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(293 \text{ K})} 0.3697 \text{ kg/s}$$

Substituting,

$$\dot{W}_{in} = \dot{m}c_{p}(T_{2} - T_{1}) = (0.3697 \text{ kg/s})(5.1926 \text{ kJ/kg} \cdot \text{K})(200 - 20)\text{K} = 346 \text{ kW}$$

The flow power input is determined from

$$\dot{W}_{fw} = \dot{m}(P_2 v_2 - P_1 v_1) = \dot{m}R(T_2 - T_1) = (0.3697 \text{ kg/s})(2.0769 \text{ kJ/kg} \cdot \text{K})(200 - 20)\text{K} = 138 \text{ kW}$$



Throttling Valves

5-62C The temperature of a fluid can increase, decrease, or remain the same during a throttling process. Therefore, this claim is valid since no thermodynamic laws are violated.

5-63C No. Because air is an ideal gas and h = h(T) for ideal gases. Thus if h remains constant, so does the temperature.

5-64C If it remains in the liquid phase, no. But if some of the liquid vaporizes during throttling, then yes.

5-65C Yes.

5-66 Refrigerant-134a is throttled by a capillary tube. The quality of the refrigerant at the exit is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Heat transfer to or from the fluid is negligible. 4 There are no work interactions involved.

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the throttling valve as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\dot{E}_{in} - \dot{E}_{out} = \Delta \dot{E}_{system} {}^{70 \text{ (steady)}} = 0$$
$$\dot{E}_{in} = \dot{E}_{out}$$
$$\dot{m}h_1 = \dot{m}h_2$$
$$h_1 = h_2$$

since $\dot{Q} \cong \dot{W} = \Delta ke \cong \Delta pe \cong 0$.

The inlet enthalpy of R-134a is, from the refrigerant tables (Table A-11),

$$\begin{array}{c} T_1 = 50^{\circ}\text{C} \\ \text{sat. liquid} \end{array} \right\} h_1 = h_f = 123.49 \text{ kJ/kg}$$

The exit quality is

$$T_2 = -20^{\circ} C h_2 = h_1$$
 $x_2 = \frac{h_2 - h_f}{h_{fg}} = \frac{123.49 - 25.49}{212.91} = 0.460$



Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Heat transfer to or from the fluid is negligible. 4 There are no work interactions involved.

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the throttling valve as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\dot{E}_{in} - \dot{E}_{out} = \Delta \dot{E}_{system}^{70 \text{ (steady)}} = 0$$

$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{m}h_1 = \dot{m}h_2$$

$$h_1 = h_2$$

$$MPa \qquad \qquad \text{Throttling valve}$$

$$100 \text{ kPa}$$

$$120^{\circ}\text{C}$$

Since $\dot{Q} \cong \dot{W} = \Delta ke \cong \Delta pe \cong 0$.

The enthalpy of steam at the exit is (Table A-6),

$$\left. \begin{array}{c} P_2 = 100 \, \mathrm{kPa} \\ T_2 = 120^{\circ} \mathrm{C} \end{array} \right\} h_2 = 2716.1 \, \mathrm{kJ/kg}$$

The quality of the steam at the inlet is (Table A-5)

$$P_1 = 2000 \text{ kPa} \\ h_1 = h_2 = 2716.1 \text{ kJ/kg}$$
 $x_1 = \frac{h_2 - h_f}{h_{fg}} = \frac{2716.1 - 908.47}{1889.8} = 0.957$

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5-68 Refrigerant-134a is throttled by a valve. The pressure and internal energy after expansion are to be determined. Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Heat transfer to or from the fluid is negligible. 4 There are no work interactions involved.

Properties The inlet enthalpy of R-134a is, from the refrigerant tables (Tables A-11 through 13),

$$P_1 = 0.8 \text{ MPa} \\ T_1 = 25^{\circ}\text{C}$$

$$h_1 \cong h_{f@25^{\circ}\text{C}} = 86.41 \text{ kJ/kg}$$

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the throttling valve as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

since $Q \cong W = \Delta ke \cong \Delta pe \cong 0$. Then,

$$T_{2} = -20^{\circ}\text{C} \quad h_{f} = 25.49 \text{ kJ/kg}, \quad u_{f} = 25.39 \text{ kJ/kg} \\ (h_{2} = h_{1}) \quad \int h_{g} = 238.41 \text{ kJ/kg} \quad u_{g} = 218.84 \text{ kJ/kg}$$

Obviously $h_f < h_2 < h_g$, thus the refrigerant exists as a saturated mixture at the exit state, and thus

$$P_2 = P_{\text{sat} @.-20^{\circ}\text{C}} = 132.82 \text{ kPa}$$

Also,

$$x_2 = \frac{h_2 - h_f}{h_{fg}} = \frac{86.41 - 25.49}{212.91} = 0.2861$$

Thus,

$$u_2 = u_f + x_2 u_{fg} = 25.39 + 0.2861 \times 193.45 = 80.74 \text{ kJ/kg}$$



5-69 Steam is throttled by a well-insulated valve. The temperature drop of the steam after the expansion is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Heat transfer to or from the fluid is negligible. 4 There are no work interactions involved.

Properties The inlet enthalpy of steam is (Tables A-6),

$$P_1 = 8 \text{ MPa}$$

 $T_1 = 350^{\circ}\text{C}$ $h_1 = 2988.1 \text{ kJ/kg}$

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the throttling valve as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\dot{E}_{in} - \dot{E}_{out} = \Delta \dot{E}_{system} 70 \text{ (steady)} = 0$$
$$\dot{E}_{in} = \dot{E}_{out}$$
$$\dot{m}h_1 = \dot{m}h_2$$
$$h_1 = h_2$$

since $\dot{Q} \cong \dot{W} = \Delta ke \cong \Delta pe \cong 0$. Then the exit temperature of steam becomes

$$\begin{array}{c} P_2 = 2 \text{ MPa} \\ \left(h_2 = h_1\right) \end{array} \right\} T_2 = \mathbf{285}^{\circ}\mathbf{C}$$

$$P_1 = 8 \text{ MPa}$$

$$T_1 = 350^{\circ}\text{C}$$

$$H_2\text{O}$$

$$P_2 = 2 \text{ MPa}$$

5-70 B

5-70 Problem 5-69 is reconsidered. The effect of the exit pressure of steam on the exit temperature after throttling as the exit pressure varies from 6 MPa to 1 MPa is to be investigated. The exit temperature of steam is to be plotted against the exit pressure.

Analysis The problem is solved using EES, and the solution is given below.

WorkingFluid\$='Steam_iapws' "WorkingFluid can be changed to ammonia or other fluids" P_in=8000 [kPa] T_in=350 [C] P_out=2000 [kPa]

"Analysis"

m_dot_in=m_dot_out "steady-state mass balance"

m_dot_in=1 "mass flow rate is arbitrary"

m_dot_in*h_in+Q_dot-W_dot-m_dot_out*h_out=0 "steady-state energy balance"

Q_dot=0 "assume the throttle to operate adiabatically"

W_dot=0 "throttles do not have any means of producing power"

h_in=enthalpy(WorkingFluid\$,T=T_in,P=P_in) "property table lookup"

T_out=temperature(WorkingFluid\$,P=P_out,h=h_out) "property table lookup"

x_out=quality(WorkingFluid\$,P=P_out,h=h_out) "x_out is the quality at the outlet"

P[1]=P_in; P[2]=P_out; h[1]=h_in; h[2]=h_out "use arrays to place points on property plot"





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Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Heat transfer to or from the fluid is negligible. 4 There are no work interactions involved.

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the throttling valve as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\dot{E}_{in} - \dot{E}_{out} = \Delta \dot{E}_{system}^{70 \text{ (steady)}} = 0$$

$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{m}h_1 = \dot{m}h_2$$

$$h_1 = h_2$$

since $\dot{Q} \cong \dot{W} = \Delta ke \cong \Delta pe \cong 0$. The properties are (Tables A-11E through 13E),

That is, the temperature drops by 94.1°F and internal energy drops by 2.7 Btu/lbm.

Mixing Chambers and Heat Exchangers

5-72C Under the conditions of no heat and work interactions between the mixing chamber and the surrounding medium.

5-73C Under the conditions of no heat and work interactions between the heat exchanger and the surrounding medium.

5-74C Yes, if the mixing chamber is losing heat to the surrounding medium.

5-75 Hot and cold streams of a fluid are mixed in a mixing chamber. Heat is lost from the chamber. The energy carried from the mixing chamber is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 There are no work and heat interactions.

Analysis We take the mixing device as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as

 $\underline{\dot{E}_{in} - \dot{E}_{out}}_{Bate of net energy transfer} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$ $\dot{E}_{in} = \dot{E}_{out}$ $\dot{m}_1 e_1 + \dot{m}_2 e_2 = \dot{m}_3 e_3 + \dot{Q}_{out}$

From a mass balance

$$\dot{m}_3 = \dot{m}_1 + \dot{m}_2 = 5 + 15 = 20 \text{ kg/s}$$

Substituting into the energy balance equation solving for the exit enthalpy gives

$$\dot{m}_1 e_1 + \dot{m}_2 e_2 = \dot{m}_3 e_3 + \dot{Q}_{out}$$

$$e_3 = \frac{\dot{m}_1 e_1 + \dot{m}_2 e_2 - \dot{Q}_{out}}{\dot{m}_3} = \frac{(5 \text{ kg/s})(150 \text{ kJ/kg} + (15 \text{ kg/s})(50 \text{ kJ/kg}) - 5.5 \text{ kW}}{20 \text{ kg/s}} = 74.7 \text{ kJ/kg}$$



5-76 A hot water stream is mixed with a cold water stream. For a specified mixture temperature, the mass flow rate of cold water is to be determined.

Assumptions **1** Steady operating conditions exist. **2** The mixing chamber is well-insulated so that heat loss to the surroundings is negligible. **3** Changes in the kinetic and potential energies of fluid streams are negligible. **4** Fluid properties are constant. **5** There are no work interactions.

Properties Noting that $T < T_{\text{sat} @ 250 \text{ kPa}} = 127.41^{\circ}\text{C}$, the water in all three streams exists as a compressed liquid, which can be approximated as a saturated liquid at the given temperature. Thus,

$$h_1 \cong h_{f@.80^{\circ}C} = 335.02 \text{ kJ/kg}$$

 $h_2 \cong h_{f@.20^{\circ}C} = 83.915 \text{ kJ/kg}$
 $h_3 \cong h_{f@.42^{\circ}C} = 175.90 \text{ kJ/kg}$

Analysis We take the mixing chamber as the system, which is a control volume. The mass and energy balances for this steady-flow system can be expressed in the rate form as

Mass balance:

$$\dot{m}_{\rm in} - \dot{m}_{\rm out} = \Delta \dot{m}_{\rm system}^{70 \text{ (steady)}} = 0 \longrightarrow \dot{m}_1 + \dot{m}_2 = \dot{m}_3$$

Energy balance:

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{by heat, work, and mass} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\underline{\dot{E}_{in}}_{bin} = \underline{\dot{E}_{out}}$$

$$\dot{m}_1 h_1 + \dot{m}_2 h_2 = \dot{m}_3 h_3 \quad (since \dot{Q} = \dot{W} = \Delta ke \cong \Delta pe \cong 0)$$

Combining the two relations and solving for \dot{m}_2 gives

$$\dot{m}_1 h_1 + \dot{m}_2 h_2 = (\dot{m}_1 + \dot{m}_2) h_3$$
$$\dot{m}_2 = \frac{h_1 - h_3}{h_2 - h_2} \dot{m}_1$$

Substituting, the mass flow rate of cold water stream is determined to be

$$\dot{m}_2 = \frac{(335.02 - 175.90) \text{ kJ/kg}}{(175.90 - 83.915) \text{ kJ/kg}} (0.5 \text{ kg/s}) = 0.865 \text{ kg/s}$$



5-77E Liquid water is heated in a chamber by mixing it with saturated water vapor. If both streams enter at the same rate, the temperature and quality (if saturated) of the exit stream is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 There are no work interactions. 4 The device is adiabatic and thus heat transfer is negligible.

Properties From steam tables (Tables A-5E through A-6E),

 $h_1 \cong h_{f(\bar{a}),65^\circ F} = 33.08 \text{ Btu/lbm}$ $h_2 = h_{g(a), 20 \text{ psia}} = 1156.2 \text{ Btu/lbm}$

Analysis We take the mixing chamber as the system, which is a control volume since mass crosses the boundary. The mass and energy balances for this steady-flow system can be expressed in the rate form as

Mass balance:

$$\dot{m}_{in} - \dot{m}_{out} = \Delta \dot{m}_{system}^{\text{AO (steady)}} = 0$$
$$\dot{m}_{in} = \dot{m}_{out}$$
$$\dot{m}_1 + \dot{m}_2 = \dot{m}_3 = 2\dot{m}$$
$$\dot{m}_1 = \dot{m}_2 = \dot{m}$$

Energy balance:

Rate

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{by heat, work, and mass} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{m}_{1}h_{1} + \dot{m}_{2}h_{2} = \dot{m}_{3}h_{3} \quad (\text{since } \dot{O} = \dot{W} = \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

$$\dot{m}_1 h_1 + \dot{m}_2 h_2 = \dot{m}_3 h_3$$
 (since $Q = W = \Delta ke \cong \Delta pe \cong$

Combining the two gives

$$\dot{m}h_1 + \dot{m}h_2 = 2\dot{m}h_3$$
 or $h_3 = (h_1 + h_2)/2$

Substituting,

$$h_3 = (33.08 + 1156.2)/2 = 594.6$$
 Btu/lbm

At 20 psia, $h_f = 196.27$ Btu/lbm and $h_g = 1156.2$ Btu/lbm. Thus the exit stream is a saturated mixture since $h_f < h_3 < h_g$. Therefore,

$$T_3 = T_{\text{sat} @ 20 \text{ psia}} = 228^{\circ} \mathbf{F}$$

and

$$x_3 = \frac{h_3 - h_f}{h_{fg}} = \frac{594.6 - 196.27}{1156.2 - 196.27} = 0.415$$



5-78 Two streams of refrigerant-134a are mixed in a chamber. If the cold stream enters at twice the rate of the hot stream, the temperature and quality (if saturated) of the exit stream are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 There are no work interactions. 4 The device is adiabatic and thus heat transfer is negligible.

Properties From R-134a tables (Tables A-11 through A-13),

$$h_1 \cong h_{f@,20^{\circ}C} = 79.32 \text{ kJ/kg}$$

 $h_2 = h_{@,1 \text{ MPa}, 80^{\circ}C} = 314.25 \text{ kJ/kg}$

Analysis We take the mixing chamber as the system, which is a control volume since mass crosses the boundary. The mass and energy balances for this steady-flow system can be expressed in the rate form as

Mass balance:

$$\dot{m}_{in} - \dot{m}_{out} = \Delta \dot{m}_{system}^{70 (steady)} = 0$$

$$\dot{m}_{in} = \dot{m}_{out}$$

$$\dot{m}_1 + \dot{m}_2 = \dot{m}_3 = 3\dot{m}_2 \text{ since } \dot{m}_1 = 2\dot{m}_2$$

Energy balance:

$$\underline{\dot{E}_{in}} - \underline{\dot{E}_{out}}_{system} = \underbrace{\Delta \dot{E}_{system}}_{system} = 0$$
Rate of net energy transfer by heat, work, and mass

$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{m}_1 h_1 + \dot{m}_2 h_2 = \dot{m}_3 h_3 \quad (\text{since } \dot{Q} = \dot{W} = \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

Combining the two gives

$$2\dot{m}_2h_1 + \dot{m}_2h_2 = 3\dot{m}_2h_3$$
 or $h_3 = (2h_1 + h_2)/3$

Substituting,

$$h_3 = (2 \times 79.32 + 314.25)/3 = 157.63 \text{ kJ/kg}$$

At 1 MPa, $h_f = 107.32$ kJ/kg and $h_g = 270.99$ kJ/kg. Thus the exit stream is a saturated mixture since $h_f < h_3 < h_g$. Therefore,

$$T_3 = T_{\text{sat}(a) \ 1 \text{ MPa}} = 39.37^{\circ}\text{C}$$

and

$$x_3 = \frac{h_3 - h_f}{h_{fg}} = \frac{157.63 - 107.32}{270.99 - 107.32} = 0.307$$



5-79 Problem 5-78 is reconsidered. The effect of the mass flow rate of the cold stream of R-134a on the temperature and the quality of the exit stream as the ratio of the mass flow rate of the cold stream to that of the hot stream varies from 1 to 4 is to be investigated. The mixture temperature and quality are to be plotted against the cold-to-hot mass flow rate ratio.

Analysis The problem is solved using EES, and the solution is given below.

"Input Data" m_frac = 2 "m_frac =m_dot_cold/m_dot_hot= m_dot_1/m_dot_2" T[1]=20 [C] P[1]=1000 [kPa] T[2]=80 [C] P[2]=1000 [kPa] m_dot_1=m_frac*m_dot_2 P[3]=1000 [kPa] m_dot_1=1

"Conservation of mass for the R134a: Sum of m_dot_in=m_dot_out" m_dot_1+ m_dot_2 =m_dot_3

"Conservation of Energy for steady-flow: neglect changes in KE and PE" "We assume no heat transfer and no work occur across the control surface." E_dot_in - E_dot_out = DELTAE_dot_cv DELTAE_dot_cv=0 "Steady-flow requirement" E_dot_in=m_dot_1*h[1] + m_dot_2*h[2] E_dot_out=m_dot_3*h[3]

"Property data are given by:"

h[1] = enthalpy(R134a, T=T[1], P=P[1])h[2] = enthalpy(R134a, T=T[2], P=P[2])T[3] = temperature(R134a, P=P[3], h=h[3])x 3=QUALITY(R134a, h=h[3], P=P[3])

m _{frac}	T ₃	X 3
1	39.37	0.5467
1.25	39.37	0.467
1.5	39.37	0.4032
1.75	39.37	0.351
2	39.37	0.3075
2.25	39.37	0.2707
2.5	39.37	0.2392
2.75	39.37	0.2119
3	39.37	0.188
3.25	39.37	0.1668
3.5	39.37	0.1481
3.75	39.37	0.1313
4	39.37	0.1162



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5-80E Steam is condensed by cooling water in a condenser. The rate of heat transfer in the heat exchanger and the rate of condensation of steam are to be determined.

Assumptions **1** Steady operating conditions exist. **2** The heat exchanger is well-insulated so that heat loss to the surroundings is negligible and thus heat transfer from the hot fluid is equal to the heat transfer to the cold fluid. **3** Changes in the kinetic and potential energies of fluid streams are negligible. **4** Fluid properties are constant.

Properties The specific heat of water is 1.0 Btu/lbm.°F (Table A-3E). The enthalpy of vaporization of water at 85°F is 1045.2 Btu/lbm (Table A-4E).

Analysis We take the tube-side of the heat exchanger where cold water is flowing as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as





Then the rate of heat transfer to the cold water in this heat exchanger becomes

 $\dot{Q} = [\dot{m}c_p(T_{out} - T_{in})]_{water} = (138 \text{ lbm/s})(1.0 \text{ Btu/lbm.}^\circ\text{F})(73^\circ\text{F} - 60^\circ\text{F}) = 1794 \text{ Btu/s}$

Noting that heat gain by the water is equal to the heat loss by the condensing steam, the rate of condensation of the steam in the heat exchanger is determined from

$$\dot{Q} = (\dot{m}h_{fg})_{\text{steam}} = \longrightarrow \dot{m}_{\text{steam}} = \frac{Q}{h_{fg}} = \frac{1794 \text{ Btu/s}}{1045.2 \text{ Btu/lbm}} = 1.72 \text{ lbm/s}$$

5-81 Steam is condensed by cooling water in the condenser of a power plant. If the temperature rise of the cooling water is not to exceed 10°C, the minimum mass flow rate of the cooling water required is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 There are no work interactions. 4 Heat loss from the device to the surroundings is negligible and thus heat transfer from the hot fluid is equal to the heat transfer to the cold fluid. 5 Liquid water is an incompressible substance with constant specific heats at room temperature.

Properties The cooling water exists as compressed liquid at both states, and its specific heat at room temperature is c = 4.18kJ/kg.°C (Table A-3). The enthalpies of the steam at the inlet and the exit states are (Tables A-5 and A-6)

$$P_{3} = 20 \text{ kPa} \\ x_{3} = 0.95 \end{cases} h_{3} = h_{f} + x_{3}h_{fg} = 251.42 + 0.95 \times 2357.5 = 2491.1 \text{ kJ/kg}$$

$$P_{4} = 20 \text{ kPa} \\ \text{sat. liquid} \end{cases} h_{4} \cong h_{f@20 \text{ kPa}} = 251.42 \text{ kJ/kg}$$

Analysis We take the heat exchanger as the system, which is a control volume. The mass and energy balances for this steady-flow system can be expressed in the rate form as

Mass balance (for each fluid stream):

$$\dot{m}_{in} - \dot{m}_{out} = \Delta \dot{m}_{system}^{70 \text{ (steady)}} = 0$$

 $\dot{m}_{in} = \dot{m}_{out}$
 $\dot{m}_1 = \dot{m}_2 = \dot{m}_w \text{ and } \dot{m}_3 = \dot{m}_4 = \dot{m}_s$
alance (for the heat exchanger):

Energy be

⊅0 (steady) $\underline{\dot{E}_{in} - \dot{E}_{out}}_{Rate of flet energy transfer} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic,}$ = 0by heat, work, and mass potential, etc. energies $\dot{E}_{in} = \dot{E}_{out}$

$$\dot{m}_1 h_1 + \dot{m}_3 h_3 = \dot{m}_2 h_2 + \dot{m}_4 h_4$$
 (since $\dot{Q} = \dot{W} = \Delta \text{ke} \cong \Delta \text{pe} \cong 0$)

Combining the two,

$$\dot{m}_w (h_2 - h_1) = \dot{m}_s (h_3 - h_4)$$

Solving for \dot{m}_w :

$$\dot{m}_{w} = \frac{h_{3} - h_{4}}{h_{2} - h_{1}} \dot{m}_{s} \cong \frac{h_{3} - h_{4}}{c_{p} (T_{2} - T_{1})} \dot{m}_{s}$$

Substituting,

$$\dot{m}_w = \frac{(2491.1 - 251.42)\text{kJ/kg}}{(4.18 \text{ kJ/kg} \cdot^\circ \text{C})(10^\circ\text{C})} (20,000/3600 \text{ kg/s}) = 297.7 \text{ kg/s}$$



T

Assumptions **1** Steady operating conditions exist. **2** The heat exchanger is well-insulated so that heat loss to the surroundings is negligible and thus heat transfer from the hot fluid is equal to the heat transfer to the cold fluid. **3** Changes in the kinetic and potential energies of fluid streams are negligible. **4** Fluid properties are constant.

Properties The specific heats of water and ethylene glycol are given to be 4.18 and 2.56 kJ/kg.°C, respectively.

Analysis (*a*) We take the ethylene glycol tubes as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{\text{by heat, work, and mass}} = \underbrace{\Delta \dot{E}_{system}^{70 \text{ (steady)}}}_{\text{potential, etc. energies}} = 0$$

$$\dot{E}_{in} = \dot{E}_{out}$$

 $\dot{m}h_1 = \dot{Q}_{out} + \dot{m}h_2$ (since $\Delta ke \cong \Delta pe \cong 0$)
 $\dot{Q}_{out} = \dot{m}c_n(T_1 - T_2)$



Then the rate of heat transfer becomes

$$\dot{Q} = [\dot{m}c_p(T_{in} - T_{out})]_{glycol} = (3.2 \text{ kg/s})(2.56 \text{ kJ/kg.}^{\circ}\text{C})(80^{\circ}\text{C} - 40^{\circ}\text{C}) = 327.7 \text{ kW}$$

(b) The rate of heat transfer from glycol must be equal to the rate of heat transfer to the water. Then,

$$\dot{Q} = [\dot{m}c_p(T_{\text{out}} - T_{\text{in}})]_{\text{water}} \longrightarrow \dot{m}_{\text{water}} = \frac{\dot{Q}}{c_p(T_{\text{out}} - T_{\text{in}})} = \frac{327.7 \text{ kJ/s}}{(4.18 \text{ kJ/kg.}^\circ\text{C})(70^\circ\text{C} - 20^\circ\text{C})} = 1.57 \text{ kg/s}$$

5-83 Problem 5-82 is reconsidered. The effect of the inlet temperature of cooling water on the mass flow rate of water as the inlet temperature varies from 10°C to 40°C at constant exit temperature) is to be investigated. The mass flow rate of water is to be plotted against the inlet temperature.

Analysis The problem is solved using EES, and the solution is given below.

"Input Data" T_w[1]=20 [C] T_w[2]=70 [C] "w: water" m_dot_eg=2 [kg/s] "eg: ethylene glycol" T_eg[1]=80 [C] T_eg[2]=40 [C] C_p_w=4.18 [kJ/kg-K] C_p_eg=2.56 [kJ/kg-K]

"Conservation of mass for the water: m_dot_w_in=m_dot_w_out=m_dot_w" "Conservation of mass for the ethylene glycol: m_dot_eg_in=m_dot_eg_out=m_dot_eg"

"Conservation of Energy for steady-flow: neglect changes in KE and PE in each mass steam"
"We assume no heat transfer and no work occur across the control surface."
E_dot_in - E_dot_out = DELTAE_dot_cv
DELTAE_dot_cv=0 "Steady-flow requirement"
E_dot_in=m_dot_w*h_w[1] + m_dot_eg*h_eg[1]
E_dot_out=m_dot_w*h_w[2] + m_dot_eg*h_eg[2]
Q_exchanged =m_dot_eg*h_eg[1] - m_dot_eg*h_eg[2]

"Property data are given by:"





5-84 Oil is to be cooled by water in a thin-walled heat exchanger. The rate of heat transfer in the heat exchanger and the exit temperature of water is to be determined.

Assumptions 1 Steady operating conditions exist. 2 The heat exchanger is well-insulated so that heat loss to the surroundings is negligible and thus heat transfer from the hot fluid is equal to the heat transfer to the cold fluid. 3 Changes in the kinetic and potential energies of fluid streams are negligible. 4 Fluid properties are constant.

Properties The specific heats of water and oil are given to be 4.18 and 2.20 kJ/kg.°C, respectively.

Analysis We take the oil tubes as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as





Then the rate of heat transfer from the oil becomes

$$\dot{Q} = [\dot{m}c_p(T_{\rm in} - T_{\rm out})]_{\rm oil} = (2 \text{ kg/s})(2.2 \text{ kJ/kg.}^{\circ}\text{C})(150^{\circ}\text{C} - 40^{\circ}\text{C}) = 484 \text{ kW}$$

Noting that the heat lost by the oil is gained by the water, the outlet temperature of the water is determined from

$$\dot{Q} = [\dot{m}c_p(T_{\text{out}} - T_{\text{in}})]_{\text{water}} \longrightarrow T_{\text{out}} = T_{\text{in}} + \frac{\dot{Q}}{\dot{m}_{\text{water}}c_p} = 22^{\circ}\text{C} + \frac{484 \text{ kJ/s}}{(1.5 \text{ kg/s})(4.18 \text{ kJ/kg.}^{\circ}\text{C})} = 99.2^{\circ}\text{C}$$

5-85 Cold water is heated by hot water in a heat exchanger. The rate of heat transfer and the exit temperature of hot water are to be determined.

Assumptions 1 Steady operating conditions exist. 2 The heat exchanger is well-insulated so that heat loss to the surroundings is negligible and thus heat transfer from the hot fluid is equal to the heat transfer to the cold fluid. 3 Changes in the kinetic and potential energies of fluid streams are negligible. 4 Fluid properties are constant.

Properties The specific heats of cold and hot water are given to be 4.18 and 4.19 kJ/kg.°C, respectively.

Analysis We take the cold water tubes as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as



Then the rate of heat transfer to the cold water in this heat exchanger becomes

$$\dot{Q} = [\dot{m}c_p(T_{\text{out}} - T_{\text{in}})]_{\text{cold water}} = (0.60 \text{ kg/s})(4.18 \text{ kJ/kg.}^{\circ}\text{C})(45^{\circ}\text{C} - 15^{\circ}\text{C}) = 75.24 \text{ kW}$$

Noting that heat gain by the cold water is equal to the heat loss by the hot water, the outlet temperature of the hot water is determined to be

$$\dot{Q} = [\dot{m}c_p(T_{\rm in} - T_{\rm out})]_{\rm hot \ water} \longrightarrow T_{\rm out} = T_{\rm in} - \frac{Q}{\dot{m}c_p} = 100^{\circ}\text{C} - \frac{75.24 \text{ kW}}{(3 \text{ kg/s})(4.19 \text{ kJ/kg.}^{\circ}\text{C})} = 94.0^{\circ}\text{C}$$

Cold

5-86 Air is preheated by hot exhaust gases in a cross-flow heat exchanger. The rate of heat transfer and the outlet temperature of the air are to be determined.

Assumptions 1 Steady operating conditions exist. 2 The heat exchanger is well-insulated so that heat loss to the surroundings is negligible and thus heat transfer from the hot fluid is equal to the heat transfer to the cold fluid. 3 Changes in the kinetic and potential energies of fluid streams are negligible. 4 Fluid properties are constant.

Properties The specific heats of air and combustion gases are given to be 1.005 and 1.10 kJ/kg.°C, respectively.

Analysis We take the exhaust pipes as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Bate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0 \rightarrow \dot{E}_{in} = \dot{E}_{out} = \dot{h}_{out} + \dot{h}_{n} = \dot{Q}_{out} + \dot{h}_{n} \text{ (since } \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

$$\dot{Q}_{out} = \dot{m}c_{p}(T_{1} - T_{2})$$

Then the rate of heat transfer from the exhaust gases becomes

$$\dot{Q} = [\dot{m}c_p (T_{\rm in} - T_{\rm out})]_{\rm gas}$$

= (0.95 kg/s)(1.1 kJ/kg.°C)(160°C - 95°C)
= **67.93 kW**

The mass flow rate of air is

$$\dot{m} = \frac{P\dot{V}}{RT} = \frac{(95 \text{ kPa})(0.6 \text{ m}^3/\text{s})}{(0.287 \text{ kPa.m}^3/\text{kg.K}) \times 293 \text{ K}} = 0.6778 \text{ kg/s}$$

Noting that heat loss by the exhaust gases is equal to the heat gain by the air, the outlet temperature of the air becomes

$$\dot{Q} = \dot{m}c_p (T_{c,out} - T_{c,in}) \longrightarrow T_{c,out} = T_{c,in} + \frac{\dot{Q}}{\dot{m}c_p} = 20^{\circ}\text{C} + \frac{67.93 \text{ kW}}{(0.6778 \text{ kg/s})(1.005 \text{ kJ/kg.}^{\circ}\text{C})} = 120^{\circ}\text{C}$$



Exhaust gases 0.95 kg/s, 95°C

5-87E An adiabatic open feedwater heater mixes steam with feedwater. The outlet mass flow rate and the outlet velocity are to be determined for two exit temperatures.

Assumptions Steady operating conditions exist.

Analysis From a mass balance

$$\dot{m}_3 = \dot{m}_1 + \dot{m}_2 = 0.1 + 2 = 2.1$$
 lbm/s

The specific volume at the exit is (Table A-4E)

$$\left. \begin{array}{l} P_3 = 10 \text{ psia} \\ T_3 = 120^{\circ} \text{F} \end{array} \right\} \, \boldsymbol{v}_3 \cong \boldsymbol{v}_{f \, @\, 120^{\circ} \text{F}} = 0.01620 \text{ ft}^3 / \text{lbm} \end{array}$$

The exit velocity is then

$$V_3 = \frac{\dot{m}_3 v_3}{A_3} = \frac{4\dot{m}_3 v_3}{\pi D^2}$$
$$= \frac{4(2.1 \text{ lbm/s})(0.01620 \text{ ft}^3/\text{lbm})}{\pi (0.5 \text{ ft})^2}$$
$$= 0.1733 \text{ ft/s}$$

When the temperature at the exit is 180°F, we have

$$\dot{m}_{3} = \dot{m}_{1} + \dot{m}_{2} = 0.1 + 2 = 2.1 \text{ lbm/s}$$

$$P_{3} = 10 \text{ psia}$$

$$T_{3} = 180^{\circ}\text{F}$$

$$v_{3} \approx v_{f@180^{\circ}\text{F}} = 0.01651 \text{ ft}^{3}/\text{lbm}$$

$$V_{3} = \frac{\dot{m}_{3}v_{3}}{A_{3}} = \frac{4\dot{m}_{3}v_{3}}{\pi D^{2}} = \frac{4(2.1 \text{ lbm/s})(0.01651 \text{ ft}^{3}/\text{lbm})}{\pi (0.5 \text{ ft})^{2}} = 0.1766 \text{ ft/s}$$

The mass flow rate at the exit is same while the exit velocity slightly increases when the exit temperature is 180°F instead of 120°F.



5-88E Air is heated in a steam heating system. For specified flow rates, the volume flow rate of air at the inlet is to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Kinetic and potential energy changes are negligible. **3** There are no work interactions. **4** Heat loss from the device to the surroundings is negligible and thus heat transfer from the hot fluid is equal to the heat transfer to the cold fluid. **5** Air is an ideal gas with constant specific heats at room temperature.

Properties The gas constant of air is 0.3704 psia.ft³/lbm.R (Table A-1E). The constant pressure specific heat of air is $c_p = 0.240$ Btu/lbm.°F (Table A-2E). The enthalpies of steam at the inlet and the exit states are (Tables A-4E through A-6E)

$$\begin{array}{l} P_{3} = 30 \text{ psia} \\ T_{3} = 400^{\circ}\text{F} \end{array} \right\} h_{3} = 1237.9 \text{ Btu/lbm} \\ P_{4} = 25 \text{ psia} \\ T_{4} = 212^{\circ}\text{F} \end{array} \right\} h_{4} \cong h_{f@212^{\circ}\text{F}} = 180.21 \text{ Btu/lbm}$$

Analysis We take the entire heat exchanger as the system, which is a control volume. The mass and energy balances for this steady-flow system can be expressed in the rate form as

Mass balance (for each fluid stream):

$$\dot{m}_{\rm in} - \dot{m}_{\rm out} = \Delta \dot{m}_{\rm system} \overset{\text{ψ_0 (steady)}}{=} 0$$
$$\dot{m}_{\rm in} = \dot{m}_{\rm out}$$
$$\dot{m}_1 = \dot{m}_2 = \dot{m}_a \text{ and } \dot{m}_3 = \dot{m}_4 = \dot{m}_s$$

Energy balance (for the entire heat exchanger):

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{Bate of net energy transfer} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\dot{E}_{in}} = \dot{E}_{out}$$

$$\dot{m}_{1}h_{1} + \dot{m}_{3}h_{3} = \dot{m}_{2}h_{2} + \dot{m}_{4}h_{4} \quad (since \dot{Q} = \dot{W} = \Delta ke \cong \Delta pe \cong 0)$$

Combining the two,

$$\dot{m}_a(h_2 - h_1) = \dot{m}_s(h_3 - h_4)$$

Solving for \dot{m}_a :

$$\dot{m}_a = \frac{h_3 - h_4}{h_2 - h_1} \dot{m}_s \cong \frac{h_3 - h_4}{c_p (T_2 - T_1)} \dot{m}_s$$

Substituting,

$$\dot{m}_a = \frac{(1237.9 - 180.21)\text{Btu/lbm}}{(0.240 \text{ Btu/lbm} \cdot ^\circ\text{F})(130 - 80)^\circ\text{F}} (15 \text{ lbm/min}) = 1322 \text{ lbm/min} = 22.04 \text{ lbm/s}$$

Also,

$$v_1 = \frac{RT_1}{P_1} = \frac{(0.3704 \text{ psia} \cdot \text{ft}^3/\text{lbm} \cdot \text{R})(540 \text{ R})}{14.7 \text{ psia}} = 13.61 \text{ ft}^3/\text{lbm}$$

Then the volume flow rate of air at the inlet becomes

$$\dot{\mathbf{V}}_1 = \dot{m}_a \mathbf{v}_1 = (22.04 \text{ lbm/s})(13.61 \text{ ft}^3/\text{lbm}) = 300 \text{ ft}^3/\text{s}$$



$$\downarrow^2$$

Steam

AIR

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 There are no work interactions. 4 The device is adiabatic and thus heat transfer is negligible.

Properties The gas constant of air is R = 0.287 kPa.m³/kg.K. The enthalpies of air are obtained from air table (Table A-17) as

$$h_1 = h_{@280 \text{ K}} = 280.13 \text{ kJ/kg}$$

 $h_2 = h_{@307 \text{ K}} = 307.23 \text{ kJ/kg}$
 $h_{\text{room}} = h_{@297 \text{ K}} = 297.18 \text{ kJ/kg}$

Analysis (*a*) We take the mixing chamber as the system, which is a control volume since mass crosses the boundary. The mass and energy balances for this steady-flow system can be expressed in the rate form as



Mass balance:

$$\dot{m}_{\rm in} - \dot{m}_{\rm out} = \Delta \dot{m}_{\rm system}^{70 \text{ (steady)}} = 0 \rightarrow \dot{m}_{\rm in} = \dot{m}_{\rm out} \rightarrow \dot{m}_1 + 1.6 \dot{m}_1 = \dot{m}_3 = 2.6 \dot{m}_1 \text{ since } \dot{m}_2 = 1.6 \dot{m}_1$$

Energy balance:

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}^{\gamma_0 \text{ (steady)}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0$$

$$\underline{\dot{E}_{in}}_{\text{in}} = \dot{E}_{out}$$

$$\dot{m}_1 h_1 + \dot{m}_2 h_2 = \dot{m}_3 h_3 \quad (\text{since } \dot{Q} \cong \dot{W} \cong \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

Combining the two gives $\dot{m}_1h_1 + 2.2\dot{m}_1h_2 = 3.2\dot{m}_1h_3$ or $h_3 = (h_1 + 2.2h_2)/3.2$

Substituting,

$$h_3 = (280.13 + 2.2 \times 307.23)/3.2 = 298.76 \text{ kJ/kg}$$

From air table at this enthalpy, the mixture temperature is

$$T_3 = T_{@h=298.76 \text{ kJ/kg}} = 298.6 \text{ K} = 25.6^{\circ}\text{C}$$

(b) The mass flow rates are determined as follows

$$\boldsymbol{\nu}_{1} = \frac{RT_{1}}{P} = \frac{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(7 + 273 \text{ K})}{105 \text{ kPa}} = 0.7654 \text{ m}^{3}/\text{kg}$$
$$\dot{m}_{1} = \frac{\dot{\boldsymbol{\nu}}_{1}}{\boldsymbol{\nu}_{1}} = \frac{0.75 \text{ m}^{3}/\text{s}}{0.7654 \text{ m}^{3}/\text{kg}} = 0.9799 \text{ kg/s}$$
$$\dot{m}_{3} = 3.2\dot{m}_{1} = 3.2(0.9799 \text{ kg/s}) = 3.136 \text{ kg/s}$$

The rate of heat gain of the room is determined from

$$\dot{Q}_{gain} = \dot{m}_3(h_{room} - h_3) = (3.136 \text{ kg/s})(297.18 - 298.76) \text{ kJ/kg} = -4.93 \text{ kW}$$

The negative sign indicates that the room actually loses heat at a rate of 4.93 kW.

5-90 A heat exchanger that is not insulated is used to produce steam from the heat given up by the exhaust gases of an internal combustion engine. The temperature of exhaust gases at the heat exchanger exit and the rate of heat transfer to the water are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 There are no work interactions. 4 Exhaust gases are assumed to have air properties with constant specific heats.

Properties The constant pressure specific heat of the exhaust gases is taken to be $c_p = 1.045 \text{ kJ/kg} \cdot ^{\circ}\text{C}$ (Table A-2). The inlet and exit enthalpies of water are (Tables A-4 and A-5)

$$T_{w,in} = 15^{\circ}C$$

$$x = 0 \text{ (sat. liq.)}$$

$$h_{w,in} = 62.98 \text{ kJ/kg}$$

$$P_{w,out} = 2 \text{ MPa}$$

$$x = 1 \text{ (sat. vap.)}$$

$$h_{w,out} = 2798.3 \text{ kJ/kg}$$

Analysis We take the entire heat exchanger as the system, which is a control volume. The mass and energy balances for this steady-flow system can be expressed in the rate form as

Mass balance (for each fluid stream):

$$\dot{m}_{\rm in} - \dot{m}_{\rm out} = \Delta \dot{m}_{\rm system}^{70 \text{ (steady)}} = 0 \longrightarrow \dot{m}_{\rm in} = \dot{m}_{\rm out}$$

Energy balance (for the entire heat exchanger):

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{Rate of net energy transfer} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\frac{\dot{E}_{in} - \dot{E}_{out}}{\dot{E}_{in}} = \dot{E}_{out}$$

$$\dot{m}_{exh}h_{exh,in} + \dot{m}_{w}h_{w,in} = \dot{m}_{exh}h_{exh,out} + \dot{m}_{w}h_{w,out} + \dot{Q}_{out} \text{ (since } \dot{W} = \Delta ke \cong \Delta pe \cong 0)$$

or

Noting that the mass flow rate of exhaust gases is 15 times that of the water, substituting gives

$$15\dot{m}_{\rm w} (1.045 \text{ kJ/kg.}^{\circ}\text{C})(400^{\circ}\text{C}) + \dot{m}_{\rm w} (62.98 \text{ kJ/kg}) = 15\dot{m}_{\rm w} (1.045 \text{ kJ/kg.}^{\circ}\text{C})T_{\rm exh,out} + \dot{m}_{\rm w} (2798.3 \text{ kJ/kg}) + \dot{Q}_{out}$$
(1)

The heat given up by the exhaust gases and heat picked up by the water are

 $\dot{m}_{\text{exh}}c_pT_{\text{exh,in}} + \dot{m}_wh_{\text{w,in}} = \dot{m}_{\text{exh}}c_pT_{\text{exh,out}} + \dot{m}_wh_{\text{w,out}} + \dot{Q}_{\text{out}}$

$$\dot{Q}_{exh} = \dot{m}_{exh} c_p (T_{exh,in} - T_{exh,out}) = 15 \dot{m}_w (1.045 \text{ kJ/kg.}^\circ\text{C})(400 - T_{exh,out})^\circ\text{C}$$
 (2)

$$\dot{Q}_{w} = \dot{m}_{w}(h_{w,\text{out}} - h_{w,\text{in}}) = \dot{m}_{w}(2798.3 - 62.98)\text{kJ/kg}$$
 (3)

The heat loss is

$$\dot{Q}_{out} = f_{\text{heat loss}} \dot{Q}_{exh} = 0.1 \dot{Q}_{exh} \tag{4}$$

The solution may be obtained by a trial-error approach. Or, solving the above equations simultaneously using EES software, we obtain

$$T_{\text{exh,out}} = 206.1 \,^{\circ}\text{C}, \dot{Q}_{\text{w}} = 97.26 \,\text{kW}, \ \dot{m}_{\text{w}} = 0.03556 \,\text{kg/s}, \ \dot{m}_{\text{exh}} = 0.5333 \,\text{kg/s}$$





Water

15°C

exchanger

400°C

2 MPa

sat. vap.

5-91 A chilled-water heat-exchange unit is designed to cool air by water. The maximum water outlet temperature is to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Kinetic and potential energy changes are negligible. **3** There are no work interactions. **4** Heat loss from the device to the surroundings is negligible and thus heat transfer from the hot fluid is equal to the heat transfer to the cold fluid. **5** Air is an ideal gas with constant specific heats at room temperature.

Properties The gas constant of air is 0.287 kPa.m³/kg.K (Table A-1). The constant pressure specific heat of air at room temperature is $c_p = 1.005 \text{ kJ/kg} \cdot \text{°C}$ (Table A-2a). The specific heat of water is 4.18 kJ/kg·K (Table A-3).

Analysis The water temperature at the heat exchanger exit will be maximum when all the heat released by the air is picked up by the water. First, the inlet specific volume and the mass flow rate of air are

$$\boldsymbol{\nu}_{1} = \frac{RT_{1}}{P_{1}} = \frac{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(303 \text{ K})}{100 \text{ kPa}} = 0.8696 \text{ m}^{3}/\text{kg}$$
$$\dot{m}_{a} = \frac{\dot{\boldsymbol{\nu}}_{1}}{\boldsymbol{\nu}_{1}} = \frac{5 \text{ m}^{3}/\text{s}}{0.8696 \text{ m}^{3}/\text{kg}} = 5.750 \text{ kg/s}$$

We take the entire heat exchanger as the system, which is a control volume. The mass and energy balances for this steadyflow system can be expressed in the rate form as

Mass balance (for each fluid stream):

$$\dot{m}_{\rm in} - \dot{m}_{\rm out} = \Delta \dot{m}_{\rm system}^{70 \text{ (steady)}} = 0 \rightarrow \dot{m}_{\rm in} = \dot{m}_{\rm out} \rightarrow \dot{m}_1 = \dot{m}_3 = \dot{m}_a \text{ and } \dot{m}_2 = \dot{m}_4 = \dot{m}_w$$

Energy balance (for the entire heat exchanger):

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Bate of net energy transfer}} = \underbrace{\Delta \dot{E}_{system}}_{\text{Nate of change in internal, kinetic, potential, etc. energies}}^{70 (steady)} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{m}_1 h_1 + \dot{m}_2 h_2} = \dot{m}_3 h_3 + \dot{m}_4 h_4 \quad (\text{since } \dot{Q} = \dot{W} = \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

Combining the two,

$$\dot{m}_a(h_1 - h_3) = \dot{m}_w(h_4 - h_2)$$

 $\dot{m}_a c_{p,a}(T_1 - T_3) = \dot{m}_w c_{p,w}(T_4 - T_2)$

Solving for the exit temperature of water,

$$T_4 = T_2 + \frac{\dot{m}_a c_{p,a} (T_1 - T_3)}{\dot{m}_w c_{p,w}} = 8^{\circ}\text{C} + \frac{(5.750 \text{ kg/s})(1.005 \text{ kJ/kg} \cdot ^{\circ}\text{C})(30 - 18)^{\circ}\text{C}}{(2 \text{ kg/s})(4.18 \text{ kJ/kg} \cdot ^{\circ}\text{C})} = 16.3^{\circ}\text{C}$$

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 There are no work and heat interactions between the condenser and the surroundings.

Analysis We take the condenser as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as



If we take the refrigerant as the system, the energy balance can be written as

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{\text{Bate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}^{70 \text{ (steady)}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = \underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{m}_R h_1 = \dot{m}_R h_2 + \dot{Q}_{out}}$$

(a) The properties of refrigerant at the inlet and exit states of the condenser are (from Tables A-11 through A-13)

0

$$P_{1} = 1200 \text{ kPa} T_{1} = 85^{\circ}\text{C} \qquad h_{1} = 316.73 \text{ kJ/kg} P_{2} = 1200 \text{ kPa} T_{2} = T_{\text{sat}@1200 \text{ kPa}} - \Delta T_{\text{subcool}} = 46.3 - 6.3 = 40^{\circ}\text{C}$$

$$h_{2} \cong h_{f@.40^{\circ}\text{C}} = 108.26 \text{ kJ/kg}$$

The rate of heat rejected to the water is

$$\dot{Q}_{out} = \dot{m}_R (h_1 - h_2) = (0.042 \text{ kg/s})(316.73 - 108.26)\text{kJ/kg} = 8.76 \text{ kW} = 525 \text{ kJ/min}$$

(b) The mass flow rate of water can be determined from the energy balance on the condenser:

$$\dot{Q}_{out} = \dot{m}_w c_p \Delta T_w$$

8.76 kW = $\dot{m}_w (4.18 \text{ kJ/kg} \cdot ^\circ\text{C})(12^\circ\text{C})$
 $\dot{m}_w = 0.175 \text{ kg/s} = 10.5 \text{ kg/min}$

The specific heat of water is taken as 4.18 kJ/kg·°C (Table A-3).

5-93 Refrigerant-22 is evaporated in an evaporator by air. The rate of heat transfer from the air and the temperature change of air are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 There are no work and heat interactions between the evaporator and the surroundings.

Analysis We take the condenser as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as



If we take the refrigerant as the system, the energy balance can be written as

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{No (steady)}} = 0$$
Rate of change in internal, kinetic, potential, etc. energies
$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{m}_R h_1 + \dot{Q}_{in} = \dot{m}_R h_2$$

$$\dot{Q}_{in} = \dot{m}_R (h_2 - h_1)$$

(a) The mass flow rate of the refrigerant is

$$\dot{m}_R = \frac{\dot{V}_1}{v_1} = \frac{(2.25/3600) \text{ m}^3/\text{s}}{0.0253 \text{ m}^3/\text{kg}} = 0.02472 \text{ kg/s}$$

The rate of heat absorbed from the air is

$$\dot{Q}_{in} = \dot{m}_R (h_2 - h_1) = (0.02472 \text{ kg/s})(398.0 - 220.2) \text{kJ/kg} = 4.39 \text{ kW}$$

(b) The temperature change of air can be determined from an energy balance on the evaporator:

$$Q_L = \dot{m}_R (h_3 - h_2) = \dot{m}_a c_p (T_{a1} - T_{a2})$$

4.39 kW = (0.5 kg/s)(1.005 kJ/kg · °C) ΔT_a
 $\Delta T_a = 8.7$ °C

The specific heat of air is taken as 1.005 kJ/kg·°C (Table A-2).

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 There are no work interactions.

Analysis (a) We take the mixing device as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{by heat, work, and mass} = \underbrace{\Delta \dot{E}_{system}}_{Pot entral, etc. energies} \stackrel{?00 (steady)}{Pot entral, etc. energies} = 0$$

$$\dot{m}_1 h_1 + \dot{m}_2 h_2 + \dot{Q}_{\rm in} = \dot{m}_3 h_3$$

From a mass balance,

$$\dot{m}_3 = \dot{m}_1 + \dot{m}_2$$

Since
$$h = c_p T$$
,

Then

$$\dot{m}_{1}c_{p}T_{1} + \dot{m}_{2}c_{p}T_{2} + \dot{Q}_{\text{in}} = \dot{m}_{3}c_{p}T_{3}$$
$$T_{3} = \frac{\dot{m}_{1}}{\dot{m}_{3}}T_{1} + \frac{\dot{m}_{2}}{\dot{m}_{3}}T_{2} + \frac{\dot{Q}_{\text{in}}}{\dot{m}_{3}c_{p}}$$

(b) Expression for volume flow rate:

$$\dot{\boldsymbol{V}}_{3} = \dot{m}_{3}\boldsymbol{v}_{3} = \dot{m}_{3}\frac{RT_{3}}{P_{3}}$$
$$\dot{\boldsymbol{V}}_{3} = \frac{\dot{m}_{3}R}{P_{3}} \left(\frac{\dot{m}_{1}}{\dot{m}_{3}}T_{1} + \frac{\dot{m}_{2}}{\dot{m}_{3}}T_{2} + \frac{\dot{Q}_{\text{in}}}{\dot{m}_{3}c_{p}}\right)$$
$$P_{3} = P_{1} = P_{2} = P$$
$$\dot{\boldsymbol{V}}_{3} = \frac{\dot{m}_{1}RT_{1}}{P_{1}} + \frac{\dot{m}_{2}RT_{2}}{P_{2}} + \frac{R\dot{Q}_{\text{in}}}{P_{3}c_{p}}$$
$$\dot{\boldsymbol{V}}_{3} = \dot{\boldsymbol{V}}_{1} + \dot{\boldsymbol{V}}_{2} + \frac{R\dot{Q}_{\text{in}}}{Pc_{p}}$$

(c) If the process is adiabatic, then

$$\dot{\boldsymbol{V}}_3 = \dot{\boldsymbol{V}}_1 + \dot{\boldsymbol{V}}_2$$



Pipe and duct Flow

5-95 Heat is supplied to the argon as it flows in a heater. The exit temperature of argon and the volume flow rate at the exit are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible.

Properties The gas constant of argon is 0.2081 kPa.m³/kg.K. The constant pressure specific heat of air at room temperature is $c_p = 0.5203 \text{ kJ/kg} \cdot ^{\circ}\text{C}$ (Table A-2*a*).

Analysis (a) We take the pipe(heater) in which the argon is heated as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as



Substituting and solving for the exit temperature,

$$T_2 = T_1 + \frac{Q_{\text{in}}}{\dot{m}c_p} = 300 \text{ K} + \frac{150 \text{ kW}}{(6.24 \text{ kg/s})(0.5203 \text{ kJ/kg} \cdot \text{K})} = 346.2 \text{ K} = 73.2 \text{°C}$$

(b) The exit specific volume and the volume flow rate are

$$v_2 = \frac{RT_2}{P_2} = \frac{(0.2081 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(346.2 \text{ K})}{100 \text{ kPa}} = 0.7204 \text{ m}^3/\text{kg}$$

 $\dot{V}_2 = \dot{m}v_2 = (6.24 \text{ kg/s})(0.8266 \text{ m}^3/\text{kg}) = 4.50 \text{ m}^3/\text{s}$

5-96 Saturated liquid water is heated in a steam boiler at a specified rate. The rate of heat transfer in the boiler is to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Kinetic and potential energy changes are negligible. **3** There are no work interactions.

Analysis We take the pipe in which the water is heated as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as



The enthalpies of water at the inlet and exit of the boiler are (Table A-5, A-6).

$$\begin{array}{l} P_{1} = 2 \text{ MPa} \\ x = 0 \end{array} \right\} h_{1} \cong h_{f@2 \text{ MPa}} = 908.47 \text{ kJ/kg} \\ P_{2} = 2 \text{ MPa} \\ T_{2} = 250^{\circ}\text{C} \end{array} \right\} h_{2} = 2903.3 \text{ kJ/kg}$$

Substituting,

$$\dot{Q}_{in} = (4 \text{ kg/s})(2903.3 - 908.47)\text{kJ/kg} = 7980 \text{ kW}$$

5-97E Saturated liquid water is heated in a steam boiler. The heat transfer per unit mass is to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Kinetic and potential energy changes are negligible. **3** There are no work interactions.

Analysis We take the pipe in which the water is heated as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as



The enthalpies of water at the inlet and exit of the boiler are (Table A-5E, A-6E).

$$\begin{array}{l} P_1 = 500 \text{ psia} \\ x = 0 \end{array} \right\} h_1 \cong h_{f@\,500 \text{ psia}} = 449.51 \text{ Btu/lbm} \\ P_2 = 500 \text{ psia} \\ T_2 = 600^\circ \text{F} \end{array} \right\} h_2 = 1298.6 \text{ Btu/lbm}$$

Substituting,

$$q_{\rm in} = 1298.6 - 449.51 = 849.1 \,\mathrm{Btu/lbm}$$

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5-98 Air at a specified rate is heated by an electrical heater. The current is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 The heat losses from the air is negligible.

Properties The gas constant of air is 0.287 kPa.m³/kg.K (Table A-1). The constant pressure specific heat of air at room temperature is $c_p = 1.005 \text{ kJ/kg} \cdot ^{\circ}\text{C}$ (Table A-2*a*).

Analysis We take the pipe in which the air is heated as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as



The inlet specific volume and the mass flow rate of air are

$$\boldsymbol{\nu}_{1} = \frac{RT_{1}}{P_{1}} = \frac{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(288 \text{ K})}{100 \text{ kPa}} = 0.8266 \text{ m}^{3}/\text{kg}$$
$$\dot{m} = \frac{\dot{\boldsymbol{\nu}}_{1}}{\boldsymbol{\nu}_{1}} = \frac{0.3 \text{ m}^{3}/\text{s}}{0.8266 \text{ m}^{3}/\text{kg}} = 0.3629 \text{ kg/s}$$

Substituting into the energy balance equation and solving for the current gives

$$I = \frac{\dot{mc}_{p}(T_{2} - T_{1})}{\mathbf{V}} = \frac{(0.3629 \text{ kg/s})(1.005 \text{ kJ/kg} \cdot \text{K})(30 - 15)\text{K}}{110 \text{ V}} \left(\frac{1000 \text{ VI}}{1 \text{ kJ/s}}\right) = 49.7 \text{ Amperes}$$
5-99E The cooling fan of a computer draws air, which is heated in the computer by absorbing the heat of PC circuits. The electrical power dissipated by the circuits is to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Kinetic and potential energy changes are negligible. **3** All the heat dissipated by the circuits are picked up by the air drawn by the fan.

Properties The gas constant of air is 0.3704 psia·ft³/lbm·R (Table A-1E). The constant pressure specific heat of air at room temperature is $c_p = 0.240$ Btu/lbm·°F (Table A-2Ea).

Analysis We take the pipe in which the air is heated as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as



The inlet specific volume and the mass flow rate of air are

$$\boldsymbol{v}_{1} = \frac{RT_{1}}{P_{1}} = \frac{(0.3704 \text{ psia} \cdot \text{ft}^{3}/\text{lbm} \cdot \text{R})(530 \text{ R})}{14.7 \text{ psia}} = 13.35 \text{ ft}^{3}/\text{lbm}$$
$$\dot{\boldsymbol{m}} = \frac{\dot{\boldsymbol{V}}_{1}}{\boldsymbol{v}_{1}} = \frac{0.5 \text{ ft}^{3}/\text{s}}{13.35 \text{ ft}^{3}/\text{lbm}} = 0.03745 \text{ lbm/s}$$

Substituting,

$$\dot{W}_{e,out} = (0.03745 \text{ lbm/s})(0.240 \text{ Btu/lbm} \cdot \text{R})(80 - 70) \text{Btu/lbm} \left(\frac{1 \text{ kW}}{0.94782 \text{ Btu/s}}\right) = 0.0948 \text{ kW}$$

5-100 A desktop computer is to be cooled safely by a fan in hot environments and high elevations. The air flow rate of the fan and the diameter of the casing are to be determined.

Assumptions 1 Steady operation under worst conditions is considered. 2 Air is an ideal gas with constant specific heats. 3 Kinetic and potential energy changes are negligible.

Properties The specific heat of air at the average temperature of $T_{avg} = (45+60)/2 = 52.5$ °C = 325.5 K is $c_p = 1.0065$ kJ/kg.°C. The gas constant for air is R = 0.287 kJ/kg.K (Table A-2).

Analysis The fan selected must be able to meet the cooling requirements of the computer at worst conditions. Therefore, we assume air to enter the computer at 66.63 kPa and 45°C, and leave at 60°C.

We take the air space in the computer as the system, which is a control volume. The energy balance for this steadyflow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}^{70 \text{ (steady)}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{Q}_{in} + \dot{m}h_1 = \dot{m}h_2} \text{ (since } \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

$$\underbrace{\dot{Q}_{in} = \dot{m}c_p(T_2 - T_1)}$$



Then the required mass flow rate of air to absorb heat at a rate of 60 W is determined to be

$$\dot{Q} = \dot{m}c_p (T_{\text{out}} - T_{\text{in}}) \rightarrow \dot{m} = \frac{\dot{Q}}{c_p (T_{\text{out}} - T_{\text{in}})} = \frac{60 \text{ W}}{(1006.5 \text{ J/kg.}^\circ\text{C})(60-45)^\circ\text{C}} = 0.00397 \text{ kg/s} = 0.238 \text{ kg/min}$$

The density of air entering the fan at the exit and its volume flow rate are

$$\rho = \frac{P}{RT} = \frac{66.63 \text{ kPa}}{(0.287 \text{ kPa.m}^3/\text{kg.K})(60 + 273)\text{K}} = 0.6972 \text{ kg/m}^3$$
$$\dot{\mathbf{V}} = \frac{\dot{m}}{\rho} = \frac{0.238 \text{ kg/min}}{0.6972 \text{ kg/m}^3} = \mathbf{0.341 m^3/min}$$

For an average exit velocity of 110 m/min, the diameter of the casing of the fan is determined from

$$\dot{\mathbf{V}} = A_c V = \frac{\pi D^2}{4} V \rightarrow D = \sqrt{\frac{4\dot{\mathbf{V}}}{\pi V}} = \sqrt{\frac{(4)(0.341 \,\mathrm{m}^3/\mathrm{min})}{\pi (110 \,\mathrm{m/min})}} = 0.063 \,\mathrm{m} = \mathbf{6.3 \,\mathrm{cm}}$$

5-101 A desktop computer is to be cooled safely by a fan in hot environments and high elevations. The air flow rate of the fan and the diameter of the casing are to be determined.

Assumptions 1 Steady operation under worst conditions is considered. 2 Air is an ideal gas with constant specific heats. 3 Kinetic and potential energy changes are negligible.

Properties The specific heat of air at the average temperature of $T_{ave} = (45+60)/2 = 52.5^{\circ}$ C is $c_p = 1.0065$ kJ/kg.°C The gas constant for air is R = 0.287 kJ/kg.K (Table A-2).

Analysis The fan selected must be able to meet the cooling requirements of the computer at worst conditions. Therefore, we assume air to enter the computer at 66.63 kPa and 45°C, and leave at 60°C.

We take the air space in the computer as the system, which is a control volume. The energy balance for this steadyflow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{Bate of net energy transfer} = \underbrace{\Delta \dot{E}_{system}^{70 \text{ (steady)}}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{Q}_{in} + \dot{m}h_{1} = \dot{m}h_{2}} \text{ (since } \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

$$\underbrace{\dot{Q}_{in} = \dot{m}c_{p}(T_{2} - T_{1})}$$



Then the required mass flow rate of air to absorb heat at a rate of 100 W is determined to be

$$\dot{Q} = \dot{m}c_{p}(T_{\text{out}} - T_{\text{in}})$$
$$\dot{m} = \frac{\dot{Q}}{c_{p}(T_{\text{out}} - T_{\text{in}})} = \frac{100 \text{ W}}{(1006.5 \text{ J/kg.}^{\circ}\text{C})(60 - 45)^{\circ}\text{C}} = 0.006624 \text{ kg/s} = 0.397 \text{ kg/min}$$

The density of air entering the fan at the exit and its volume flow rate are

$$\rho = \frac{P}{RT} = \frac{66.63 \text{ kPa}}{(0.287 \text{ kPa.m}^3/\text{kg.K})(60 + 273)\text{K}} = 0.6972 \text{ kg/m}^3$$
$$\dot{\mathcal{V}} = \frac{\dot{m}}{\rho} = \frac{0.397 \text{ kg/min}}{0.6972 \text{ kg/m}^3} = 0.57 \text{ m}^3/\text{min}$$

For an average exit velocity of 110 m/min, the diameter of the casing of the fan is determined from

$$\dot{\mathbf{V}} = A_c V = \frac{\pi D^2}{4} V \longrightarrow D = \sqrt{\frac{4\dot{\mathbf{V}}}{\pi V}} = \sqrt{\frac{(4)(0.57 \text{ m}^3/\text{min})}{\pi (110 \text{ m/min})}} = 0.081 \text{ m} = \mathbf{8.1 \text{ cm}}$$

Rate

Assumptions 1 Steady operating conditions exist. 2 About 15 percent of the heat generated is dissipated from the components to the surroundings by convection and radiation. **3** Kinetic and potential energy changes are negligible.

Properties The properties of water at room temperature are $\rho = 62.1 \text{ lbm/ft}^3$ and $c_p = 1.00 \text{ Btu/lbm.}^\circ\text{F}$ (Table A-3E).

Analysis We take the tubes of the cold plate to be the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{by heat, work, and mass} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\underline{\dot{E}_{in}}_{bin} = \underline{\dot{E}_{out}}$$

$$\underline{\dot{Q}_{in}}_{in} + \underline{\dot{m}h_1} = \underline{\dot{m}h_2} \quad (since \ \Delta ke \cong \Delta pe \cong 0)$$

$$\underline{\dot{Q}_{in}} = \underline{\dot{m}c_p}(T_2 - T_1)$$



Then mass flow rate of water and the rate of heat removal by the water are determined to be

$$\dot{m} = \rho AV = \rho \frac{\pi D^2}{4} V = (62.1 \text{ lbm/ft}^3) \frac{\pi (0.25/12 \text{ ft})^2}{4} (40 \text{ ft/min}) = 0.8483 \text{ lbm/min} = 50.9 \text{ lbm/h}$$
$$\dot{Q} = \dot{m}c_p (T_{\text{out}} - T_{\text{in}}) = (50.9 \text{ lbm/h})(1.00 \text{ Btu/lbm.°F})(105 - 70)^{\circ}\text{F} = 1781 \text{ Btu/h}$$

which is 85 percent of the heat generated by the electronic devices. Then the total amount of heat generated by the electronic devices becomes

$$\dot{Q} = \frac{1781 \operatorname{Btu/h}}{0.85} = 2096 \operatorname{Btu/h} = 614 \operatorname{W}$$

5-103 The components of an electronic device located in a horizontal duct of rectangular cross section are cooled by forced air. The heat transfer from the outer surfaces of the duct is to be determined.

Assumptions 1 Steady operating conditions exist. 2 Air is an ideal gas with constant specific heats at room temperature. 3 Kinetic and potential energy changes are negligible

Properties The gas constant of air is R = 0.287 kJ/kg.°C (Table A-1). The specific heat of air at room temperature is $c_p = 1.005$ kJ/kg.°C (Table A-2).

Analysis The density of air entering the duct and the mass flow rate are

$$\rho = \frac{P}{RT} = \frac{101.325 \text{ kPa}}{(0.287 \text{ kPa.m}^3/\text{kg.K})(30 + 273)\text{K}} = 1.165 \text{ kg/m}^3$$
$$\dot{m} = \rho \dot{\mathbf{V}} = (1.165 \text{ kg/m}^3)(0.6 \text{ m}^3 / \text{min}) = 0.700 \text{ kg/min}$$

We take the channel, excluding the electronic components, to be the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Bate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{70 \text{ (steady)}} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{Q}_{in} + \dot{m}h_1 = \dot{m}h_2} \text{ (since } \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

$$\underbrace{\dot{Q}_{in} = \dot{m}c_p(T_2 - T_1)}^{70 \text{ (steady)}}$$



Then the rate of heat transfer to the air passing through the duct becomes

$$\dot{Q}_{air} = [\dot{m}c_p(T_{out} - T_{in})]_{air} = (0.700/60 \text{ kg/s})(1.005 \text{ kJ/kg.}^{\circ}\text{C})(40 - 30)^{\circ}\text{C} = 0.117 \text{ kW} = 117 \text{ W}$$

The rest of the 180 W heat generated must be dissipated through the outer surfaces of the duct by natural convection and radiation,

$$\dot{Q}_{\text{external}} = \dot{Q}_{\text{total}} - \dot{Q}_{\text{internal}} = 180 - 117 = 63 \text{ W}$$

5-104 The components of an electronic device located in a horizontal duct of circular cross section is cooled by forced air. The heat transfer from the outer surfaces of the duct is to be determined.

Assumptions 1 Steady operating conditions exist. 2 Air is an ideal gas with constant specific heats at room temperature. 3 Kinetic and potential energy changes are negligible

Properties The gas constant of air is R = 0.287 kJ/kg.°C (Table A-1). The specific heat of air at room temperature is $c_p = 1.005$ kJ/kg.°C (Table A-2).

Analysis The density of air entering the duct and the mass flow rate are

$$\rho = \frac{P}{RT} = \frac{101.325 \text{ kPa}}{(0.287 \text{ kPa.m}^3/\text{kg.K})(30+273)\text{K}} = 1.165 \text{ kg/m}^3$$
$$\dot{m} = \rho \dot{\mathbf{V}} = (1.165 \text{ kg/m}^3)(0.6 \text{ m}^3 / \text{min}) = 0.700 \text{ kg/min}$$

We take the channel, excluding the electronic components, to be the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as





Then the rate of heat transfer to the air passing through the duct becomes

$$Q_{\text{air}} = [\dot{m}c_p(T_{\text{out}} - T_{\text{in}})]_{\text{air}} = (0.700/60 \text{ kg/s})(1.005 \text{ kJ/kg.}^{\circ}\text{C})(40 - 30)^{\circ}\text{C} = 0.117 \text{ kW} = 117 \text{ W}$$

The rest of the 180 W heat generated must be dissipated through the outer surfaces of the duct by natural convection and radiation,

$$\dot{Q}_{\text{external}} = \dot{Q}_{\text{total}} - \dot{Q}_{\text{internal}} = 180 - 117 = 63 \text{ W}$$

5-105 Air enters a hollow-core printed circuit board. The exit temperature of the air is to be determined.

Assumptions 1 Steady operating conditions exist. 2 Air is an ideal gas with constant specific heats at room temperature. 3 The local atmospheric pressure is 1 atm. 4 Kinetic and potential energy changes are negligible.

Properties The gas constant of air is R = 0.287 kJ/kg.°C (Table A-1). The specific heat of air at room temperature is $c_p = 1.005$ kJ/kg.°C (Table A-2).

Analysis The density of air entering the duct and the mass flow rate are

$$\rho = \frac{P}{RT} = \frac{101.325 \text{ kPa}}{(0.287 \text{ kPa.m}^3/\text{kg.K})(25 + 273)\text{K}} = 1.185 \text{ kg/m}$$
$$\dot{m} = \rho \dot{\mathbf{V}} = (1.185 \text{ kg/m}^3)(0.0008 \text{ m}^3/\text{s}) = 0.0009477 \text{ kg/s}$$

We take the hollow core to be the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{Bate of net energy transfer} = \Delta \dot{E}_{system}^{70 (steady)} = 0$$
Rate of net energy transfer by heat, work, and mass $\dot{E}_{in} = \dot{E}_{out}$

$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{Q}_{in} + \dot{m}h_{1} = \dot{m}h_{2} \quad (since \Delta ke \cong \Delta pe \cong 0)$$

$$\dot{Q}_{in} = \dot{m}c_{p}(T_{2} - T_{1})$$

Then the exit temperature of air leaving the hollow core becomes

$$\dot{Q}_{in} = \dot{m}c_p (T_2 - T_1) \rightarrow T_2 = T_1 + \frac{Q_{in}}{\dot{m}c_p} = 25 \text{ °C} + \frac{15 \text{ J/s}}{(0.0009477 \text{ kg/s})(1005 \text{ J/kg. °C})} = 46.0 \text{ °C}$$



5-106 A computer is cooled by a fan blowing air through the case of the computer. The required flow rate of the air and the fraction of the temperature rise of air that is due to heat generated by the fan are to be determined.

Assumptions 1 Steady flow conditions exist. 2 Air is an ideal gas with constant specific heats. 3 The pressure of air is 1 atm. 4 Kinetic and potential energy changes are negligible

Properties The specific heat of air at room temperature is $c_p = 1.005 \text{ kJ/kg.}^{\circ}\text{C}$ (Table A-2).

Analysis (*a*) We take the air space in the computer as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{\pi_0 \text{ (steady)}} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{Q}_{in} + \dot{W}_{in} + \dot{m}h_1 = \dot{m}h_2} \text{ (since } \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

$$\underbrace{\dot{Q}_{in} + \dot{W}_{in} = \dot{m}c_p(T_2 - T_1)}$$



Noting that the fan power is 25 W and the 8 PCBs transfer a total of 80 W of heat to air, the mass flow rate of air is determined to be

$$\dot{Q}_{\rm in} + \dot{W}_{\rm in} = \dot{m}c_p(T_e - T_i) \rightarrow \dot{m} = \frac{Q_{\rm in} + \dot{W}_{\rm in}}{c_p(T_e - T_i)} = \frac{(8 \times 10) \,\mathrm{W} + 25 \,\mathrm{W}}{(1005 \,\mathrm{J/kg.^\circ C})(10^\circ \mathrm{C})} = 0.0104 \,\mathrm{kg/s}$$

(b) The fraction of temperature rise of air that is due to the heat generated by the fan and its motor can be determined from

$$\dot{Q} = \dot{m}c_p \Delta T \rightarrow \Delta T = \frac{\dot{Q}}{\dot{m}c_p} = \frac{25 \text{ W}}{(0.0104 \text{ kg/s})(1005 \text{ J/kg.}^\circ\text{C})} = 2.4^\circ\text{C}$$

 $f = \frac{2.4^\circ\text{C}}{10^\circ\text{C}} = 0.24 = 24\%$

5-107 Hot water enters a pipe whose outer surface is exposed to cold air in a basement. The rate of heat loss from the water is to be determined.

Assumptions 1 Steady flow conditions exist. 2 Water is an incompressible substance with constant specific heats. 3 The changes in kinetic and potential energies are negligible.

Properties The properties of water at the average temperature of $(90+88)/2 = 89^{\circ}$ C are $\rho = 965 \text{ kg/m}^3$ and $c_p = 4.21 \text{ kJ/kg.}^{\circ}$ C (Table A-3).

Analysis The mass flow rate of water is

$$\dot{m} = \rho A_c V = (965 \text{ kg/m}^3) \frac{\pi (0.025 \text{ m})^2}{4} (0.6 \text{ m/s}) = 0.2842 \text{ kg/s}$$

We take the section of the pipe in the basement to be the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as





Then the rate of heat transfer from the hot water to the surrounding air becomes

 $\dot{Q}_{out} = \dot{m}c_p [T_{in} - T_{out}]_{water} = (0.2842 \text{ kg/s})(4.21 \text{ kJ/kg.}^{\circ}\text{C})(90 - 88)^{\circ}\text{C} = 2.39 \text{ kW}$

5-108 Problem 5-107 is reconsidered. The effect of the inner pipe diameter on the rate of heat loss as the pipe diameter varies from 1.5 cm to 7.5 cm is to be investigated. The rate of heat loss is to be plotted against the diameter.

Analysis The problem is solved using EES, and the solution is given below.

"Knowns:" D = 0.025 [m] rho = 965 [kg/m^3] Vel = 0.6 [m/s] T_1 = 90 [C] T_2 = 88 [C] c_p = 4.21 [kJ/kg-C]

"Analysis:" "The mass flow rate of water is:" Area = pi*D^2/4 m_dot = rho*Area*Vel

"We take the section of the pipe in the basement to be the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as" $E_dot_in - E_dot_out = DELTAE_dot_sys$ DELTAE_dot_sys = 0 "Steady-flow assumption" $E_dot_in = m_dot*h_in$ $E_dot_out = Q_dot_out+m_dot*h_out$ $h_in = c_p * T_1$ $h_out = c_p * T_2$

D	Q _{out}
[m]	[kW]
0.015	1.149
0.025	3.191
0.035	6.254
0.045	10.34
0.055	15.44
0.065	21.57
0.075	28.72



Assumptions **1** Steady operating conditions exist. **2** Air is an ideal gas with constant specific heats at room temperature. **3** Kinetic and potential energy changes are negligible. **4** The heating duct is adiabatic, and thus heat transfer through it is negligible. **5** No air leaks in and out of the room.

Properties The gas constant of air is 0.287 kPa.m³/kg.K (Table A-1). The specific heats of air at room temperature are $c_p = 1.005$ and $c_v = 0.718$ kJ/kg·K (Table A-2).

Analysis (a) The total mass of air in the room is

$$V = 5 \times 6 \times 8 \text{ m}^3 = 240 \text{ m}^3$$

 $m = \frac{P_1 V}{RT_1} = \frac{(98 \text{ kPa})(240 \text{ m}^3)}{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(288 \text{ K})} = 284.6 \text{ kg}$

We first take the *entire room* as our system, which is a closed system since no mass leaks in or out. The power rating of the electric heater is determined by applying the conservation of energy relation to this constant volume closed system:

$$\underbrace{E_{in} - E_{out}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}}$$
$$W_{e,in} + W_{fan,in} - Q_{out} = \Delta U \quad (\text{since } \Delta \text{KE} = \Delta \text{PE} = 0)$$
$$\Delta t \left(\dot{W}_{e,in} + \dot{W}_{fan,in} - \dot{Q}_{out} \right) = mc_{\nu,avg} (T_2 - T_1)$$

Solving for the electrical work input gives

$$\dot{W}_{e,in} = \dot{Q}_{out} - \dot{W}_{fan,in} + mc_v (T_2 - T_1) / \Delta t$$

= (200/60 kJ/s) - (0.2 kJ/s) + (284.6 kg)(0.718 kJ/kg.°C)(25 - 15)°C/(15 × 60 s)
= **5.40 kW**

(b) We now take the *heating duct* as the system, which is a control volume since mass crosses the boundary. There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. The energy balance for this adiabatic steady-flow system can be expressed in the rate form as

$$\dot{E}_{in} - \dot{E}_{out}$$
Rate of net energy transfer
by heat, work, and mass
$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{W}_{e,in} + \dot{W}_{fan,in} + \dot{m}h_1 = \dot{m}h_2 \quad (\text{since } \dot{Q} = \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

$$\dot{W}_{e,in} + \dot{W}_{fan,in} = \dot{m}(h_2 - h_1) = \dot{m}c_p(T_2 - T_1)$$

Thus,

$$\Delta T = T_2 - T_1 = \frac{W_{e,in} + W_{fan,in}}{\dot{m}c_p} = \frac{(5.40 + 0.2) \text{ kJ/s}}{(50/60 \text{ kg/s})(1.005 \text{ kJ/kg} \cdot \text{K})} = 6.7^{\circ} \text{C}$$



5-110E Water is heated in a parabolic solar collector. The required length of parabolic collector is to be determined.

Assumptions 1 Steady operating conditions exist. 2 Heat loss from the tube is negligible so that the entire solar energy incident on the tube is transferred to the water. 3 Kinetic and potential energy changes are negligible

Properties The specific heat of water at room temperature is $c_p = 1.00$ Btu/lbm.°F (Table A-3E).

Analysis We take the thin aluminum tube to be the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as

$$\frac{\dot{E}_{in} - \dot{E}_{out}}{\dot{E}_{in} - \dot{E}_{out}} = \underbrace{\Delta \dot{E}_{system}}_{Rate of net energy transfer} = 0$$
Rate of net energy transfer

$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{Q}_{in} + \dot{m}h_1 = \dot{m}h_2 \quad (since \Delta ke \cong \Delta pe \cong 0)$$

$$\dot{Q}_{in} = \dot{m}_{water}c_p(T_2 - T_1)$$

$$\frac{\dot{E}_{in} - \dot{E}_{out}}{1}$$

$$\frac{\dot{D}_{in} + \dot{m}h_1 = \dot{m}h_2 \quad (since \Delta ke \cong \Delta pe \cong 0)$$

$$\dot{Q}_{in} = \dot{m}_{water}c_p(T_2 - T_1)$$

$$\frac{\dot{D}_{in} + \dot{D}_{in} + \dot{D}_{in}}{1}$$

$$\frac{\dot{D}_{in} + \dot{D}_{in}}{1}$$

Then the total rate of heat transfer to the water flowing through the tube becomes

$$\dot{Q}_{\text{total}} = \dot{m}c_p (T_e - T_i) = (4 \text{ lbm/s})(1.00 \text{ Btu/lbm.}^{\circ}\text{F})(180 - 55)^{\circ}\text{F} = 500 \text{ Btu/s} = 1,800,000 \text{ Btu/h}$$

The length of the tube required is

$$L = \frac{Q_{\text{total}}}{\dot{Q}} = \frac{1,800,000 \text{ Btu/h}}{400 \text{ Btu/h.ft}} = 4500 \text{ ft}$$

5-111 A house is heated by an electric resistance heater placed in a duct. The power rating of the electric heater is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Air is an ideal gas with constant specific heats at room temperature. 3 Kinetic and potential energy changes are negligible.

Properties The constant pressure specific heat of air at room temperature is $c_p = 1.005 \text{ kJ/kg·K}$ (Table A-2)

Analysis We take the *heating duct* as the system, which is a control volume since mass crosses the boundary. There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{70 \text{ (steady)}} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{E}_{e,in}} + \dot{W}_{fan,in} + \dot{m}h_1 = \dot{Q}_{out} + \dot{m}h_2 \text{ (since } \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

$$\underbrace{\dot{W}_{e,in} + \dot{W}_{fan,in}}_{e,in} = \dot{Q}_{out} + \dot{m}(h_2 - h_1) = \dot{Q}_{out} + \dot{m}c_p(T_2 - T_1)$$



Substituting, the power rating of the heating element is determined to be

 $\dot{W}_{e,in} = \dot{Q}_{out} + \dot{m}c_p\Delta T - \dot{W}_{fan,in} = (0.3 \text{ kJ/s}) + (0.6 \text{ kg/s})(1.005 \text{ kJ/kg} \cdot ^\circ\text{C})(7^\circ\text{C}) - 0.3 \text{ kW} = 4.22 \text{ kW}$

5-112 Steam pipes pass through an unheated area, and the temperature of steam drops as a result of heat losses. The mass flow rate of steam and the rate of heat loss from are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 4 There are no work interactions involved.

Properties From the steam tables (Table A-6),

$$P_{1} = 2 \text{ MPa} \quad v_{1} = 0.12551 \text{ m}^{3}/\text{kg}$$

$$T_{1} = 300^{\circ}\text{C} \quad h_{1} = 3024.2 \text{ kJ/kg}$$

$$P_{2} = 800 \text{ kPa} \quad h_{2} = 2950.4 \text{ kJ/kg}$$

$$T_{2} = 250^{\circ}\text{C} \quad h_{2} = 2950.4 \text{ kJ/kg}$$

$$Q$$

$$STEAM \quad 800 \text{ kPa} \quad 250^{\circ}\text{C} \quad V \quad V \quad 250^{\circ}\text{C}$$

Analysis (a) The mass flow rate of steam is determined directly from

$$\dot{m} = \frac{1}{\nu_1} A_1 V_1 = \frac{1}{0.12551 \text{ m}^3/\text{kg}} \left[\pi (0.06 \text{ m})^2 \right] (3 \text{ m/s}) = 0.270 \text{ kg/s}$$

(b) We take the *steam pipe* as the system, which is a control volume since mass crosses the boundary. There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{\pi 0 \text{ (steady)}} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{m}h_1 = \dot{Q}_{out} + \dot{m}h_2} \quad (\text{since } \dot{W} \cong \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

$$\dot{Q}_{out} = \dot{m}(h_1 - h_2)$$

Substituting, the rate of heat loss is determined to be

$$Q_{\text{loss}} = (0.270 \text{ kg/s})(3024.2 - 2950.4) \text{ kJ/kg} = 19.9 \text{ kJ/s}$$

5-113 Steam flows through a non-constant cross-section pipe. The inlet and exit velocities of the steam are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Potential energy change is negligible. 3 There are no work interactions. 4 The device is adiabatic and thus heat transfer is negligible.

Analysis We take the pipe as the system, which is a control volume since mass crosses the boundary. The mass and energy balances for this steady-flow system can be expressed in the rate form as

 $\begin{array}{ccc}
D_1 & & D_2 \\
200 \text{ kPa} & & 150 \text{ kPa} \\
200^{\circ}\text{C} & & & 150^{\circ}\text{C} \\
V_1 & & & V_2
\end{array}$

Mass balance:

 $\dot{m}_{\rm in} - \dot{m}_{\rm out} = \Delta \dot{m}_{\rm system}^{70 \text{ (steady)}} = 0$ $\dot{m}_{\rm in} = \dot{m}_{\rm out} \longrightarrow A_1 \frac{V_1}{\upsilon_1} = A_1 \frac{V_1}{\upsilon_1} \longrightarrow \frac{\pi D_1^2}{4} \frac{V_1}{\upsilon_1} = \frac{\pi D_2^2}{4} \frac{V_2}{\upsilon_2}$

Energy balance:

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{70 \text{ (steady)}} = 0$$

$$\underline{\dot{E}_{in}}_{\text{in}} = \dot{E}_{out}$$

$$h_1 + \frac{V_1^2}{2} = h_2 + \frac{V_2^2}{2} \quad (\text{since } \dot{Q} \cong \dot{W} \cong \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

The properties of steam at the inlet and exit are (Table A-6)

$$P_{1} = 200 \text{ kPa} \left\{ \boldsymbol{v}_{1} = 1.0805 \text{ m}^{3}/\text{kg} \right.$$
$$T_{1} = 200^{\circ}\text{C} \quad \int h_{1} = 2870.7 \text{ kJ/kg}$$
$$P_{2} = 150 \text{ kPa} \left\{ \boldsymbol{v}_{2} = 1.2855 \text{ m}^{3}/\text{kg} \right.$$
$$T_{1} = 150^{\circ}\text{C} \quad \int h_{2} = 2772.9 \text{ kJ/kg}$$

.

Assuming inlet diameter to be 1.8 m and the exit diameter to be 1.0 m, and substituting,

$$\frac{\pi (1.8 \text{ m})^2}{4} \frac{V_1}{(1.0805 \text{ m}^3/\text{kg})} = \frac{\pi (1.0 \text{ m})^2}{4} \frac{V_2}{(1.2855 \text{ m}^3/\text{kg})}$$
(1)
$$2870.7 \text{ kJ/kg} + \frac{V_1^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) = 2772.9 \text{ kJ/kg} + \frac{V_2^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right)$$
(2)

There are two equations and two unknowns. Solving equations (1) and (2) simultaneously using an equation solver such as EES, the velocities are determined to be

$$V_1 =$$
 118.8 m/s
 $V_2 =$ **458.0 m/s**

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5-114 R-134a is condensed in a condenser. The heat transfer per unit mass is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 There are no work interactions.

Analysis We take the pipe in which R-134a is condensed as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as



The enthalpies of R-134a at the inlet and exit of the condenser are (Table A-12, A-13).

$$\begin{array}{c} P_{1} = 900 \text{ kPa} \\ T_{1} = 60^{\circ}\text{C} \end{array} \right\} h_{1} = 295.13 \text{ kJ/kg} \\ P_{2} = 900 \text{ kPa} \\ x = 0 \end{array} \right\} h_{2} = h_{f@900 \text{ kPa}} = 101.61 \text{ kJ/kg}$$

Substituting,

$$q_{\rm out} = 295.13 - 101.61 = 193.5 \, \text{kJ/kg}$$

5-115 Water is heated at constant pressure so that it changes a state from saturated liquid to saturated vapor. The heat transfer per unit mass is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 There are no work interactions.

Analysis We take the pipe in which water is heated as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as



Thus,

 $q_{\rm in} =$ 2047.5 kJ/kg

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Heat loss from the dryer is negligible.

Properties The gas constant of argon is 0.287 kPa.m³/kg.K. The constant pressure specific heat of air at room temperature is $c_p = 1.005 \text{ kJ/kg} \cdot ^{\circ}\text{C}$ (Table A-2*a*).

Analysis (a) We take the pipe as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as





Substituting and solving for the mass flow rate,

$$\dot{m} = \frac{\dot{W}_{in}}{c_p (T_2 - T_1) + \frac{V_2^2 - V_1^2}{2}}$$
$$= \frac{1.50 \text{ kW}}{(1.005 \text{ kJ/kg} \cdot \text{K})(353 - 300)\text{K} + \frac{(21 \text{ m/s})^2 - 0}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2}\right)}$$

= 0.0280 kg/s

(b) The exit specific volume and the volume flow rate are

$$v_2 = \frac{RT_2}{P_2} = \frac{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(353 \text{ K})}{100 \text{ kPa}} = 1.013 \text{ m}^3/\text{kg}$$

 $\dot{v}_2 = \dot{m}v_2 = (0.02793 \text{ kg/s})(1.013 \text{ m}^3/\text{kg}) = 0.0284 \text{ m}^3/\text{s}$

5-117 Problem 5-116 is reconsidered. The effect of the exit velocity on the mass flow rate and the exit volume flow rate is to be investigated.

Analysis The problem is solved using EES, and the solution is given below.

"Given" T1=300 [K] P=100 [kPa] Vel_1=0 [m/s] W_dot_e_in=1.5 [kW] T2=(80+273) [K] Vel_2=21 [m/s] "Properties" c_p=1.005 [kJ/kg-K] R=0.287 [kJ/kg-K] "Analysis" W_dot_e_in=m_dot*c_p*(T2-T1)+m_dot*(Vel_2^2-Vel_1^2)*Convert(m^2/s^2,kJ/kg) "energy balance on hair dryer" v2=(R*T2)/P Vol_dot_2=m_dot*v2



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Charging and Discharging Processes

5-118 An insulated rigid tank is evacuated. A valve is opened, and air is allowed to fill the tank until mechanical equilibrium is established. The final temperature in the tank is to be determined.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. **2** Air is an ideal gas with constant specific heats. **3** Kinetic and potential energies are negligible. **4** There are no work interactions involved. **5** The device is adiabatic and thus heat transfer is negligible.

Properties The specific heat ratio for air at room temperature is k = 1.4 (Table A-2).

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_i = m_2 \quad (\text{since } m_{\rm out} = m_{\rm initial} = 0)$$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{by heat, work, and mass}} = \underbrace{\Delta E_{\text{system}}}_{\text{change in internal, kinetic, potential, etc. energies}}$$

$$m_i h_i = m_2 u_2$$
 (since $Q \cong W \cong E_{out} = E_{initial} = ke \cong pe \cong 0$)

Combining the two balances:

$$u_2 = h_i \rightarrow c_v T_2 = c_p T_i \rightarrow T_2 = (c_p / c_v) T_i = kT_i$$

Substituting,

$$T_2 = 1.4 \times 290 \text{ K} = 406 \text{ K} = 133^{\circ}\text{C}$$



5-119 Helium flows from a supply line to an initially evacuated tank. The flow work of the helium in the supply line and the final temperature of the helium in the tank are to be determined.

Properties The properties of helium are R = 2.0769 kJ/kg.K, $c_p = 5.1926$ kJ/kg.K, $c_v = 3.1156$ kJ/kg.K (Table A-2a).

Analysis The flow work is determined from its definition but we first determine the specific volume

$$v = \frac{RT_{\text{line}}}{P} = \frac{(2.0769 \text{ kJ/kg.K})(120 + 273 \text{ K})}{(200 \text{ kPa})} = 4.0811 \text{ m}^3/\text{kg}$$

$$w_{\text{flow}} = P v = (200 \text{ kPa})(4.0811 \text{ m}^3/\text{kg}) = 816.2 \text{ kJ/kg}$$

Noting that the flow work in the supply line is converted to sensible internal energy in the tank, the final helium temperature in the tank is determined as follows

$$u_{\text{tank}} = h_{\text{line}}$$

$$h_{\text{line}} = c_p T_{\text{line}} = (5.1926 \text{ kJ/kg.K})(120 + 273 \text{ K}) = 2040.7 \text{ kJ/kg}$$

$$u_{\text{-tank}} = c_v T_{\text{tank}} \longrightarrow 2040.7 \text{ kJ/kg} = (3.1156 \text{ kJ/kg.K})T_{\text{tank}} \longrightarrow T_{\text{tank}} = 655.0 \text{ K}$$

Alternative Solution: Noting the definition of specific heat ratio, the final temperature in the tank can also be determined from

$$T_{\text{tank}} = kT_{\text{line}} = 1.667(120 + 273 \text{ K}) = 655.1 \text{ K}$$

which is practically the same result.



5-120 An evacuated bottle is surrounded by atmospheric air. A valve is opened, and air is allowed to fill the bottle. The amount of heat transfer through the wall of the bottle when thermal and mechanical equilibrium is established is to be determined.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. **2** Air is an ideal gas with variable specific heats. **3** Kinetic and potential energies are negligible. **4** There are no work interactions involved. **5** The direction of heat transfer is to the air in the bottle (will be verified).

Properties The gas constant of air is 0.287 kPa.m³/kg.K (Table A-1).

Analysis We take the bottle as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_i = m_2$$
 (since $m_{\rm out} = m_{\rm initial} = 0$)
Energy balance:

 $\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}}$ $Q_{\text{in}} + m_i h_i = m_2 u_2 \quad (\text{since } W \cong E_{\text{out}} = E_{\text{initial}} = ke \cong pe \cong 0)$

Combining the two balances:

$$Q_{\rm in} = m_2 \big(u_2 - h_i \big)$$

where

$$m_{2} = \frac{P_{2}V}{RT_{2}} = \frac{(100 \text{ kPa})(0.020 \text{ m}^{3})}{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(300 \text{ K})} = 0.02323 \text{ kg}$$

$$T_{i} = T_{2} = 300 \text{ K} \xrightarrow{\text{Table A-17}} \frac{h_{i}}{\mu_{i}} = 300.19 \text{ kJ/kg}$$

$$u_{2} = 214.07 \text{ kJ/kg}$$

Substituting,

$$Q_{\rm in} = (0.02323 \text{ kg})(214.07 - 300.19) \text{ kJ/kg} = -2.0 \text{ kJ}$$

or

$Q_{\rm out} = 2.0 \text{ kJ}$

Discussion The negative sign for heat transfer indicates that the assumed direction is wrong. Therefore, we reverse the direction.



5-121 A rigid tank initially contains air at atmospheric conditions. The tank is connected to a supply line, and air is allowed to enter the tank until mechanical equilibrium is established. The mass of air that entered and the amount of heat transfer are to be determined.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. **2** Air is an ideal gas with variable specific heats. **3** Kinetic and potential energies are negligible. **4** There are no work interactions involved. **5** The direction of heat transfer is to the tank (will be verified).

Properties The gas constant of air is 0.287 kPa.m³/kg.K (Table A-1). The properties of air are (Table A-17)

$$T_i = 295 \text{ K} \longrightarrow h_i = 295.17 \text{ kJ/kg}$$

 $T_1 = 295 \text{ K} \longrightarrow u_1 = 210.49 \text{ kJ/kg}$
 $T_2 = 350 \text{ K} \longrightarrow u_2 = 250.02 \text{ kJ/kg}$

Analysis (*a*) We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_i = m_2 - m_1$$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}}$$
$$Q_{\text{in}} + m_i h_i = m_2 u_2 - m_1 u_1 \text{ (since } W \cong ke \cong pe \cong 0\text{)}$$

The initial and the final masses in the tank are

$$m_{1} = \frac{P_{1}\boldsymbol{V}}{RT_{1}} = \frac{(100 \text{ kPa})(2 \text{ m}^{3})}{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(295 \text{ K})} = 2.362 \text{ kg}$$
$$m_{2} = \frac{P_{2}\boldsymbol{V}}{RT_{2}} = \frac{(600 \text{ kPa})(2 \text{ m}^{3})}{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(350 \text{ K})} = 11.946 \text{ kg}$$

Then from the mass balance,

$$m_i = m_2 - m_1 = 11.946 - 2.362 = 9.584 \text{ kg}$$

(b) The heat transfer during this process is determined from

$$Q_{\rm in} = -m_i h_i + m_2 u_2 - m_1 u_1$$

= -(9.584 kg)(295.17 kJ/kg) + (11.946 kg)(250.02 kJ/kg) - (2.362 kg)(210.49 kJ/kg)
= -339 kJ $\rightarrow Q_{\rm out} = 339 \,\text{kJ}$

Discussion The negative sign for heat transfer indicates that the assumed direction is wrong. Therefore, we reversed the direction.



5-122 A rigid tank initially contains superheated steam. A valve at the top of the tank is opened, and vapor is allowed to escape at constant pressure until the temperature rises to 500°C. The amount of heat transfer is to be determined.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process by using constant average properties for the steam leaving the tank. **2** Kinetic and potential energies are negligible. **3** There are no work interactions involved. **4** The direction of heat transfer is to the tank (will be verified).

Properties The properties of water are (Tables A-4 through A-6)

$$P_{1} = 2 \text{ MPa} \left\{ \boldsymbol{v}_{1} = 0.12551 \text{ m}^{3}/\text{kg} \\ T_{1} = 300^{\circ}\text{C} \right\} u_{1} = 2773.2 \text{ kJ/kg}, \quad h_{1} = 3024.2 \text{ kJ/kg} \\ P_{2} = 2 \text{ MPa} \left\{ \boldsymbol{v}_{2} = 0.17568 \text{ m}^{3}/\text{kg} \\ T_{2} = 500^{\circ}\text{C} \right\} u_{2} = 3116.9 \text{ kJ/kg}, \quad h_{2} = 3468.3 \text{ kJ/kg}$$



Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_e = m_1 - m_2$$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}} \underbrace{Q_{\text{in}} - m_e h_e}_{pet = m_2 u_2 - m_1 u_1} \text{ (since } W \cong ke \cong pe \cong 0)$$

The state and thus the enthalpy of the steam leaving the tank is changing during this process. But for simplicity, we assume constant properties for the exiting steam at the average values. Thus,

$$h_e \cong \frac{h_1 + h_2}{2} = \frac{3024.2 + 3468.3 \text{ kJ/kg}}{2} = 3246.2 \text{ kJ/kg}$$

The initial and the final masses in the tank are

$$m_1 = \frac{\mathbf{V}_1}{\mathbf{v}_1} = \frac{0.2 \text{ m}^3}{0.12551 \text{ m}^3/\text{kg}} = 1.594 \text{ kg}$$
$$m_2 = \frac{\mathbf{V}_2}{\mathbf{v}_2} = \frac{0.2 \text{ m}^3}{0.17568 \text{ m}^3/\text{kg}} = 1.138 \text{kg}$$

 $m_e = m_1 - m_2 = 1.594 - 1.138 = 0.456 \text{ kg}$

Then from the mass and energy balance relations,

$$Q_{in} = m_e h_e + m_2 u_2 - m_1 u_1$$

= (0.456 kg)(3246.2 kJ/kg) + (1.138 kg)(3116.9 kJ/kg) - (1.594 kg)(2773.2 kJ/kg)
= **606.8 kJ**

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. 2 The expansion process is quasi-equilibrium. 3 Kinetic and potential energies are negligible. 3 There are no work interactions involved other than boundary work. 4 The device is insulated and thus heat transfer is negligible.

Properties The properties of steam are (Tables A-4 through A-6)

$$\begin{array}{c} P_{1} = 200 \text{ kPa} \\ x_{1} = 0.6 \end{array} \right\} h_{1} = h_{f} + x_{1}h_{fg} \\ = 504.71 + 0.6 \times 2201.6 = 1825.6 \text{ kJ/kg} \\ P_{2} = 200 \text{ kPa} \\ \text{sat. vapor} \end{array} \right\} h_{2} = h_{g@200 \text{ kPa}} = 2706.3 \text{ kJ/kg} \\ P_{i} = 0.5 \text{ MPa} \\ T_{i} = 350^{\circ}\text{C} \end{array} \right\} h_{i} = 3168.1 \text{ kJ/kg}$$

Analysis (*a*) The cylinder contains saturated vapor at the final state at a pressure of 200 kPa, thus the final temperature in the cylinder must be

$$T_2 = T_{\text{sat}(a) 200 \text{ kPa}} = 120.2^{\circ}\text{C}$$

(b) We take the cylinder as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance: $m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_i = m_2 - m_1$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}}$$
$$m_i h_i = W_{\text{b,out}} + m_2 u_2 - m_1 u_1 \text{ (since } Q \cong ke \cong pe \cong 0\text{)}$$

Combining the two relations gives

$$0 = W_{b,out} - (m_2 - m_1)h_i + m_2u_2 - m_1u_1$$

or,

$$0 = -(m_2 - m_1)h_i + m_2h_2 - m_1h_1$$

since the boundary work and ΔU combine into ΔH for constant pressure expansion and compression processes. Solving for m₂ and substituting,

$$m_2 = \frac{h_i - h_1}{h_i - h_2} m_1 = \frac{(3168.1 - 1825.6) \text{ kJ/kg}}{(3168.1 - 2706.3) \text{ kJ/kg}} (10 \text{ kg}) = 29.07 \text{ kg}$$

Thus,

$$m_{\rm i} = m_2 - m_1 = 29.07 - 10 =$$
19.07 kg

5-124E A scuba diver's air tank is to be filled with air from a compressed air line. The temperature and mass in the tank at the final state are to be determined.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. **2** Air is an ideal gas with constant specific heats. **3** Kinetic and potential energies are negligible. **4** There are no work interactions involved. **5** The tank is well-insulated, and thus there is no heat transfer.

Properties The gas constant of air is 0.3704 psia·ft³/lbm·R (Table A-1E). The specific heats of air at room temperature are $c_p = 0.240$ Btu/lbm·R and $c_v = 0.171$ Btu/lbm·R (Table A-2Ea).

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_i = m_2 - m_1$$

Energy balance:



Combining the two balances:

$$(m_2 - m_1)c_pT_i = m_2c_vT_2 - m_1c_vT_1$$

The initial and final masses are given by

$$m_1 = \frac{P_1 \mathbf{V}}{RT_1} = \frac{(20 \text{ psia})(2 \text{ ft}^3)}{(0.3704 \text{ psia} \cdot \text{ft}^3/\text{lbm} \cdot \text{R})(70 + 460 \text{ R})} = 0.2038 \text{ lbm}$$
$$m_2 = \frac{P_2 \mathbf{V}}{RT_2} = \frac{(120 \text{ psia})(2 \text{ ft}^3)}{(0.3704 \text{ psia} \cdot \text{ft}^3/\text{lbm} \cdot \text{R})T_2} = \frac{647.9}{T_2}$$

Substituting,

$$\left(\frac{647.9}{T_2} - 0.2038\right)(0.24)(560) = \frac{647.9}{T_2}(0.171)T_2 - (0.2038)(0.171)(530)$$

whose solution is

$$T_2 = 727.4 \text{ R} = 267.4^{\circ}\text{F}$$

The final mass is then

$$m_2 = \frac{647.9}{T_2} = \frac{647.9}{727.4} = 0.890$$
 lbm



5-125 R-134a from a tank is discharged to an air-conditioning line in an isothermal process. The final quality of the R-134a in the tank and the total heat transfer are to be determined.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the exit remains constant. **2** Kinetic and potential energies are negligible. **3** There are no work interactions involved.

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system}$$
$$- m_e = m_2 - m_1$$
$$m_e = m_1 - m_2$$

Energy balance:



$$Q_{\rm in} = m_2 u_2 - m_1 u_1 + m_e h_e$$

Combining the two balances:

$$Q_{\rm in} = m_2 u_2 - m_1 u_1 + (m_1 - m_2) h_e$$

The initial state properties of R-134a in the tank are

$$T_{1} = 24 \text{°C} \\ x = 0$$
 $\begin{cases} \boldsymbol{v}_{1} = 0.0008261 \text{ m}^{3}/\text{kg} \\ u_{1} = 84.44 \text{ kJ/kg} \\ h_{a} = 84.98 \text{ kJ/kg} \end{cases}$ (Table A-11)

Note that we assumed that the refrigerant leaving the tank is at saturated liquid state, and found the exiting enthalpy accordingly. The volume of the tank is

$$V = m_1 v_1 = (5 \text{ kg})(0.0008261 \text{ m}^3/\text{kg}) = 0.004131 \text{ m}^3$$

The final specific volume in the container is

$$v_2 = \frac{v}{m_2} = \frac{0.004131 \,\mathrm{m}^3}{0.25 \,\mathrm{kg}} = 0.01652 \,\mathrm{m}^3/\mathrm{kg}$$

The final state is now fixed. The properties at this state are (Table A-11)

$$T_{2} = 24^{\circ}\text{C}$$

$$v_{2} = 0.01652 \text{ m}^{3}/\text{kg} \begin{cases} x_{2} = \frac{v_{2} - v_{f}}{v_{fg}} = \frac{0.01652 - 0.0008261}{0.031834 - 0.0008261} = \textbf{0.5061} \\ u_{2} = u_{f} + x_{2}u_{fg} = 84.44 \text{ kJ/kg} + (0.5061)(158.65 \text{ kJ/kg}) = 164.73 \text{ kJ/kg} \end{cases}$$

Substituting into the energy balance equation,

$$Q_{in} = m_2 u_2 - m_1 u_1 + (m_1 - m_2) h_e$$

= (0.25 kg)(164.73 kJ/kg) - (5 kg)(84.44 kJ/kg) + (4.75 kg)(84.98 kJ/kg)
= **22.64 kJ**



5-126E Oxygen is supplied to a medical facility from 10 compressed oxygen tanks in an isothermal process. The mass of oxygen used and the total heat transfer to the tanks are to be determined.

Assumptions 1 This is an unsteady process but it can be analyzed as a uniform-flow process. 2 Oxygen is an ideal gas with constant specific heats. 3 Kinetic and potential energies are negligible. 4 There are no work interactions involved.

Properties The gas constant of oxygen is 0.3353 psia·ft³/lbm·R (Table A-1E). The specific heats of oxygen at room temperature are $c_p = 0.219$ Btu/lbm·R and $c_y = 0.157$ Btu/lbm·R (Table A-2Ea).

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system}$$
$$- m_e = m_2 - m_1$$
$$m_e = m_1 - m_2$$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{potential, etc. energies}}$$
$$Q_{\text{in}} - m_e h_e = m_2 u_2 - m_1 u_1$$
$$Q_{\text{in}} = m_2 u_2 - m_1 u_1 + m_e h_e$$
$$Q_{\text{in}} = m_2 c_v T_2 - m_1 c_v T_1 + m_e c_p T_e$$



Combining the two balances:

$$Q_{\rm in} = m_2 c_v T_2 - m_1 c_v T_1 + (m_1 - m_2) c_p T_e$$

The initial and final masses, and the mass used are

$$m_1 = \frac{P_1 \mathbf{V}}{RT_1} = \frac{(1500 \text{ psia})(15 \text{ ft}^3)}{(0.3353 \text{ psia} \cdot \text{ft}^3/\text{lbm} \cdot \text{R})(80 + 460 \text{ R})} = 124.3 \text{ lbm}$$
$$m_2 = \frac{P_2 \mathbf{V}}{RT_2} = \frac{(300 \text{ psia})(15 \text{ ft}^3)}{(0.3353 \text{ psia} \cdot \text{ft}^3/\text{lbm} \cdot \text{R})(80 + 460 \text{ R})} = 24.85 \text{ lbm}$$

 $m_e = m_1 - m_2 = 124.3 - 24.85 =$ **99.41 lbm**

Substituting into the energy balance equation,

$$Q_{in} = m_2 c_v T_2 - m_1 c_v T_1 + m_e c_p T_e$$

= (24.85)(0.157)(540) - (124.3)(0.157)(540) + (99.41)(0.219)(540)
= **3328 Btu**

5-127E Steam is supplied from a line to a weighted piston-cylinder device. The final temperature (and quality if appropriate) of the steam in the piston cylinder and the total work produced as the device is filled are to be determined.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. **2** Kinetic and potential energies are negligible. **3** The process is adiabatic.

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, and also noting that the initial mass in the system is zero, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system}$$

 $m_i = m_2$

Energy balance:

$$\underbrace{E_{in} - E_{out}}_{Net \text{ energy transfer}} = \underbrace{\Delta E_{system}}_{Change in internal, kinetic, potential, etc. energies}$$

$$m_i h_i - W_{b,out} = m_2 u_2$$

$$W_{b,out} = m_i h_i - m_2 u_2$$

Combining the two balances:

$$W_{b.\text{out}} = m_2(h_i - u_2)$$

The boundary work is determined from

$$W_{h \text{ out}} = P(U_2 - U_1) = P(m_2 u_2 - m_1 u_1) = Pm_2 u_2$$

Substituting, the energy balance equation simplifies into

$$Pm_2 \boldsymbol{v}_2 = m_2 (h_i - u_2)$$
$$P\boldsymbol{v}_2 = h_i - u_2$$

The enthalpy of steam at the inlet is

$$P_i = 300 \text{ psia}$$

$$T_i = 450^{\circ} \text{F}$$

$$h_i = 1226.4 \text{ Btu/lbm} \quad \text{(Table A - 6E)}$$

Substituting this value into the energy balance equation and using an iterative solution of this equation gives (or better yet using EES software)

$$T_2 = 425.1^{\circ}F$$

 $u_2 = 1135.5 \text{ Btu/lbm}$
 $v_2 = 2.4575 \text{ ft}^3/\text{lbm}$

The final mass is

$$m_2 = \frac{\mathbf{v}_2}{\mathbf{v}_2} = \frac{10 \text{ ft}^3}{2.4575 \text{ ft}^3/\text{lbm}} = 4.069 \text{ lbm}$$

and the work produced is

$$W_{b,\text{out}} = P \boldsymbol{V}_2 = (200 \text{ psia})(10 \text{ ft}^3) \left(\frac{1 \text{ Btu}}{5.404 \text{ psia} \cdot \text{ ft}^3}\right) = 370.1 \text{ Btu}$$

5-128E Oxygen is supplied from a line to a weighted piston-cylinder device. The final temperature of the oxygen in the piston cylinder and the total work produced as the device is filled are to be determined.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. 2 Kinetic and potential energies are negligible. 3 The process is adiabatic. 4 Oxygen is an ideal gas with constant specific heats.

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, and also noting that the initial mass in the system is zero, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system}$$

 $m_i = m_2$

Energy balance:

$$\underbrace{E_{in} - E_{out}}_{Net \text{ energy transfer}} = \underbrace{\Delta E_{system}}_{Change in internal, kinetic, potential, etc. energies}$$

$$m_i h_i - W_{b,out} = m_2 u_2$$

$$W_{b,out} = m_i h_i - m_2 u_2$$

Combining the two balances:

$$W_{b,\text{out}} = m_2(h_i - u_2)$$

The boundary work is determined from

$$W_{h \text{ out}} = P(U_2 - U_1) = P(m_2 u_2 - m_1 u_1) = Pm_2 u_2$$

Substituting, the energy balance equation simplifies into

$$Pm_2 \mathbf{v}_2 = m_2(h_i - u_2)$$
$$P\mathbf{v}_2 = h_i - u_2$$
$$RT_2 = c_p T_i - c_v T_2$$

Solving for the final temperature,

$$RT_2 = c_p T_i - c_v T_2 \longrightarrow T_2 = \frac{c_p}{R + c_v} T_i = \frac{c_p}{c_p} T_i = T_i = 450^{\circ} F$$

The work produced is

$$W_{b,\text{out}} = P \mathcal{U}_2 = (200 \text{ psia})(10 \text{ ft}^3) \left(\frac{1 \text{ Btu}}{5.404 \text{ psia} \cdot \text{ft}^3}\right) = 370.1 \text{ Btu}$$

5-129 A rigid tank initially contains saturated R-134a vapor. The tank is connected to a supply line, and R-134a is allowed to enter the tank. The mass of the R-134a that entered and the heat transfer are to be determined.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. **2** Kinetic and potential energies are negligible. **3** There are no work interactions involved. **4** The direction of heat transfer is to the tank (will be verified).

Properties The properties of refrigerant are (Tables A-11 through A-13)

$$P_{1} = 1.4 \text{ MPa} \left\{ \boldsymbol{\nu}_{1} = \boldsymbol{\nu}_{g@1.4 \text{ MPa}} = 0.01411 \text{ m}^{3}/\text{kg} \\ \text{sat.vapor} \right\} u_{1} = u_{g@1.4 \text{ MPa}} = 256.37 \text{ kJ/kg} \\ P_{2} = 1.6 \text{ MPa} \left\{ \boldsymbol{\nu}_{2} = \boldsymbol{\nu}_{f@1.6 \text{ MPa}} = 0.0009400 \text{ m}^{3}/\text{kg} \\ \text{sat. liquid} \right\} u_{2} = u_{f@1.6 \text{ MPa}} = 134.43 \text{ kJ/kg} \\ P_{i} = 1.6 \text{ MPa} \\ T_{i} = 36^{\circ}\text{C} \end{array} \right\} h_{i} = h_{f@36^{\circ}\text{C}} = 102.33 \text{ kJ/kg}$$



Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_i = m_2 - m_1$$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}}$$
$$\underbrace{Q_{\text{in}} + m_i h_i}_{i} = m_2 u_2 - m_1 u_1 \text{ (since } W \cong ke \cong pe \cong 0)$$

(a) The initial and the final masses in the tank are

$$m_1 = \frac{\mathbf{V}_1}{\mathbf{v}_1} = \frac{0.03 \text{ m}^3}{0.01411 \text{ m}^3/\text{kg}} = 2.127 \text{ kg}$$
$$m_2 = \frac{\mathbf{V}_2}{\mathbf{v}_2} = \frac{0.03 \text{ m}^3}{0.0009400 \text{ m}^3/\text{kg}} = 31.92 \text{ kg}$$

Then from the mass balance

ł

$$m_i = m_2 - m_1 = 31.92 - 2.127 = 29.79$$
 kg

(c) The heat transfer during this process is determined from the energy balance to be

$$Q_{in} = -m_i h_i + m_2 u_2 - m_1 u_1$$

= -(29.79 kg)(102.33 kJ/kg) + (31.92 kg)(134.43 kJ/kg) - (2.127 kg)(256.37 kJ/kg)
= **697 kJ**

5-130 A rigid tank initially contains saturated liquid water. A valve at the bottom of the tank is opened, and half of the mass in liquid form is withdrawn from the tank. The temperature in the tank is maintained constant. The amount of heat transfer is to be determined.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid leaving the device remains constant. 2 Kinetic and potential energies are negligible. 3 There are no work interactions involved. 4 The direction of heat transfer is to the tank (will be verified).

Properties The properties of water are (Tables A-4 through A-6)

$$T_{1} = 200^{\circ}\text{C} \left\{ \begin{array}{l} \mathbf{v}_{1} = \mathbf{v}_{f@200^{\circ}\text{C}} = 0.001157 \text{ m}^{3}/\text{kg} \\ \text{sat. liquid} \end{array} \right\} u_{1} = u_{f@200^{\circ}\text{C}} = 850.46 \text{ kJ/kg} \\ T_{e} = 200^{\circ}\text{C} \\ \text{sat. liquid} \end{array} \right\} h_{e} = h_{f@200^{\circ}\text{C}} = 852.26 \text{ kJ/kg}$$

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:
$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_e = m_1 - m_2$$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}}$$
$$Q_{\text{in}} = m_e h_e + m_2 u_2 - m_1 u_1 \text{ (since } W \cong ke \cong pe \cong 0)$$

The initial and the final masses in the tank are

$$m_1 = \frac{\mathbf{v}_1}{\mathbf{v}_1} = \frac{0.3 \text{ m}^3}{0.001157 \text{ m}^3/\text{kg}} = 259.4 \text{ kg}$$
$$m_2 = \frac{1}{2}m_1 = \frac{1}{2}(259.4 \text{ kg}) = 129.7 \text{ kg}$$

Then from the mass balance,

$$m_e = m_1 - m_2 = 259.4 - 129.7 = 129.7 \text{ kg}$$

Now we determine the final internal energy,

$$\boldsymbol{v}_{2} = \frac{\boldsymbol{v}}{m_{2}} = \frac{0.3 \text{ m}^{3}}{129.7 \text{ kg}} = 0.002313 \text{ m}^{3}/\text{kg}$$

$$x_{2} = \frac{\boldsymbol{v}_{2} - \boldsymbol{v}_{f}}{\boldsymbol{v}_{fg}} = \frac{0.002313 - 0.001157}{0.12721 - 0.001157} = 0.009171$$

$$T_{2} = 200^{\circ}\text{C}$$

$$x_{2} = 0.009171 \left\{ u_{2} = u_{f} + x_{2}u_{fg} = 850.46 + (0.009171)(1743.7) = 866.46 \text{ kJ/kg} \right\}$$

Then the heat transfer during this process is determined from the energy balance by substitution to be

$$Q = (129.7 \text{ kg})(852.26 \text{ kJ/kg}) + (129.7 \text{ kg})(866.46 \text{ kJ/kg}) - (259.4 \text{ kg})(850.46 \text{ kJ/kg})$$

= 2308 kJ



Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid leaving the device remains constant. **2** Kinetic and potential energies are negligible. **3** There are no work interactions involved. **4** The direction of heat transfer is to the tank (will be verified).

Properties The properties of R-134a are (Tables A-11 through A-13)

$$P_{1} = 800 \text{ kPa} \rightarrow \mathbf{v}_{f} = 0.0008458 \text{ m}^{3}/\text{kg}, \mathbf{v}_{g} = 0.025621 \text{ m}^{3}/\text{kg}$$

$$u_{f} = 94.79 \text{ kJ/kg}, u_{g} = 246.79 \text{ kJ/kg}$$

$$P_{2} = 800 \text{ kPa}$$

$$v_{2} = \mathbf{v}_{g@800 \text{ kPa}} = 0.025621 \text{ m}^{3}/\text{kg}$$
sat. vapor
$$u_{2} = u_{g@800 \text{ kPa}} = 246.79 \text{ kJ/kg}$$

$$P_{e} = 800 \text{ kPa}$$
sat. liquid
$$h_{e} = h_{f@800 \text{ kPa}} = 95.47 \text{ kJ/kg}$$

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_e = m_1 - m_2$$

Energy balance:

$$\underbrace{E_{in} - E_{out}}_{Net energy transfer} = \underbrace{\Delta E_{system}}_{Change in internal, kinetic, potential, etc. energies} Q_{in} = m_e h_e + m_2 u_2 - m_1 u_1 \text{ (since } W \cong ke \cong pe \cong 0)$$

The initial mass, initial internal energy, and final mass in the tank are

$$m_{1} = m_{f} + m_{g} = \frac{V_{f}}{v_{f}} + \frac{V_{g}}{v_{g}} = \frac{0.12 \times 0.25 \text{ m}^{3}}{0.0008458 \text{ m}^{3}/\text{kg}} + \frac{0.12 \times 0.75 \text{ m}^{3}}{0.025621 \text{ m}^{3}/\text{kg}} = 35.47 + 3.513 = 38.98 \text{ kg}$$

$$U_{1} = m_{I}u_{1} = m_{f}u_{f} + m_{g}u_{g} = (35.47)(94.79) + (3.513)(246.79) = 4229.2 \text{ kJ}$$

$$m_{2} = \frac{V}{v_{2}} = \frac{0.12 \text{ m}^{3}}{0.025621 \text{ m}^{3}/\text{kg}} = 4.684 \text{ kg}$$

Then from the mass and energy balances,

$$m_e = m_1 - m_2 = 38.98 - 4.684 = 34.30 \text{ kg}$$

 $Q_{\text{in}} = (34.30 \text{ kg})(95.47 \text{ kJ/kg}) + (4.684 \text{ kg})(246.79 \text{ kJ/kg}) - 4229 \text{ kJ} = 201.2 \text{ kJ}$

R-134a

Q

5-132E A rigid tank initially contains saturated liquid-vapor mixture of R-134a. A valve at the top of the tank is opened, and vapor is allowed to escape at constant pressure until all the liquid in the tank disappears. The amount of heat transfer is to be determined.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid leaving the device remains constant. **2** Kinetic and potential energies are negligible. **3** There are no work interactions involved.

Properties The properties of R-134a are (Tables A-11E through A-13E)

$$P_{1} = 160 \text{ psia} \rightarrow v_{f} = 0.01413 \text{ ft}^{3}/\text{lbm}, v_{g} = 0.29316 \text{ ft}^{3}/\text{lbm}$$

$$u_{f} = 48.10 \text{ Btu/lbm}, u_{g} = 108.50 \text{ Btu/lbm}$$

$$P_{2} = 160 \text{ psia} \quad v_{2} = v_{g@160 \text{ psia}} = 0.29316 \text{ ft}^{3}/\text{lbm}$$
sat. vapor
$$\begin{cases} u_{2} = u_{g@160 \text{ psia}} = 108.50 \text{ Btu/lbm} \\ u_{2} = u_{g@160 \text{ psia}} = 108.50 \text{ Btu/lbm} \end{cases}$$

$$P_{e} = 160 \text{ psia} \\ \text{sat. vapor} \end{cases} h_{e} = h_{g@160 \text{ psia}} = 117.18 \text{ Btu/lbm}$$

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_e = m_1 - m_2$$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}}$$
$$Q_{\text{in}} - m_e h_e = m_2 u_2 - m_1 u_1 \text{ (since } W \cong ke \cong pe \cong 0)$$

The initial mass, initial internal energy, and final mass in the tank are

$$m_{1} = m_{f} + m_{g} = \frac{V_{f}}{v_{f}} + \frac{V_{g}}{v_{g}} = \frac{2 \times 0.2 \text{ ft}^{3}}{0.01413 \text{ ft}^{3}/\text{lbm}} + \frac{2 \times 0.8 \text{ ft}^{3}}{0.29316 \text{ ft}^{3}/\text{lbm}} = 7.077 + 6.48 = 13.56 \text{ lbm}$$

$$U_{1} = m_{1}u_{1} = m_{f}u_{f} + m_{g}u_{g} = (7.077)(48.10) + (6.48)(108.50) = 1043 \text{ Btu}$$

$$m_{2} = \frac{V}{v_{2}} = \frac{2 \text{ ft}^{3}}{0.29316 \text{ ft}^{3}/\text{lbm}} = 6.822 \text{ lbm}$$

Then from the mass and energy balances,

$$m_e = m_1 - m_2 = 13.56 - 6.822 = 6.736 \text{ lbm}$$

$$Q_{\text{in}} = m_e h_e + m_2 u_2 - m_1 u_1$$

$$= (6.736 \text{ lbm})(117.18 \text{ Btu/lbm}) + (6.822 \text{ lbm})(108.50 \text{ Btu/lbm}) - 1043 \text{ Btu}$$

$$= 486 \text{ Btu}$$

Sat. vapor P = 160 psia $V = 2 \text{ ft}^3$ **5-133** A rigid tank initially contains saturated R-134a liquid-vapor mixture. The tank is connected to a supply line, and R-134a is allowed to enter the tank. The final temperature in the tank, the mass of R-134a that entered, and the heat transfer are to be determined.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. 2 Kinetic and potential energies are negligible. 3 There are no work interactions involved. 4 The direction of heat transfer is to the tank (will be verified).

Properties The properties of refrigerant are (Tables A-11 through A-13)

$$T_{1} = 14^{\circ}C \qquad v_{1} = v_{f} + x_{1}v_{fg} = 0.0008020 + 0.7 \times (0.04342 - 0.0008020) = 0.03063 \text{ m}^{3}/\text{kg}$$

$$x_{1} = 0.7 \qquad u_{1} = u_{f} + x_{1}u_{fg} = 70.57 + 0.7 \times 167.26 = 187.65 \text{ kJ/kg}$$

$$P_{2} = 700 \text{ kPa} \qquad v_{2} = v_{g@700 \text{ kPa}} = 0.02936 \text{ m}^{3}/\text{kg}$$
sat. vapor
$$\begin{cases} u_{2} = u_{g@700 \text{ kPa}} = 244.48 \text{ kJ/kg} \end{cases}$$

$$P_{i} = 1.0 \text{ MPa} \\ T_{i} = 100^{\circ}C \end{cases} \qquad h_{i} = 335.06 \text{ kJ/kg}$$

$$R-134a \xrightarrow{1}{10}$$

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance: $m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_i = m_2 - m_1$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}}$$
$$Q_{\text{in}} + m_i h_i = m_2 u_2 - m_1 u_1 \text{ (since } W \cong ke \cong pe \cong 0\text{)}$$

(a) The tank contains saturated vapor at the final state at 800 kPa, and thus the final temperature is the saturation temperature at this pressure,

$$T_2 = T_{\text{sat}(\bar{a}),700 \text{ kPa}} = 26.7^{\circ}\text{C}$$

(b) The initial and the final masses in the tank are

$$m_1 = \frac{\mathbf{v}}{\mathbf{v}_1} = \frac{0.4 \text{ m}^3}{0.03063 \text{ m}^3/\text{kg}} = 13.06 \text{ kg}$$
$$m_2 = \frac{\mathbf{v}}{\mathbf{v}_2} = \frac{0.4 \text{ m}^3}{0.02936 \text{ m}^3/\text{kg}} = 13.62 \text{ kg}$$

Then from the mass balance

$$m_i = m_2 - m_1 = 13.62 - 13.06 = 0.5653 \text{ kg}$$

(c) The heat transfer during this process is determined from the energy balance to be

$$Q_{in} = -m_i h_i + m_2 u_2 - m_1 u_1$$

= -(0.5653 kg)(335.06 kJ/kg) + (13.62 kg)(244.48 kJ/kg) - (13.06 kg)(187.65 kJ/kg)
= **691 kJ**







5-134 A hot-air balloon is considered. The final volume of the balloon and work produced by the air inside the balloon as it expands the balloon skin are to be determined.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process. 2 Air is an ideal gas with constant specific heats. 3 Kinetic and potential energies are negligible. 4 There is no heat transfer.

Properties The gas constant of air is 0.287 kPa·m³/kg·K (Table A-1).

Analysis The specific volume of the air at the entrance and exit, and in the balloon is

$$v = \frac{RT}{P} = \frac{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(35 + 273 \text{ K})}{100 \text{ kPa}} = 0.8840 \text{ m}^3/\text{kg}$$

The mass flow rate at the entrance is then

$$\dot{m}_i = \frac{A_i V_i}{v} = \frac{(1 \text{ m}^2)(2 \text{ m/s})}{0.8840 \text{ m}^3/\text{kg}} = 2.262 \text{ kg/s}$$

while that at the outlet is

$$\dot{m}_e = \frac{A_e V_e}{v} = \frac{(0.5 \text{ m}^2)(1 \text{ m/s})}{0.8840 \text{ m}^3/\text{kg}} = 0.5656 \text{ kg/s}$$

Applying a mass balance to the balloon,

$$\begin{split} m_{\rm in} &- m_{\rm out} = \Delta m_{\rm system} \\ m_i &- m_e = m_2 - m_1 \\ m_2 &- m_1 = (\dot{m}_i - \dot{m}_e) \Delta t = [(2.262 - 0.5656) \, \text{kg/s}](2 \times 60 \, \text{s}) = 203.6 \, \text{kg} \end{split}$$

.

The volume in the balloon then changes by the amount

$$\Delta \mathbf{V} = (m_2 - m_1)\mathbf{v} = (203.6 \text{ kg})(0.8840 \text{ m}^3/\text{kg}) = 180 \text{ m}^3$$

and the final volume of the balloon is

 $V_2 = V_1 + \Delta V = 75 + 180 = 255 \text{ m}^3$

In order to push back the boundary of the balloon against the surrounding atmosphere, the amount of work that must be done is

$$W_{b,\text{out}} = P\Delta \mathbf{V} = (100 \text{ kPa})(180 \text{ m}^3) \left(\frac{1 \text{ kJ}}{1 \text{ kPa} \cdot \text{m}^3}\right) = 18,000 \text{ kJ}$$

5-135 An insulated rigid tank initially contains helium gas at high pressure. A valve is opened, and half of the mass of helium is allowed to escape. The final temperature and pressure in the tank are to be determined.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process by using constant average properties for the helium leaving the tank. 2 Kinetic and potential energies are negligible. 3 There are no work interactions involved. 4 The tank is insulated and thus heat transfer is negligible. 5 Helium is an ideal gas with constant specific heats.

Properties The specific heat ratio of helium is k = 1.667 (Table A-2).

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:
$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_e = m_1 - m_2$$

$$m_2 = \frac{1}{2}m_1 \text{ (given)} \longrightarrow m_e = m_2 = \frac{1}{2}m_1$$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}} - m_e h_e = m_2 u_2 - m_1 u_1 \text{ (since } W \cong Q \cong ke \cong pe \cong 0)$$



Note that the state and thus the enthalpy of helium leaving the tank is changing during this process. But for simplicity, we assume constant properties for the exiting steam at the average values.

Combining the mass and energy balances: $0 = \frac{1}{2}m_1h_e + \frac{1}{2}m_1u_2 - m_1u_1$

Dividing by
$$m_1/2$$
 $0 = h_e + u_2 - 2u_1$ or $0 = c_p \frac{T_1 + T_2}{2} + c_v T_2 - 2c_v T_1$

Dividing by c_{v} : $0 = k(T_1 + T_2) + 2T_2 - 4T_1$ since $k = c_p / c_v$

Solving for
$$T_2$$
: $T_2 = \frac{(4-k)}{(2+k)}T_1 = \frac{(4-1.667)}{(2+1.667)}(403 \text{ K}) = 257 \text{ K}$

The final pressure in the tank is

$$\frac{P_1 \mathbf{V}}{P_2 \mathbf{V}} = \frac{m_1 R T_1}{m_2 R T_2} \longrightarrow P_2 = \frac{m_2 T_2}{m_1 T_1} P_1 = \frac{1}{2} \frac{257}{403} (3000 \text{ kPa}) = 956 \text{ kPa}$$

5-136E An insulated rigid tank equipped with an electric heater initially contains pressurized air. A valve is opened, and air is allowed to escape at constant temperature until the pressure inside drops to 30 psia. The amount of electrical work transferred is to be determined.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the exit temperature (and enthalpy) of air remains constant. 2 Kinetic and potential energies are negligible. 3 The tank is insulated and thus heat transfer is negligible. 4 Air is an ideal gas with variable specific heats.

Properties The gas constant of air is R = 0.3704 psia.ft³/lbm.R (Table A-1E). The properties of air are (Table A-17E)

 $\begin{array}{cccc} T_i = 580 \ \mathrm{R} & \longrightarrow & h_i = 138.66 \ \mathrm{Btu} \, / \, \mathrm{lbm} \\ T_1 = 580 \ \mathrm{R} & \longrightarrow & u_1 = 98.90 \ \mathrm{Btu} \, / \, \mathrm{lbm} \\ T_2 = 580 \ \mathrm{R} & \longrightarrow & u_2 = 98.90 \ \mathrm{Btu} \, / \, \mathrm{lbm} \end{array}$

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_e = m_1 - m_2$$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}}$$
$$W_{\text{a,in}} - m_{a}h_{a} = m_{2}u_{2} - m_{1}u_{1} \text{ (since } Q \cong ke \cong pe \cong 0)$$

The initial and the final masses of air in the tank are

$$m_1 = \frac{P_1 \mathbf{V}}{RT_1} = \frac{(75 \text{ psia})(60 \text{ ft}^3)}{(0.3704 \text{ psia} \cdot \text{ft}^3/\text{lbm} \cdot \text{R})(580 \text{ R})} = 20.95 \text{ lbm}$$
$$m_2 = \frac{P_2 \mathbf{V}}{RT_2} = \frac{(30 \text{ psia})(60 \text{ ft}^3)}{(0.3704 \text{ psia} \cdot \text{ft}^3/\text{lbm} \cdot \text{R})(580 \text{ R})} = 8.38 \text{ lbm}$$

Then from the mass and energy balances,

 $m_e = m_1 - m_2 = 20.95 - 8.38 = 12.57$ lbm

$$W_{e,in} = m_e h_e + m_2 u_2 - m_1 u_1$$

= (12.57 lbm)(138.66 Btu/lbm) + (8.38 lbm)(98.90 Btu/lbm) - (20.95 lbm)(98.90 Btu/lbm)
= **500 Btu**


5-137 A vertical cylinder initially contains air at room temperature. Now a valve is opened, and air is allowed to escape at constant pressure and temperature until the volume of the cylinder goes down by half. The amount air that left the cylinder and the amount of heat transfer are to be determined.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the exit temperature (and enthalpy) of air remains constant. 2 Kinetic and potential energies are negligible. 3 There are no work interactions other than boundary work. 4 Air is an ideal gas with constant specific heats. 5 The direction of heat transfer is to the cylinder (will be verified).

Properties The gas constant of air is R = 0.287 kPa.m³/kg.K (Table A-1).

Analysis (*a*) We take the cylinder as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_e = m_1 - m_2$$

Energy balance:

$$\underbrace{E_{in} - E_{out}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}}$$
$$\underbrace{Q_{in} + W_{b in} - m_e h_e}_{pe} = m_2 u_2 - m_1 u_1 \text{ (since } ke \cong pe \cong 0\text{)}$$

The initial and the final masses of air in the cylinder are

$$m_{1} = \frac{P_{1} \mathbf{V}_{1}}{RT_{1}} = \frac{(300 \text{ kPa})(0.2 \text{ m}^{3})}{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(293 \text{ K})} = 0.714 \text{ kg}$$
$$m_{2} = \frac{P_{2} \mathbf{V}_{2}}{RT_{2}} = \frac{(300 \text{ kPa})(0.1 \text{ m}^{3})}{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(293 \text{ K})} = 0.357 \text{ kg} = \frac{1}{2} m_{1}$$

Then from the mass balance,

$$m_e = m_1 - m_2 = 0.714 - 0.357 = 0.357 \text{ kg}$$

(b) This is a constant pressure process, and thus the W_b and the ΔU terms can be combined into ΔH to yield

$$Q = m_e h_e + m_2 h_2 - m_1 h_1$$

Noting that the temperature of the air remains constant during this process, we have

$$h_i = h_1 = h_2 = h.$$

Also,

$$m_e = m_2 = \frac{1}{2}m_1$$

Thus,

$$Q = \left(\frac{1}{2}m_1 + \frac{1}{2}m_1 - m_1\right)h = \mathbf{0}$$



5-138 A balloon is initially filled with helium gas at atmospheric conditions. The tank is connected to a supply line, and helium is allowed to enter the balloon until the pressure rises from 100 to 125 kPa. The final temperature in the balloon is to be determined.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. 2 Helium is an ideal gas with constant specific heats. 3 The expansion process is quasi-equilibrium. 4 Kinetic and potential energies are negligible. 5 There are no work interactions involved other than boundary work. 6 Heat transfer is negligible.

Properties The gas constant of helium is $R = 2.0769 \text{ kJ/kg} \cdot \text{K}$ (Table A-1). The specific heats of helium are $c_p = 5.1926$ and $c_v = 3.1156 \text{ kJ/kg} \cdot \text{K}$ (Table A-2a).

Analysis We take the cylinder as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_i = m_2 - m_1$$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}} \\ m_i h_i = W_{\text{b,out}} + m_2 u_2 - m_1 u_1 \text{ (since } Q \cong ke \cong pe \cong 0) \\ m_1 = \frac{P_1 V_1}{RT_1} = \frac{(100 \text{ kPa})(40 \text{ m}^3)}{(2.0769 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(290 \text{ K})} = 6.641 \text{ kg} \\ \frac{P_1}{P_2} = \frac{V_1}{V_2} \longrightarrow V_2 = \frac{P_2}{P_1} V_1 = \frac{125 \text{ kPa}}{100 \text{ kPa}} (40 \text{ m}^3) = 50 \text{ m}^3 \\ m_2 = \frac{P_2 V_2}{RT_2} = \frac{(125 \text{ kPa})(50 \text{ m}^3)}{(2.0769 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(T_2 \text{ K})} = \frac{3009.3}{T_2} \text{ kg}$$



Then from the mass balance,

$$m_i = m_2 - m_1 = \frac{3009.3}{T_2} - 6.641 \,\mathrm{kg}$$

Noting that P varies linearly with U, the boundary work done during this process is

$$W_b = \frac{P_1 + P_2}{2} (\mathbf{\nu}_2 - \mathbf{\nu}_1) = \frac{(100 + 125) \text{kPa}}{2} (50 - 40) \text{m}^3 = 1125 \text{ kJ}$$

Using specific heats, the energy balance relation reduces to

$$W_{\rm b,out} = m_i c_p T_i - m_2 c_v T_2 + m_1 c_v T_1$$

Substituting,

$$1125 = \left(\frac{3009.3}{T_2} - 6.641\right)(5.1926)(298) - \frac{3009.3}{T_2}(3.1156)T_2 + (6.641)(3.1156)(290)$$

It yields

$$T_2 = 315 \text{ K}$$

5-139 The air in an insulated, rigid compressed-air tank is released until the pressure in the tank reduces to a specified value. The final temperature of the air in the tank is to be determined.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process. **2** Air is an ideal gas with constant specific heats. **3** Kinetic and potential energies are negligible. **4** There are no work interactions involved. **5** The tank is well-insulated, and thus there is no heat transfer.

Properties The gas constant of air is 0.287 kPa·m³/kg·K (Table A-1). The specific heats of air at room temperature are $c_p = 1.005$ kJ/kg·K and $c_v = 0.718$ kJ/kg·K (Table A-2a).

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system}$$
$$- m_e = m_2 - m_1$$
$$m_e = m_1 - m_2$$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}} \\ - m_e h_e = m_2 u_2 - m_1 u_1 \\ 0 = m_2 u_2 - m_1 u_1 + m_e h_e \\ 0 = m_2 c_v T_2 - m_1 c_v T_1 + m_e c_p T_e$$



Combining the two balances:

$$0 = m_2 c_v T_2 - m_1 c_v T_1 + (m_1 - m_2) c_n T_e$$

The initial and final masses are given by

$$m_1 = \frac{P_1 \mathbf{V}}{RT_1} = \frac{(4000 \text{ kPa})(0.5 \text{ m}^3)}{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(20 + 273 \text{ K})} = 23.78 \text{ kg}$$
$$m_2 = \frac{P_2 \mathbf{V}}{RT_2} = \frac{(2000 \text{ kPa})(0.5 \text{ m}^3)}{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})T_2} = \frac{3484}{T_2}$$

The temperature of air leaving the tank changes from the initial temperature in the tank to the final temperature during the discharging process. We assume that the temperature of the air leaving the tank is the average of initial and final temperatures in the tank. Substituting into the energy balance equation gives

$$0 = m_2 c_v T_2 - m_1 c_v T_1 + (m_1 - m_2) c_p T_e$$

$$0 = \frac{3484}{T_2} (0.718) T_2 - (23.78)(0.718)(293) + \left(23.78 - \frac{3484}{T_2}\right)(1.005) \left(\frac{293 + T_2}{2}\right)$$

whose solution by trial-error or by an equation solver such as EES is

$$T_2 = 241 \,\mathrm{K} = -32^{\circ}\mathrm{C}$$

5-140 An insulated piston-cylinder device with a linear spring is applying force to the piston. A valve at the bottom of the cylinder is opened, and refrigerant is allowed to escape. The amount of refrigerant that escapes and the final temperature of the refrigerant are to be determined.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process assuming that the state of fluid leaving the device remains constant. 2 Kinetic and potential energies are negligible.

Properties The initial properties of R-134a are (Tables A-11 through A-13)

$$P_{1} = 1.2 \text{ MPa} \begin{cases} \boldsymbol{\nu}_{1} = 0.02423 \text{ m}^{3}/\text{kg} \\ \boldsymbol{\nu}_{1} = 325.03 \text{ kJ/kg} \\ \boldsymbol{h}_{1} = 354.11 \text{ kJ/kg} \end{cases}$$

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:
$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_e = m_1 - m_2$$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}}$$
$$W_{\text{b in}} - m_e h_e = m_2 u_2 - m_1 u_1 \text{ (since } Q \cong ke \cong pe \cong 0)$$

The initial mass and the relations for the final and exiting masses are

$$m_{1} = \frac{\mathbf{V}_{1}}{\mathbf{v}_{1}} = \frac{0.8 \text{ m}^{3}}{0.02423 \text{ m}^{3}/\text{kg}} = 33.02 \text{ kg}$$
$$m_{2} = \frac{\mathbf{V}_{2}}{\mathbf{v}_{2}} = \frac{0.5 \text{ m}^{3}}{v_{2}}$$
$$m_{e} = m_{1} - m_{2} = 33.02 - \frac{0.5 \text{ m}^{3}}{v_{2}}$$



Noting that the spring is linear, the boundary work can be determined from

$$W_{\rm b,in} = \frac{P_1 + P_2}{2} (\mathbf{V}_1 - \mathbf{V}_2) = \frac{(1200 + 600) \,\text{kPa}}{2} (0.8 - 0.5) \text{m}^3 = 270 \,\text{kJ}$$

Substituting the energy balance,

$$270 - \left(33.02 - \frac{0.5 \text{ m}^3}{\nu_2}\right)h_e = \left(\frac{0.5 \text{ m}^3}{\nu_2}\right)u_2 - (33.02 \text{ kg})(325.03 \text{ kJ/kg}) \quad (\text{Eq. 1})$$

where the enthalpy of exiting fluid is assumed to be the average of initial and final enthalpies of the refrigerant in the cylinder. That is,

$$h_e = \frac{h_1 + h_2}{2} = \frac{(354.11 \,\mathrm{kJ/kg}) + h_2}{2}$$

Final state properties of the refrigerant (h_2 , u_2 , and v_2) are all functions of final pressure (known) and temperature (unknown). The solution may be obtained by a trial-error approach by trying different final state temperatures until Eq. (1) is satisfied. Or solving the above equations simultaneously using an equation solver with built-in thermodynamic functions such as EES, we obtain

$$T_2 =$$
 96.8°C, $m_e =$ **22.47 kg**, $h_2 =$ 336.20 kJ/kg,
 $u_2 =$ 307.77 kJ/kg, $v_2 =$ 0.04739 m³/kg, $m_2 =$ 10.55 kg

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5-141 Steam at a specified state is allowed to enter a piston-cylinder device in which steam undergoes a constant pressure expansion process. The amount of mass that enters and the amount of heat transfer are to be determined.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid entering the device remains constant. **2** Kinetic and potential energies are negligible.

Properties The properties of steam at various states are (Tables A-4 through A-6)

$$\begin{aligned}
 \nu_1 &= \frac{\nu_1}{m_1} = \frac{0.1 \text{ m}^3}{0.6 \text{ kg}} = 0.16667 \text{ m}^3/\text{kg} \\
 P_2 &= P_1 \\
 P_1 &= 800 \text{ kPa} \\
 \nu_1 &= 0.16667 \text{ m}^3/\text{kg} \\
 \nu_2 &= 800 \text{ kPa} \\
 V_2 &= 0.29321 \text{ m}^3/\text{kg} \\
 T_2 &= 250^{\circ}\text{C} \quad \int u_2 &= 2715.9 \text{ kJ/kg} \\
 P_i &= 5 \text{ MPa} \\
 T_i &= 500^{\circ}\text{C} \\
 P_i &= 3434.7 \text{ kJ/kg}
 \end{aligned}$$

Analysis (*a*) We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_i = m_2 - m_1$$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{potential, etc. energies}}$$

$$\underbrace{Q_{\text{in}} - W_{\text{b.out}} + m_i h_i}_{\text{potential, etc. energies}} = \underbrace{\Delta E_{\text{system}}}_{\text{potential, etc. energies}}$$

Noting that the pressure remains constant, the boundary work is determined from

$$W_{b,out} = P(V_2 - V_1) = (800 \text{ kPa})(2 \times 0.1 - 0.1)\text{m}^3 = 80 \text{ kJ}$$

The final mass and the mass that has entered are

$$m_2 = \frac{V_2}{V_2} = \frac{0.2 \text{ m}^3}{0.29321 \text{ m}^3/\text{kg}} = 0.682 \text{ kg}$$

 $m_i = m_2 - m_1 = 0.682 - 0.6 = 0.082 \text{ kg}$

(b) Finally, substituting into energy balance equation

$$Q_{\rm in} - 80 \,\text{kJ} + (0.082 \,\text{kg})(3434.7 \,\text{kJ/kg}) = (0.682 \,\text{kg})(2715.9 \,\text{kJ/kg}) - (0.6 \,\text{kg})(2004.4 \,\text{kJ/kg})$$

 $Q_{\rm in} = 447.9 \,\text{kJ}$

5-142 Steam is supplied from a line to a piston-cylinder device equipped with a spring. The final temperature (and quality if appropriate) of the steam in the cylinder and the total work produced as the device is filled are to be determined.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. **2** Kinetic and potential energies are negligible. **3** The process is adiabatic.

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy *h* and internal energy *u*, respectively, and also noting that the initial mass in the system is zero, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance: $m_{in} - m_{out} = \Delta m_{system} \longrightarrow m_i = m_2$

Energy balance:



Combining the two balances:

$$W_{h \text{ out}} = m_2(h_i - u_2)$$



Because of the spring, the relation between the pressure and volume is a linear relation. According to the data in the problem statement,

$$P-300=\frac{2700}{5}V$$

The final vapor volume is then

$$V_2 = \frac{5}{2700} (1500 - 300) = 2.222 \text{ m}^3$$

The work needed to compress the spring is

$$W_{b,\text{out}} = \int P d\mathbf{V} = \int_{0}^{V_2} \left(\frac{2700}{5}\mathbf{V} + 300\right) d\mathbf{V} = \frac{2700}{2 \times 5}\mathbf{V}_2^2 + 300\mathbf{V}_2 = 270 \times 2.222^2 + 300 \times 2.222 = 2000 \text{ kJ}$$

The enthalpy of steam at the inlet is

$$\left. \begin{array}{l} P_i = 1500 \, \mathrm{kPa} \\ T_i = 200^\circ \mathrm{C} \end{array} \right\} h_i = 2796.0 \, \mathrm{kJ/kg} \quad (\mathrm{Table A-6}) \end{array}$$

Substituting the information found into the energy balance equation gives

$$W_{b,\text{out}} = m_2(h_i - u_2) \longrightarrow W_{b,\text{out}} = \frac{V_2}{V_2}(h_i - u_2) \longrightarrow 2000 = \frac{2.222}{V_2}(2796.0 - u_2)$$

Using an iterative solution of this equation with steam tables gives (or better yet using EES software)

$$T_2 = 233.2^{\circ}\text{C}$$

 $u_2 = 2664.8 \text{ kJ/kg}$
 $v_2 = 0.1458 \text{ m}^3/\text{kg}$

5-143 Air is supplied from a line to a piston-cylinder device equipped with a spring. The final temperature of the steam in the cylinder and the total work produced as the device is filled are to be determined.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. 2 Kinetic and potential energies are negligible. 3 The process is adiabatic. 4 Air is an ideal gas with constant specific heats.

Properties The gas constant of air is 0.287 kPa·m³/kg·K (Table A-1). The specific heats of air at room temperature are $c_p = 1.005 \text{ kJ/kg}\cdot\text{K}$ and $c_v = 0.718 \text{ kJ/kg}\cdot\text{K}$ (Table A-2a).

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, and also noting that the initial mass in the system is zero, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance: $m_{in} - m_{out} = \Delta m_{system} \longrightarrow m_i = m_2$

Energy balance:



Combining the two balances:

$$W_{h,\text{out}} = m_2(h_i - u_2)$$



Because of the spring, the relation between the pressure and volume is a linear relation. According to the data in the problem statement,

$$P-300 = \frac{2700}{5} V$$

The final air volume is then

$$V_2 = \frac{5}{2700} (2000 - 300) = 3.148 \text{ m}^3$$

The work needed to compress the spring is

$$W_{b,\text{out}} = \int P d\boldsymbol{\mathcal{V}} = \int_{0}^{\boldsymbol{\mathcal{V}}_{2}} \left(\frac{2700}{5}\boldsymbol{\mathcal{V}} + 300\right) d\boldsymbol{\mathcal{V}} = \frac{2700}{2 \times 5}\boldsymbol{\mathcal{V}}_{2}^{2} + 300\boldsymbol{\mathcal{V}}_{2} = 270 \times 3.148^{2} + 300 \times 3.148 = 3620 \text{ kJ}$$

Substituting the information found into the energy balance equation gives

$$W_{b,\text{out}} = m_2(h_i - u_2)$$

$$W_{b,\text{out}} = \frac{P_2 V_2}{RT_2} (c_p T_i - c_v T_2)$$

$$3620 = \frac{2000 \times 3.148}{(0.287)T_2} (1.005 \times 600 - 0.718 \times T_2)$$

The final temperature is then

$$T_2 = 682.9 \text{ K} = 409.9^{\circ}\text{C}$$

Review Problems

5-144 Carbon dioxide flows through a throttling valve. The temperature change of CO_2 is to be determined if CO_2 is assumed an ideal gas and a real gas.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Heat transfer to or from the fluid is negligible. 4 There are no work interactions involved.

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the throttling value as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\dot{E}_{in} - \dot{E}_{out} = \Delta \dot{E}_{system}^{70 \text{ (steady)}} = 0$$

$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{m}h_1 = \dot{m}h_2$$

$$h_1 = h_2$$

$$CO_2$$

$$5 \text{ MPa}$$

$$100 \text{ kPa}$$

since $\dot{Q} \cong \dot{W} = \Delta ke \cong \Delta pe \cong 0$.

(*a*) For an ideal gas, h = h(T), and therefore,

$$T_2 = T_1 = 100^{\circ}C \longrightarrow \Delta T = T_1 - T_2 = \mathbf{0}^{\circ}\mathbf{C}$$

(b) We obtain real gas properties of CO₂ from EES software as follows

$$P_{1} = 5 \text{ MPa} T_{1} = 100^{\circ}\text{C}$$

$$h_{1} = 34.77 \text{ kJ/kg} P_{2} = 100 \text{ kPa} h_{2} = h_{1} = 34.77 \text{ kJ/kg} T_{2} = 66.0^{\circ}\text{C}$$

Note that EES uses a different reference state from the textbook for CO₂ properties. The temperature difference in this case becomes

$$\Delta T = T_1 - T_2 = 100 - 66.0 = 34.0^{\circ}C$$

That is, the temperature of CO₂ decreases by 34°C in a throttling process if its real gas properties are used.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible.

Properties The gas constant of helium is 2.0769 kPa.m³/kg.K. The constant pressure specific heat of air at room temperature is $c_p = 5.1926 \text{ kJ/kg} \cdot ^{\circ}\text{C}$ (Table A-2*a*).

Analysis (a) We take the pipe in which the argon is heated as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as



Substituting,

 $\dot{Q}_{out} = \dot{m}c_p(T_1 - T_2) = (8 \text{ kg/s})(5.1926 \text{ kJ/kg} \cdot \text{K})(427 - 27)\text{K} = 16,620 \text{ kW}$

(b) The exit specific volume and the volume flow rate are

$$v_2 = \frac{RT_2}{P_2} = \frac{(2.0769 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(300 \text{ K})}{100 \text{ kPa}} = 6.231 \text{ m}^3/\text{kg}$$

 $\dot{v}_2 = \dot{m}v_2 = (8 \text{ kg/s})(6.231 \text{ m}^3/\text{kg}) = 49.85 \text{ m}^3/\text{s}$

Assumptions 1 Water is supplied and discharged steadily. 2 The rate of evaporation of water is negligible. 3 No water is supplied or removed through other means.

Analysis The conservation of mass principle applied to the pool requires that the rate of increase in the amount of water in the pool be equal to the difference between the rate of supply of water and the rate of discharge. That is,

$$\frac{dm_{\text{pool}}}{dt} = \dot{m}_i - \dot{m}_e \qquad \rightarrow \qquad \dot{m}_i = \frac{dm_{\text{pool}}}{dt} + \dot{m}_e \qquad \rightarrow \qquad \dot{V}_i = \frac{dV_{\text{pool}}}{dt} + \dot{V}_e$$

since the density of water is constant and thus the conservation of mass is equivalent to conservation of volume. The rate of discharge of water is

$$\dot{V}_e = A_e V_e = (\pi D^2/4) V_e = [\pi (0.07 \text{ m})^2/4] (4 \text{ m/s}) = 0.01539 \text{ m}^3/\text{s}$$

The rate of accumulation of water in the pool is equal to the crosssection of the pool times the rate at which the water level rises,

$$\frac{d\mathbf{V}_{\text{pool}}}{dt} = A_{\text{cross-section}} V_{\text{level}} = (6 \text{ m} \times 9 \text{ m})(0.025 \text{ m/min}) = 1.35 \text{ m}^3/\text{min} = 0.0225 \text{ m}^3/\text{s}$$

Substituting, the rate at which water is supplied to the pool is determined to be

$$\dot{\mathbf{V}}_i = \frac{d\mathbf{V}_{\text{pool}}}{dt} + \dot{\mathbf{V}}_e = 0.0225 + 0.01539 = 0.0379 \text{ m}^3/\text{s}$$

Therefore, water is supplied at a rate of 0.0379 $\text{m}^3/\text{s} = 37.9 \text{ L/s}$.

5-147 Air is accelerated in a nozzle. The density of air at the nozzle exit is to be determined.

Assumptions Flow through the nozzle is steady.

Properties The density of air is given to be 4.18 kg/m³ at the inlet.

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. Then,

$$\dot{m}_1 = \dot{m}_2$$

$$\rho_1 A_1 V_1 = \rho_2 A_2 V_2$$

$$\rho_2 = \frac{A_1}{A_2} \frac{V_1}{V_2} \rho_1 = 2 \frac{120 \text{ m/s}}{380 \text{ m/s}} (4.18 \text{ kg/m}^3) = 2.64 \text{ kg/m}^3$$

Discussion Note that the density of air decreases considerably despite a decrease in the cross-sectional area of the nozzle.





<u>AIR</u>

1

2

5-148E A heat exchanger that uses hot air to heat cold water is considered. The total flow power and the flow works for both the air and water streams are to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Kinetic and potential energy changes are negligible. **3** There are no work interactions. **4** Heat loss from the device to the surroundings is negligible and thus heat transfer from the hot fluid is equal to the heat transfer to the cold fluid. **5** Air is an ideal gas with constant specific heats at room temperature.

Properties The gas constant of air is $0.3704 \text{ psia.ft}^3/\text{lbm.R} = 0.06855 \text{ Btu/lbm.R}$ (Table A-1E). The specific volumes of water at the inlet and exit are (Table A-4E)

$$\begin{array}{l} P_3 = 20 \text{ psia} \\ T_3 = 50^{\circ}\text{F} \end{array} \right\} \boldsymbol{v}_3 \cong \boldsymbol{v}_{f @ 50^{\circ}\text{F}} = 0.01602 \text{ ft}^3/\text{lbm} \\ P_4 = 17 \text{ psia} \\ T_4 = 90^{\circ}\text{F} \end{array} \right\} \boldsymbol{v}_4 \cong \boldsymbol{v}_{f @ 90^{\circ}\text{F}} = 0.01610 \text{ ft}^3/\text{lbm}$$

Analysis The specific volume of air at the inlet and the mass flow rate are

$$\boldsymbol{v}_{1} = \frac{RT_{1}}{P_{1}} = \frac{(0.3704 \text{ psia} \cdot \text{ft}^{3}/\text{lbm} \cdot \text{R})(200 + 460 \text{ R})}{20 \text{ psia}} = 12.22 \text{ ft}^{3}/\text{lbm}$$
$$\dot{\boldsymbol{m}} = \frac{\dot{\boldsymbol{V}}_{1}}{\boldsymbol{v}_{1}} = \frac{(100/60) \text{ ft}^{3}/\text{s}}{12.22 \text{ ft}^{3}/\text{lbm}} = 0.1364 \text{ lbm/s}$$



Combining the flow work expression with the ideal gas equation of state gives

$$w_{\text{flow}} = P_2 \boldsymbol{v}_2 - P_1 \boldsymbol{v}_1 = R(T_2 - T_1) = (0.06855 \text{ Btu/lbm} \cdot \text{R})(100 - 200)\text{R} = -6.855 \text{ Btu/lbm}$$

The flow work of water is

$$w_{\text{flow}} = P_4 \boldsymbol{v}_4 - P_3 \boldsymbol{v}_3$$

= $\left[(17 \text{ psia})(0.01610 \text{ ft}^3/\text{lbm}) - (20 \text{ psia})(0.01602 \text{ ft}^3/\text{lbm}) \left(\frac{1 \text{ Btu}}{5.404 \text{ psia} \cdot \text{ft}^3} \right) \right]$

= -0.00864 Btu/Ibm

The net flow power for the heat exchanger is

$$W_{\text{flow}} = \dot{m}_{\text{air}} w_{\text{flow}} + \dot{m}_{\text{air}} w_{\text{flow}}$$

= (0.1364 lbm/s)(-6.855 Btu/lbm) + (0.5 lbm/s)(-0.00864 Btu/lbm)
= -0.9393 Btu/s $\left(\frac{1 \text{ hp}}{0.7068 \text{ Btu/s}}\right)$ = -**1.329 hp**

5-149 An air compressor consumes 6.2 kW of power to compress a specified rate of air. The flow work required by the compressor is to be compared to the power used to increase the pressure of the air.

Assumptions 1 Flow through the compressor is steady. 2 Air is an ideal gas.

Properties The gas constant of air is 0.287 kPa·m³/kg·K (Table A-1).

Analysis The specific volume of the air at the inlet is

$$v_1 = \frac{RT_1}{P_1} = \frac{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(20 + 273 \text{ K})}{120 \text{ kPa}} = 0.7008 \text{ m}^3/\text{kg}$$

The mass flow rate of the air is

$$\dot{m} = \frac{V_1}{V_1} = \frac{0.015 \text{ m}^3/\text{s}}{0.7008 \text{ m}^3/\text{kg}} = 0.02140 \text{ kg/s}$$

Combining the flow work expression with the ideal gas equation of state gives the flow work as

$$w_{\text{flow}} = P_2 \boldsymbol{v}_2 - P_1 \boldsymbol{v}_1 = R(T_2 - T_1) = (0.287 \text{ kJ/kg} \cdot \text{K})(300 - 20)\text{K} = 80.36 \text{ kJ/kg}$$

The flow power is

$$\dot{W}_{\text{flow}} = \dot{m}w_{\text{flow}} = (0.02140 \text{ kg/s})(80.36 \text{ kJ/kg}) = 1.72 \text{ kW}$$

The remainder of compressor power input is used to increase the pressure of the air:

$$\dot{W} = \dot{W}_{\text{total in}} - \dot{W}_{\text{flow}} = 6.2 - 1.72 = 4.48 \text{ kW}$$

5-150 Steam expands in a turbine whose power production is 9000 kW. The rate of heat lost from the turbine is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible.

Properties From the steam tables (Tables A-6 and A-4)

$$\begin{array}{c} P_1 = 1.6 \text{ MPa} \\ T_1 = 350^{\circ}\text{C} \end{array} \right\} h_1 = 3146.0 \text{ kJ/kg} \\ T_2 = 30^{\circ}\text{C} \\ x_2 = 1 \end{array} \right\} h_2 = 2555.6 \text{ kJ / kg}$$

Analysis We take the turbine as the system, which is a control volume since mass crosses the boundary. Noting that there is one inlet and one exiti the energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{E}_{in} = \dot{m}_{2}h_{2} + \dot{W}_{out} + \dot{Q}_{out}}_{\dot{Q}_{out} = \dot{m}(h_{1} - h_{2}) - \dot{W}_{out}}$$

Substituting,

$$Q_{\text{out}} = (16 \text{ kg/s})(3146.0 - 2555.6) \text{ kJ/kg} - 9000 \text{ kW} = 446.4 \text{ kW}$$





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5-151E Nitrogen gas flows through a long, constant-diameter adiabatic pipe. The velocities at the inlet and exit are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Nitrogen is an ideal gas with constant specific heats. 3 Potential energy changes are negligible. 4 There are no work interactions. 5 The re is no heat transfer from the nitrogen.

Properties The specific heat of nitrogen at the room temperature iss $c_p = 0.248$ Btu/lbm·R (Table A-2Ea).

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the pipe as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}^{70 \text{ (steady)}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0$$

$$\frac{\dot{E}_{\text{in}} = \dot{E}_{\text{out}}}{\dot{E}_{\text{in}} = \dot{E}_{\text{out}}} = \underbrace{100 \text{ psia}}_{120^{\circ}\text{F}} \xrightarrow{N_2} 50 \text{ psia}}_{120^{\circ}\text{F}} \xrightarrow{70^{\circ}\text{F}}_{70^{\circ}\text{F}}}$$

$$\frac{\dot{H}_{1} + V_{1}^{2} / 2}{2} = h_{2} + V_{2}^{2} / 2$$

$$\frac{V_{1}^{2} - V_{2}^{2}}{2} = c_{p} (T_{2} - T_{1})$$

Combining the mass balance and ideal gas equation of state yields

$$\dot{m}_{1} = \dot{m}_{2}$$

$$\frac{A_{1}V_{1}}{v_{1}} = \frac{A_{2}V_{2}}{v_{2}}$$

$$V_{2} = \frac{A_{1}}{A_{2}}\frac{v_{2}}{v_{1}}V_{1} = \frac{v_{2}}{v_{1}}V_{1} = \frac{T_{2}}{T_{1}}\frac{P_{1}}{P_{2}}V_{1}$$

Substituting this expression for V_2 into the energy balance equation gives

$$V_{1} = \left[\frac{2c_{p}(T_{2} - T_{1})}{1 - \left(\frac{T_{2}}{T_{1}}\frac{P_{1}}{P_{2}}\right)^{2}}\right]^{0.5} = \left[\frac{2(0.248)(70 - 120)}{1 - \left(\frac{530}{580}\frac{100}{50}\right)^{2}}\left(\frac{25,037 \text{ ft}^{2}/\text{s}^{2}}{1 \text{ Btu/lbm}}\right)\right]^{0.5} = 515 \text{ ft/s}$$

The velocity at the exit is

$$V_2 = \frac{T_2}{T_1} \frac{P_1}{P_2} V_1 = \frac{530}{580} \frac{100}{50} 515 = 941 \,\text{ft/s}$$

70°F

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 The heat losses from the water is negligible.

Properties The specific heat and the density of water are taken to be $c_p = 4.18 \text{ kJ/kg} \cdot ^{\circ}\text{C}$ and $\rho = 1 \text{ kg/L}$ (Table A-3).

Analysis We take the pipe in which water is heated as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as



The mass flow rate of the water is

$$\dot{m} = \rho V = (1 \text{ kg/L})(0.1 \text{ L/s}) = 0.1 \text{ kg/s}$$

Substituting into the energy balance equation and solving for the current gives

$$I = \frac{mc_p (T_2 - T_1)}{\mathbf{V}} = \frac{(0.1 \text{ kg/s})(4.18 \text{ kJ/kg} \cdot \text{K})(30 - 18)\text{K}}{110 \text{ V}} \left(\frac{1000 \text{ VI}}{1 \text{ kJ/s}}\right) = 45.6 \text{ A}$$

5-153 Water is boiled at $T_{\text{sat}} = 100^{\circ}$ C by an electric heater. The rate of evaporation of water is to be determined.

Assumptions **1** Steady operating conditions exist. **2** Heat losses from the outer surfaces of the water tank are negligible.

Properties The enthalpy of vaporization of water at 100°C is $h_{fg} = 2256.4 \text{ kJ/kg}$ (Table A-4).

Analysis Noting that the enthalpy of vaporization represents the amount of energy needed to vaporize a unit mass of a liquid at a specified temperature, the rate of evaporation of water is determined to be

$$\dot{m}_{\text{evaporation}} = \frac{\dot{W}_{\text{e,boiling}}}{h_{fg}} = \frac{3 \text{ kJ/s}}{2256.4 \text{ kJ/kg}}$$
$$= 0.00133 \text{ kg/s} = 4.79 \text{ kg/h}$$



Assumptions **1** This is a steady-flow process since there is no change with time. **2** Kinetic and potential energy changes are negligible. **3** There are no work and heat interactions.

Analysis We take the pipe in which steam flows as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{system}^{\pi 0 \text{ (steady)}}}_{\text{Potential, etc. energies}} = 0$$

$$\frac{\dot{E}_{in} = \dot{E}_{out}}{\dot{m}h_1 = \dot{m}h_2} = \underbrace{b_{out}}_{h_1 = h_2} = \underbrace{b_{out}$$

The properties of the steam at the inlet and exit are (Table A-6)

$$P = 1400 \text{ kPa}$$

$$T = 350^{\circ}\text{C}$$

$$\begin{cases}
\nu_{1} = 0.20029 \text{ m}^{3}/\text{kg} \\
h_{1} = 3150.1 \text{ kJ/kg}
\end{cases}$$

$$P_{2} = 1000 \text{ kPa} \\
h_{2} = h_{1} = 3150.1 \text{ kJ/kg}
\end{cases}$$

$$\nu_{2} = 0.28064 \text{ m}^{3}/\text{kg}$$

The mass flow rate is

$$\dot{m} = \frac{A_{\rm l}V_{\rm l}}{\nu_{\rm l}} = \frac{\pi D_{\rm l}^2}{4} \frac{V_{\rm l}}{\nu_{\rm l}} = \frac{\pi (0.15\,{\rm m})^2}{4} \frac{10\,{\rm m/s}}{0.20029\,{\rm m}^3/{\rm kg}} = 0.8823\,{\rm kg/s}$$

The outlet velocity will then be

$$V_2 = \frac{\dot{m}\boldsymbol{v}_2}{A_2} = \frac{4\dot{m}\boldsymbol{v}_2}{\pi D_2^2} = \frac{4(0.8823 \text{ kg/s})(0.28064 \text{ m}^3/\text{kg})}{\pi (0.10 \text{ m})^2} = 31.53 \text{ m/s}$$

5-155 The mass flow rate of a compressed air line is divided into two equal streams by a T-fitting in the line. The velocity of the air at the outlets and the rate of change of flow energy (flow power) across the T-fitting are to be determined.

Assumptions 1 Air is an ideal gas with constant specific heats. 2 The flow is steady. 3 Since the outlets are identical, it is presumed that the flow divides evenly between the two.

Properties The gas constant of air is
$$R = 0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K}$$
 (Table A-1).

Analysis The specific volumes of air at the inlet and outlets are

$$\boldsymbol{v}_{1} = \frac{RT_{1}}{P_{1}} = \frac{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(40 + 273 \text{ K})}{1600 \text{ kPa}} = 0.05614 \text{ m}^{3}/\text{kg}$$
$$\boldsymbol{v}_{2} = \boldsymbol{v}_{3} = \frac{RT_{2}}{P_{2}} = \frac{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(36 + 273 \text{ K})}{1400 \text{ kPa}} = 0.06335 \text{ m}^{3}/\text{kg}$$

Assuming an even division of the inlet flow rate, the mass balance can be written as

$$\frac{A_1V_1}{v_1} = 2\frac{A_2V_2}{v_2} \longrightarrow V_2 = V_3 = \frac{A_1}{A_2}\frac{v_2}{v_1}\frac{V_1}{2} = \frac{0.06335}{0.05614}\frac{50}{2} = 28.21 \text{ m/s}$$

The mass flow rate at the inlet is

$$\dot{m}_1 = \frac{A_1 V_1}{\nu_1} = \frac{\pi D_1^2}{4} \frac{V_1}{\nu_1} = \frac{\pi (0.025 \text{ m})^2}{4} \frac{50 \text{ m/s}}{0.05614 \text{ m}^3/\text{kg}} = 0.4372 \text{ kg/s}$$

while that at the outlets is

$$\dot{m}_2 = \dot{m}_3 = \frac{\dot{m}_1}{2} = \frac{0.4372 \text{ kg/s}}{2} = 0.2186 \text{ kg/s}$$

Substituting the above results into the flow power expression produces

$$W_{\text{flow}} = 2\dot{m}_2 P_2 v_2 - \dot{m}_1 P_1 v_1$$

= 2(0.2186 kg/s)(1400 kPa)(0.06335 m³/kg) - (0.4372 kg/s)(1600 kPa)(0.05614 m³/kg)
= -**0.496 kW**



5-156 Air flows through a non-constant cross-section pipe. The inlet and exit velocities of the air are to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Potential energy change is negligible. **3** There are no work interactions. **4** The device is adiabatic and thus heat transfer is negligible. **5** Air is an ideal gas with constant specific heats.

Properties The gas constant of air is R = 0.287 kPa.m³/kg.K. Also, $c_p = 1.005$ kJ/kg.K for air at room temperature (Table A-2)

Analysis We take the pipe as the system, which is a control volume since mass crosses the boundary. The mass and energy balances for this steady-flow system can be expressed in the rate form as

 $c_p T_1 + \frac{V_1^2}{2} = c_p T_2 + \frac{V_2^2}{2} + q_{\text{out}}$

Z0 (steady)

Mass balance:



$$\dot{m}_{\rm in} - \dot{m}_{\rm out} = \Delta \dot{m}_{\rm system} = 0$$

$$\dot{m}_{\rm in} = \dot{m}_{\rm out} \longrightarrow \rho_1 A_1 V_1 = \rho_2 A_2 V_2 \longrightarrow \frac{P_1}{RT_1} \frac{\pi D_1^2}{4} V_1 = \frac{P_2}{RT_2} \frac{\pi D_2^2}{4} V_2 \longrightarrow \frac{P_1}{T_1} D_1^2 V_1 = \frac{P_2}{T_2} D_2^2 V_2$$

Energy balance:

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{by heat, work, and mass} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0 \quad \text{since } \dot{W} \cong \Delta pe \cong 0)$$

$$\dot{E}_{internal} = \frac{\dot{E}_{internal}}{\dot{E}_{internal}} = \frac{\dot{E}_{internal}}{\dot{E}_{inter$$

$$\dot{E}_{in} = \dot{E}_{out} \longrightarrow h_1 + \frac{V_1^2}{2} = h_2 + \frac{V_2^2}{2} + q_{out}$$

or

Assuming inlet diameter to be 1.4 m and the exit diameter to be 1.0 m, and substituting given values into mass and energy balance equations

$$\left(\frac{200 \,\mathrm{kPa}}{338 \,\mathrm{K}}\right) (1.4 \,\mathrm{m})^2 V_1 = \left(\frac{175 \,\mathrm{kPa}}{333 \,\mathrm{K}}\right) (1.0 \,\mathrm{m})^2 V_2 \tag{1}$$

$$(1.005 \text{ kJ/kg.K})(338 \text{ K}) + \frac{V_1^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) = (1.005 \text{ kJ/kg.K})(333 \text{ K}) + \frac{V_2^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) + 3.3 \text{ kJ/kg}$$
(2)

There are two equations and two unknowns. Solving equations (1) and (2) simultaneously using an equation solver such as EES, the velocities are determined to be

 $V_1 =$ 29.9 m/s $V_2 =$ 66.1 m/s 5-157 Heat is lost from the steam flowing in a nozzle. The exit velocity and the mass flow rate are to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Potential energy change is negligible. **3** There are no work interactions.

Analysis (*a*) We take the steam as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

Energy balance:

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{Rate of net energy transfer} = \Delta \dot{E}_{system}^{70 \text{ (steady)}} = 0$$
Rate of change in internal, kinetic, potential, etc. energies
$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{m} \left(h_1 + \frac{V_1^2}{2} \right) = \dot{m} \left(h_2 + \frac{V_2^2}{2} \right) + \dot{Q}_{out} \quad \text{since} \quad \dot{W} \cong \Delta pe \cong 0$$

$$V_2 = \sqrt{2(h_1 - h_2 - q_{out})}$$

or

The properties of steam at the inlet and exit are (Table A-6)

$$P_{1} = 200 \text{ kPa} T_{1} = 150^{\circ}\text{C}$$

$$h_{1} = 2769.1 \text{ kJ/kg} P_{2} = 75 \text{ kPa} v_{2} = 2.2172 \text{ m}^{3}/\text{kg} sat. vap.$$

$$h_{2} = 2662.4 \text{ kJ/kg}$$

Substituting,

$$V_2 = \sqrt{2(h_1 - h_2 - q_{\text{out}})} = \sqrt{2(2769.1 - 2662.4 - 26)\text{kJ/kg} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2}\right)} = 401.7 \text{ m/s}$$

(b) The mass flow rate of the steam is

$$\dot{m} = \frac{1}{v_2} A_2 V_2 = \frac{1}{2.2172 \text{ m}^3/\text{kg}} (0.001 \text{ m}^2)(401.7 \text{ m/s}) = 0.181 \text{ kg/s}$$



q

Assumptions **1** Steady operating conditions exist. **2** Heat losses from the outer surfaces of the boiler are negligible.

Properties The enthalpy of vaporization of water at 150°C is $h_{fg} = 2113.8 \text{ kJ/kg}$ (Table A-4).

Analysis The rate of heat transfer to water is given to be 74 kJ/s. Noting that the enthalpy of vaporization represents the amount of energy needed to vaporize a unit mass of a liquid at a specified temperature, the rate of evaporation of water is determined to be

$$\dot{m}_{\text{evaporation}} = \frac{Q_{\text{boiling}}}{h_{fg}} = \frac{74 \text{ kJ/s}}{2113.8 \text{ kJ/kg}} = 0.0350 \text{ kg/s}$$



5-159 Cold water enters a steam generator at 20°C, and leaves as saturated vapor at $T_{sat} = 200$ °C. The fraction of heat used to preheat the liquid water from 20°C to saturation temperature of 200°C is to be determined.

Assumptions 1 Steady operating conditions exist. 2 Heat losses from the steam generator are negligible. 3 The specific heat of water is constant at the average temperature.

Properties The heat of vaporization of water at 200°C is $h_{fg} = 1939.8 \text{ kJ/kg}$ (Table A-4), and the specific heat of liquid water is c = 4.18 kJ/kg.°C (Table A-3).

Analysis The heat of vaporization of water represents the amount of heat needed to vaporize a unit mass of liquid at a specified temperature. Using the average specific heat, the amount of heat transfer needed to preheat a unit mass of water from 20°C to 200°C is

$$q_{\text{preheating}} = c\Delta T$$

= (4.18 kJ/kg · °C)(200 - 20)°C
= 752.4 kJ/kg

and

 $q_{\text{total}} = q_{\text{boiling}} + q_{\text{preheating}}$ = 1939.8 + 364.1 = 2692.2 kJ/kg

Therefore, the fraction of heat used to preheat the water is

Fraction to preheat =
$$\frac{q_{\text{preheating}}}{q_{\text{total}}} = \frac{752.4}{2692.2} = 0.2795$$
 (or 28.0%)



5-160 Cold water enters a steam generator at 20°C and is boiled, and leaves as saturated vapor at boiler pressure. The boiler pressure at which the amount of heat needed to preheat the water to saturation temperature that is equal to the heat of vaporization is to be determined.

Assumptions Heat losses from the steam generator are negligible.

Properties The enthalpy of liquid water at 20°C is 83.91 kJ/kg. Other properties needed to solve this problem are the heat of vaporization h_{fg} and the enthalpy of saturated liquid at the specified temperatures, and they can be obtained from Table A-4.

Analysis The heat of vaporization of water represents the amount of heat needed to vaporize a unit mass of liquid at a specified temperature, and Δh represents the amount of heat needed to preheat a unit mass of water from 20°C to the saturation temperature. Therefore,

$$q_{\text{preheating}} = q_{\text{boiling}}$$

$$(h_{f@T_{\text{sat}}} - h_{f@20^{\circ}\text{C}}) = h_{fg@T_{\text{sat}}}$$

$$h_{f@T_{\text{sat}}} - 83.91 \text{ kJ/kg} = h_{fg@T_{\text{sat}}} \rightarrow h_{f@T_{\text{sat}}} - h_{fg@T_{\text{sat}}} = 83.91 \text{ kJ/kg}$$

The solution of this problem requires choosing a boiling temperature, reading $h_{\rm f}$ and $h_{\rm fg}$ at that temperature, and substituting the values into the relation above to see if it is satisfied. By trial and error, (Table A-4)

At 310°C: $h_{f@T_{sat}} - h_{fg@T_{sat}} = 1402.0 - 1325.9 = 76.1 \text{ kJ/kg}$ At 315°C: $h_{f(\hat{w})T_{sat}} - h_{fg(\hat{w})T_{sat}} = 1431.6 - 1283.4 = 148.2 \text{ kJ/kg}$



The temperature that satisfies this condition is determined from the two values above by interpolation to be 310.6°C. The saturation pressure corresponding to this temperature is 9.94 MPa.

5-161 An ideal gas expands in a turbine. The volume flow rate at the inlet for a power output of 350 kW is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. **3** The device is adiabatic and thus heat transfer is negligible.

Properties The properties of the ideal gas are given as R = 0.30 kPa.m³/kg.K, $c_p = 1.13$ kJ/kg.°C, $c_v = 0.83$ kJ/kg.°C.

Analysis We take the turbine as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\frac{\dot{E}_{in} - \dot{E}_{out}}{h_{i} - h_{2}} = \frac{\dot{\Delta}\dot{E}_{system}^{70} (steady)}{h_{1} - h_{2}} = \frac{\dot{\Delta}\dot{E}_{system}^{70} (steady)}{c_{p}(T_{1} - T_{2})} = \frac{350 \text{ kW}}{(1.13 \text{ kJ/kg.K})(1200 - 700)\text{K}} = 0.6195 \text{ kg/s}$$

$$P_{1} = 900 \text{ kPa}$$

$$T_{1} = 1200 \text{ K}$$

$$P_{1} = 900 \text{ kPa}$$

$$T_{1} = 1200 \text{ K}$$

$$T_{1} = 1200 \text{ K}$$

$$T_{1} = 1200 \text{ K}$$

The inlet specific volume and the volume flow rate are

$$v_1 = \frac{RT_1}{P_1} = \frac{(0.3 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(1200 \text{ K})}{900 \text{ kPa}} = 0.4 \text{ m}^3/\text{kg}$$

 $\dot{\boldsymbol{V}} = \dot{\boldsymbol{m}}\boldsymbol{v}_1 = (0.6195 \text{ kg/s})(0.4 \text{ m}^3/\text{kg}) = 0.248 \text{ m}^3/\text{s}$ Thus,



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5-162 Chickens are to be cooled by chilled water in an immersion chiller. The rate of heat removal from the chicken and the mass flow rate of water are to be determined.

Assumptions 1 Steady operating conditions exist. 2 The thermal properties of chickens and water are constant.

Properties The specific heat of chicken are given to be 3.54 kJ/kg.°C. The specific heat of water is 4.18 kJ/kg.°C (Table A-3).

Analysis (*a*) Chickens are dropped into the chiller at a rate of 500 per hour. Therefore, chickens can be considered to flow steadily through the chiller at a mass flow rate of

 $\dot{m}_{chicken} = (500 \text{ chicken} / h)(2.2 \text{ kg} / \text{ chicken}) = 1100 \text{ kg} / h = 0.3056 \text{ kg} / \text{ s}$

Taking the chicken flow stream in the chiller as the system, the energy balance for steadily flowing chickens can be expressed in the rate form as

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{by heat, work, and mass} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$E_{\text{in}} = E_{\text{out}}$$

$$\dot{m}h_1 = \dot{Q}_{out} + \dot{m}h_2 \quad (\text{since } \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

$$\dot{Q}_{\text{out}} = \dot{Q}_{\text{chicken}} = \dot{m}_{\text{chicken}} c_p (T_1 - T_2)$$

Immersion chilling. 0.5°C

Then the rate of heat removal from the chickens as they are cooled from 15°C to 3°C becomes

$$\dot{Q}_{\text{chicken}} = (\dot{m}c_p \Delta T)_{\text{chicken}} = (0.3056 \text{ kg/s})(3.54 \text{ kJ/kg.}^{\circ} \text{C})(15-3)^{\circ} \text{C} = 13.0 \text{ kW}$$

The chiller gains heat from the surroundings at a rate of 200 kJ/h = 0.0556 kJ/s. Then the total rate of heat gain by the water is

$$\dot{Q}_{\text{water}} = \dot{Q}_{\text{chicken}} + \dot{Q}_{\text{heat gain}} = 13.0 + 0.056 = 13.056 \,\text{kW}$$

Noting that the temperature rise of water is not to exceed 2°C as it flows through the chiller, the mass flow rate of water must be at least

$$\dot{m}_{\text{water}} = \frac{\dot{Q}_{\text{water}}}{(c_n \Delta T)_{\text{water}}} = \frac{13.056 \text{ kW}}{(4.18 \text{ kJ/kg.}^{\circ} \text{ C})(2^{\circ} \text{ C})} = 1.56 \text{ kg/s}$$

If the mass flow rate of water is less than this value, then the temperature rise of water will have to be more than 2°C.

5-163 Chickens are to be cooled by chilled water in an immersion chiller. The rate of heat removal from the chicken and the mass flow rate of water are to be determined.

Assumptions 1 Steady operating conditions exist. 2 The thermal properties of chickens and water are constant. 3 Heat gain of the chiller is negligible.

Properties The specific heat of chicken are given to be 3.54 kJ/kg.°C. The specific heat of water is 4.18 kJ/kg.°C (Table A-3).

Analysis (*a*) Chickens are dropped into the chiller at a rate of 500 per hour. Therefore, chickens can be considered to flow steadily through the chiller at a mass flow rate of

 $\dot{m}_{\text{chicken}} = (500 \text{ chicken} / \text{h})(2.2 \text{ kg} / \text{chicken}) = 1100 \text{ kg} / \text{h} = 0.3056 \text{ kg} / \text{s}$

Taking the chicken flow stream in the chiller as the system, the energy balance for steadily flowing chickens can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{\pi 0 \text{ (steady)}} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{m}h_1 = \dot{Q}_{out} + \dot{m}h_2} \text{ (since } \Delta \text{ke} \cong \Delta \text{pe} \cong 0)}_{\dot{Q}_{out} = \dot{Q}_{chicken} = \dot{m}_{chicken}c_p(T_1 - T_2)}$$



Then the rate of heat removal from the chickens as they are cooled from 15°C to 3°C becomes

$$\dot{Q}_{\text{chicken}} = (\dot{m}c_p \Delta T)_{\text{chicken}} = (0.3056 \text{ kg/s})(3.54 \text{ kJ/kg.}^{\circ} \text{C})(15-3)^{\circ} \text{C} = 13.0 \text{ kW}$$

Heat gain of the chiller from the surroundings is negligible. Then the total rate of heat gain by the water is

$$\dot{Q}_{\text{water}} = \dot{Q}_{\text{chicken}} = 13.0 \text{ kW}$$

Noting that the temperature rise of water is not to exceed 2°C as it flows through the chiller, the mass flow rate of water must be at least

$$\dot{m}_{\text{water}} = \frac{\dot{Q}_{\text{water}}}{(c_p \Delta T)_{\text{water}}} = \frac{13.0 \text{ kW}}{(4.18 \text{ kJ/kg.}^{\circ} \text{ C})(2^{\circ} \text{ C})} = 1.56 \text{ kg/s}$$

If the mass flow rate of water is less than this value, then the temperature rise of water will have to be more than 2°C.

5-164 A regenerator is considered to save heat during the cooling of milk in a dairy plant. The amounts of fuel and money such a generator will save per year are to be determined.

Assumptions 1 Steady operating conditions exist. 2 The properties of the milk are constant.

Properties The average density and specific heat of milk can be taken to be $\rho_{\text{milk}} \cong \rho_{\text{water}} = 1 \text{ kg/L}$ and $c_{p, \text{milk}} = 3.79 \text{ kJ/kg.°C}$ (Table A-3).

Analysis The mass flow rate of the milk is

$$\dot{m}_{\text{milk}} = \rho \boldsymbol{V}_{\text{milk}}$$
$$= (1 \text{ kg/L})(20 \text{ L/s}) = 20 \text{ kg/s}$$
$$= 72,000 \text{ kg/h}$$

Taking the pasteurizing section as the system, the energy balance for this steady-flow system can be expressed in the rate form as





Therefore, to heat the milk from 4 to 72°C as being done currently, heat must be transferred to the milk at a rate of

$$Q_{\text{current}} = [\dot{m}c_{\text{p}}(T_{\text{pasturization}} - T_{\text{refrigeration}})]_{\text{milk}}$$
$$= (20 \text{ kg/s})(3.79 \text{ kJ/kg.ëC})(72 - 4)^{\circ}\text{C} = 5154 \text{ kJ/s}$$

The proposed regenerator has an effectiveness of $\varepsilon = 0.82$, and thus it will save 82 percent of this energy. Therefore,

$$\dot{Q}_{saved} = \varepsilon \dot{Q}_{current} = (0.82)(5154 \text{ kJ/s}) = 4227 \text{ kJ/s}$$

Noting that the boiler has an efficiency of $\eta_{\text{boiler}} = 0.90$, the energy savings above correspond to fuel savings of

FuelSaved =
$$\frac{Q_{\text{saved}}}{\eta_{\text{boiler}}} = \frac{4227 \text{ kJ/s}}{0.90} \frac{1 \text{ therm}}{105,500 \text{ kJ}} = 0.04452 \text{ therm/s}$$

Noting that 1 year = $365 \times 24 = 8760$ h and unit cost of natural gas is 1.10/therm, the annual fuel and money savings will be

Fuel Saved = $(0.04452 \text{ therms/s})(8760 \times 3600 \text{ s}) = 1.404 \times 10^{6} \text{ therms/yr}$

Money saved = (Fuel saved)(Unit cost of fuel)

 $= (1.404 \times 10^{6} \text{ therm/yr})(\$1.10/\text{therm})$

$$=$$
 \$1.544 \times 10⁶ /yr

5-165E A refrigeration system is to cool eggs by chilled air at a rate of 10,000 eggs per hour. The rate of heat removal from the eggs, the required volume flow rate of air, and the size of the compressor of the refrigeration system are to be determined.

Assumptions 1 Steady operating conditions exist. 2 The eggs are at uniform temperatures before and after cooling. 3 The cooling section is well-insulated. 4 The properties of eggs are constant. 5 The local atmospheric pressure is 1 atm.

Properties The properties of the eggs are given to $\rho = 67.4 \text{ lbm/ft}^3$ and $c_p = 0.80 \text{ Btu/lbm.}^\circ\text{F}$. The specific heat of air at room temperature $c_p = 0.24 \text{ Btu/lbm.}^\circ\text{F}$ (Table A-2E). The gas constant of air is $R = 0.3704 \text{ psia.ft}^3/\text{lbm.R}$ (Table A-1E).

Analysis (a) Noting that eggs are cooled at a rate of 10,000 eggs per hour, eggs can be considered to flow steadily through the cooling section at a mass flow rate of

 $\dot{m}_{egg} = (10,000 \text{ eggs/h})(0.14 \text{ lbm/egg}) = 1400 \text{ lbm/h} = 0.3889 \text{ lbm/s}$

Taking the egg flow stream in the cooler as the system, the energy balance for steadily flowing eggs can be expressed in the rate form as



Air 34°F

Then the rate of heat removal from the eggs as they are cooled from 90°F to 50°F at this rate becomes

 $\dot{Q}_{egg} = (\dot{m}c_p \Delta T)_{egg} = (1400 \text{ lbm/h})(0.80 \text{ Btu/lbm.}^{\circ}\text{F})(90 - 50)^{\circ}\text{F} = 44,800 \text{ Btu/h}$

(b) All the heat released by the eggs is absorbed by the refrigerated air since heat transfer through he walls of cooler is negligible, and the temperature rise of air is not to exceed 10° F. The minimum mass flow and volume flow rates of air are determined to be

$$\dot{m}_{air} = \frac{\dot{Q}_{air}}{(c_p \Delta T)_{air}} = \frac{44,800 \text{ Btu/h}}{(0.24 \text{ Btu/lbm.°F})(10°\text{F})} = 18,667 \text{ lbm/h}$$

$$\rho_{air} = \frac{P}{RT} = \frac{14.7 \text{ psia}}{(0.3704 \text{ psia.ft}^3/\text{lbm.R})(34 + 460)\text{R}} = 0.0803 \text{ lbm/ft}^3$$

$$\dot{V}_{air} = \frac{\dot{m}_{air}}{\rho_{air}} = \frac{18,667 \text{ lbm/h}}{0.0803 \text{ lbm/ft}^3} = 232,500 \text{ ft}^3/\text{h}$$

5-166 Glass bottles are washed in hot water in an uncovered rectangular glass washing bath. The rates of heat and water mass that need to be supplied to the water are to be determined.

Assumptions **1** Steady operating conditions exist. **2** The entire water body is maintained at a uniform temperature of 55°C. **3** Heat losses from the outer surfaces of the bath are negligible. **4** Water is an incompressible substance with constant properties.

Properties The specific heat of water at room temperature is $c_p = 4.18 \text{ kJ/kg.}^{\circ}\text{C}$. Also, the specific heat of glass is 0.80 kJ/kg.°C (Table A-3).

Analysis (a) The mass flow rate of glass bottles through the water bath in steady operation is

 $\dot{m}_{\text{bottle}} = m_{\text{bottle}} \times \text{Bottle flow rate} = (0.150 \text{ kg} / \text{bottle})(800 \text{ bottles} / \text{min}) = 120 \text{ kg} / \text{min} = 2 \text{ kg} / \text{s}$

Taking the bottle flow section as the system, which is a steadyflow control volume, the energy balance for this steady-flow system can be expressed in the rate form as



Then the rate of heat removal by the bottles as they are heated from 20 to 55°C is

$$\dot{Q}_{\text{bottle}} = \dot{m}_{\text{bottle}} c_p \Delta T = (2 \text{ kg/s})(0.8 \text{ kJ/kg.}^{\circ} \text{ C})(55 - 20)^{\circ} \text{ C} = 56,000 \text{ W}$$

The amount of water removed by the bottles is

 $\dot{m}_{\text{water out}} = (\text{Flow rate of bottles})(\text{Water removed per bottle})$

$$= (800 \text{ bottles / min})(0.2 \text{ g/bottle}) = 160 \text{ g/min} = 2.67 \times 10^{-3} \text{ kg/s}$$

Noting that the water removed by the bottles is made up by fresh water entering

at 15°C, the rate of heat removal by the water that sticks to the bottles is

$$\dot{Q}_{\text{water removed}} = \dot{m}_{\text{water removed}} c_p \Delta T = (2.67 \times 10^{-3} \text{ kg/s})(4180 \text{ J/kg} \cdot ^{\circ}\text{C})(55 - 15)^{\circ}\text{ C} = 446 \text{ W}$$

Therefore, the total amount of heat removed by the wet bottles is

$$\dot{Q}_{\text{total, removed}} = \dot{Q}_{\text{glass removed}} + \dot{Q}_{\text{water removed}} = 56,000 + 446 = 56,446 \text{ W}$$

Discussion In practice, the rates of heat and water removal will be much larger since the heat losses from the tank and the moisture loss from the open surface are not considered.



Water bath

Assumptions 1 Steady operating conditions exist. 2 The thermal properties of the wire are constant.

Properties The properties of aluminum are given to be $\rho = 2702 \text{ kg/m}^3$ and $c_p = 0.896 \text{ kJ/kg.}^\circ\text{C}$.

Analysis The mass flow rate of the extruded wire through the air is

 $\dot{m} = \rho \dot{\boldsymbol{V}} = \rho (\pi r_0^2) \boldsymbol{V} = (2702 \text{ kg/m}^3) \pi (0.0025 \text{ m})^2 (8 \text{ m/min}) = 0.4244 \text{ kg/min} = 0.007074 \text{ kg/s}$

Taking the volume occupied by the extruded wire as the system, which is a steady-flow control volume, the energy balance for this steady-flow system can be expressed in the rate form as



Then the rate of heat transfer from the wire to the air becomes

$$\dot{Q} = \dot{m}c_p[T(t) - T_\infty] = (0.007074 \text{ kg/s})(0.896 \text{ kJ/kg.}^\circ\text{C})(350 - 50)^\circ\text{C} = 1.90 \text{ kW}$$

5-168 Long copper wires are extruded at a velocity of 8 m/min, and are exposed to atmospheric air. The rate of heat transfer from the wire is to be determined.

Assumptions 1 Steady operating conditions exist. 2 The thermal properties of the wire are constant.

Properties The properties of copper are given to be $\rho = 8950 \text{ kg/m}^3$ and $c_p = 0.383 \text{ kJ/kg.°C}$.

Analysis The mass flow rate of the extruded wire through the air is

 $\dot{m} = \rho \dot{\boldsymbol{V}} = \rho (\pi r_0^2) \boldsymbol{V} = (8950 \text{ kg/m}^3) \pi (0.0025 \text{ m})^2 (8/60 \text{ m/s}) = 0.02343 \text{ kg/s}$

Taking the volume occupied by the extruded wire as the system, which is a steady-flow control volume, the energy balance for this steady-flow system can be expressed in the rate form as



Then the rate of heat transfer from the wire to the air becomes

$$Q = \dot{m}c_{p}[T(t) - T_{\infty}] = (0.02343 \text{ kg/s})(0.383 \text{ kJ/kg.}^{\circ}\text{C})(350 - 50)^{\circ}\text{C} = 2.69 \text{ kW}$$

5-169 Steam at a saturation temperature of $T_{sat} = 40^{\circ}$ C condenses on the outside of a thin horizontal tube. Heat is transferred to the cooling water that enters the tube at 25°C and exits at 35°C. The rate of condensation of steam is to be determined.

Assumptions 1 Steady operating conditions exist. 2 Water is an incompressible substance with constant properties at room temperature. 3 The changes in kinetic and potential energies are negligible.

Properties The properties of water at room temperature are $\rho = 997 \text{ kg/m}^3$ and $c_p = 4.18 \text{ kJ/kg.}$ °C (Table A-3). The enthalpy of vaporization of water at 40°C is $h_{\rm fg} = 2406.0$ kJ/kg (Table A-4).

Analysis The mass flow rate of water through the tube is

$$\dot{m}_{\text{water}} = \rho V A_c = (997 \text{ kg/m}^3)(2 \text{ m/s})[\pi (0.03 \text{ m})^2 / 4] = 1.409 \text{ kg/s}$$

Taking the volume occupied by the cold water in the tube as the system, which is a steady-flow control volume, the energy balance for this steady-flow system can be expressed in the rate form as



Then the rate of heat transfer to the water and the rate of condensation become

$$\dot{Q} = \dot{m}c_p (T_{\text{out}} - T_{\text{in}}) = (1.409 \text{ kg/s})(4.18 \text{ kJ/kg} \cdot ^\circ\text{C})(35 - 25)^\circ\text{C} = 58.9 \text{ kW}$$

 $\dot{Q} = \dot{m}_{\text{cond}} h_{fg} \rightarrow \dot{m}_{\text{cond}} = \frac{\dot{Q}}{h_{fg}} = \frac{58.9 \text{ kJ/s}}{2406.0 \text{ kJ/kg}} = 0.0245 \text{ kg/s}$

2406.0 kJ/kg

5-170E Steam is mixed with water steadily in an adiabatic device. The temperature of the water leaving this device is to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Kinetic and potential energy changes are negligible. **3** There are no work and heat interactions. **4** There is no heat transfer between the mixing device and the surroundings.

Analysis We take the mixing device as the system, which is a control volume. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{70 \text{ (steady)}} = 0$$

$$\underline{\dot{E}_{in}}_{in} = \dot{E}_{out}$$

$$\dot{m}_1 h_1 + \dot{m}_2 h_2 = \dot{m}_3 h_3$$

From a mass balance

$$\dot{m}_3 = \dot{m}_1 + \dot{m}_2 = 0.05 + 1 = 1.05 \text{ lbm/s}$$

The enthalpies of steam and water are (Table A-6E and A-4E)

$$\begin{array}{l} 1 = 60 \text{ psia} \\ T_1 = 350^{\circ}\text{F} \end{array} \right\} h_1 = 1208.3 \text{ Btu/lbm} \\ P_2 = 60 \text{ psia} \\ T_2 = 40^{\circ}\text{F} \end{array} \right\} h_2 \cong h_{f@.40^{\circ}\text{F}} = 8.032 \text{ Btu/lbm}$$



Substituting into the energy balance equation solving for the exit enthalpy gives

$$h_3 = \frac{\dot{m}_1 h_1 + \dot{m}_2 h_2}{\dot{m}_3} = \frac{(0.05 \text{ lbm/s})(1208.3 \text{ Btu/lbm}) + (1 \text{ lbm/s})(8.032 \text{ Btu/lbm})}{1.05 \text{ lbm/s}} = 65.19 \text{ Btu/lbm}$$

At the exit state P_3 =60 psia and h_3 =65.19 kJ/kg. An investigation of Table A-5E reveals that this is compressed liquid state. Approximating this state as saturated liquid at the specified temperature, the temperature may be determined from Table A-E to be 4

$$P_3 = 60 \text{ psia}$$

 $h_3 = 65.19 \text{ Btu/lbm}$ | $T_3 \cong T_{f@h=65.19 \text{ Btu/lbm}} = 97.2^{\circ}\text{F}$

Discussion The exact answer is determined at the compressed liquid state using EES to be **97.0°F**, indicating that the saturated liquid approximation is a reasonable one.

5-171 A constant-pressure R-134a vapor separation unit separates the liquid and vapor portions of a saturated mixture into two separate outlet streams. The flow power needed to operate this unit and the mass flow rate of the two outlet streams are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 There are no work and heat interactions.

Analysis The specific volume at the inlet is (Table A-12)

$$P_1 = 320 \text{ kPa} \\ x_1 = 0.55$$
 $v_1 = v_f + x_1 (v_g - v_f) = 0.0007772 + (0.55)(0.06360 - 0.0007772) = 0.03533 \text{ m}^3/\text{kg}$

The mass flow rate at the inlet is then

$$\dot{m}_1 = \frac{\dot{\nu}_1}{\nu_1} = \frac{0.006 \text{ m}^3/\text{s}}{0.03533 \text{ m}^3/\text{kg}} = 0.1698 \text{ kg/s}$$

For each kg of mixture processed, 0.55 kg of vapor are processed. Therefore,

$$\dot{m}_2 = 0.7\dot{m}_1 = 0.55 \times 0.1698 =$$
0.09340 kg/s
 $\dot{m}_3 = \dot{m}_1 - \dot{m}_2 = 0.45\dot{m}_1 = 0.45 \times 0.1698 =$ **0.07642 kg/s**

The flow power for this unit is

$$\dot{W}_{\text{flow}} = \dot{m}_2 P_2 \boldsymbol{v}_2 + \dot{m}_3 P_3 \boldsymbol{v}_3 - \dot{m}_1 P_1 \boldsymbol{v}_1$$

= (0.09340 kg/s)(320 kPa)(0.06360 m³/kg) + (0.07642 kg/s)(320 kPa)(0.0007772 m³/kg)
- (0.1698 kg/s)(320 kPa)(0.03533 m³/kg)
= **0 kW**



5-172E A small positioning control rocket in a satellite is driven by a container filled with R-134a at saturated liquid state. The number of bursts this rocket experience before the quality in the container is 90% or more is to be determined.

Analysis The initial and final specific volumes are

$$\begin{array}{l} T_1 = -10^{\circ} \mathrm{F} \\ x_1 = 0 \end{array} \right\} \boldsymbol{v}_1 = 0.01171 \, \mathrm{ft}^3 / \mathrm{lbm} \quad (\mathrm{Table \ A-11E}) \\ T_2 = -10^{\circ} \mathrm{F} \\ x_2 = 0.90 \end{array} \right\} \boldsymbol{v}_2 = \boldsymbol{v}_f + x_2 \boldsymbol{v}_{fg} = 0.01171 + (0.90)(2.7091 - 0.01171) = 2.4394 \, \mathrm{ft}^3 / \mathrm{lbm} \end{array}$$

The initial and final masses in the container are

$$m_1 = \frac{\mathbf{v}}{\mathbf{v}_1} = \frac{2 \text{ ft}^3}{0.01171 \text{ m}^3/\text{kg}} = 170.8 \text{ lbm}$$
$$m_2 = \frac{\mathbf{v}}{\mathbf{v}_2} = \frac{2 \text{ ft}^3}{2.4394 \text{ ft}^3/\text{lbm}} = 0.8199 \text{ lbm}$$

Then,

$$\Delta m = m_1 - m_2 = 170.8 - 0.8199 = 170.0 \text{ lbm}$$

The amount of mass released during each control burst is

$$\Delta m_b = \dot{m} \Delta t = (0.05 \text{ lbm/s})(5 \text{ s}) = 0.25 \text{ lbm}$$

The number of bursts that can be executed is then

$$N_b = \frac{\Delta m}{\Delta m_b} = \frac{170.0 \text{ lbm}}{0.25 \text{ lbm/burst}} = 680 \text{ bursts}$$

Assumptions Uniform flow exists at the inlet and outlet.

Properties The specific volume of air during inflation and deflation are given to be 15 and 13 ft³/lbm, respectively.

Analysis The volume of the airbag at any time is given by

$$\boldsymbol{V}(t) = \int_{\text{in flow time}} (\dot{m}\boldsymbol{v})_{\text{in}} dt - \int_{\text{out flow time}} (\dot{m}\boldsymbol{v})_{\text{out}} dt$$

Applying at different time periods as given in problem statement give

$$\begin{aligned} \boldsymbol{V}(t) &= \int_{0}^{t} (15 \text{ ft}^{3}/\text{lbm}) \frac{20 \text{ lbm/s}}{10 \text{ ms}} \left(\frac{1 \text{ s}}{1000 \text{ ms}}\right) t dt \quad 0 \le t \le 10 \text{ ms} \\ \boldsymbol{V}(t) &= \int_{0}^{t} 0.015t^{2} \text{ ft}^{3}/\text{ms}^{2} \quad 0 \le t \le 10 \text{ ms} \\ \boldsymbol{V}(t) &= \boldsymbol{V}(10 \text{ ms}) + \int_{10 \text{ ms}}^{t} (15 \text{ ft}^{3}/\text{lbm})(20 \text{ lbm/s}) \left(\frac{1 \text{ s}}{1000 \text{ ms}}\right) t dt \quad 10 < t \le 12 \text{ ms} \\ \boldsymbol{V}(t) &= \boldsymbol{V}(10 \text{ ms}) + 0.03 \text{ ft}^{3}/\text{ms}^{2} (t - 10 \text{ ms}) \quad 10 < t \le 12 \text{ ms} \\ \boldsymbol{V}(t) &= \boldsymbol{V}(12 \text{ ms}) + 0.03 \text{ ft}^{3}/\text{ms}^{2} (t - 12 \text{ ms}) \\ &\quad - \int_{12 \text{ ms}}^{t} (13 \text{ ft}^{3}/\text{lbm}) \frac{16 \text{ lbm/s}}{(30 - 12) \text{ ms}} \left(\frac{1 \text{ s}}{1000 \text{ ms}}\right) (t - 12 \text{ ms}) dt \quad 12 < t \le 25 \text{ ms} \\ \boldsymbol{V}(t) &= \boldsymbol{V}(12 \text{ ms}) + 0.03 \text{ ft}^{3}/\text{ms}^{2} (t - 12 \text{ ms}) \\ &\quad - \int_{12 \text{ ms}}^{t} 0.011556 \text{ ft}^{3}/\text{ms}^{2} (t - 12 \text{ ms}) \\ &\quad - \int_{12 \text{ ms}}^{t} 0.011556 \text{ ft}^{3}/\text{ms}^{2} (t - 12 \text{ ms}) dt \quad 12 < t \le 25 \text{ ms} \\ \boldsymbol{V}(t) &= \boldsymbol{V}(12 \text{ ms}) + 0.03 \text{ ft}^{3}/\text{ms}^{2} (t - 12 \text{ ms}) \\ &\quad - \frac{0.011556}{12 \text{ ms}} (t^{2} - 144 \text{ ms}^{2}) + 0.13867(t - 12 \text{ ms}) \quad 12 < t \le 25 \text{ ms} \\ \boldsymbol{V}(t) &= \boldsymbol{V}(25 \text{ ms}) - \frac{0.011556}{2} (t^{2} - 625 \text{ ms}^{2}) + 0.13867(t - 25 \text{ ms}) \quad 25 < t \le 30 \text{ ms} \end{aligned}$$

$$V(t) = V(30 \text{ ms}) - \int_{12 \text{ ms}}^{t} (13 \text{ ft}^3/\text{lbm})(16 \text{ lbm/s}) \left(\frac{1 \text{ s}}{1000 \text{ ms}}\right) dt \quad 30 < t \le 50 \text{ ms}$$

$$V(t) = V(30 \text{ ms}) - (0.208 \text{ ft}^3/\text{ms})(t - 30 \text{ ms})$$
 $30 < t \le 50 \text{ ms}$

The results with some suitable time intervals are



Alternative solution

The net volume flow rate is obtained from

$$\dot{\boldsymbol{V}} = (\dot{m}\boldsymbol{v})_{\rm in} - (\dot{m}\boldsymbol{v})_{\rm out}$$

which is sketched on the figure below. The volume of the airbag is given by

$$V = \int \dot{V} dt$$

The results of a graphical interpretation of the volume is also given in the figure below. Note that the evaluation of the above integral is simply the area under the process curve.



Assumptions **1** The house is maintained at 72°F at all times. **2** The latent heat load during the heating season is negligible. **3** The infiltrating air is heated to 72°F before it exfiltrates. **4** Air is an ideal gas with constant specific heats at room temperature. **5** The changes in kinetic and potential energies are negligible. **6** Steady flow conditions exist.

Properties The gas constant of air is $0.3704 \text{ psia.ft}^3/\text{lbm}\cdot\text{R}$ (Table A-1E). The specific heat of air at room temperature is $0.24 \text{ Btu/lbm}\cdot^\circ\text{F}$ (Table A-2E).

Analysis The density of air at the outdoor conditions is

$$\rho_o = \frac{P_o}{RT_o} = \frac{13.5 \text{ psia}}{(0.3704 \text{ psia.ft}^3/\text{lbm.R})(496.5 \text{ R})} = 0.0734 \text{ lbm/ft}^3$$

The volume of the house is

 $U_{\text{building}} = (\text{Floor area})(\text{Height}) = (4500 \text{ ft}^2)(9 \text{ ft}) = 40,500 \text{ ft}^3$



We can view infiltration as a steady stream of air that is heated as it flows in an imaginary duct passing through the house. The energy balance for this imaginary steady-flow system can be expressed in the rate form as

$$\dot{\underline{E}}_{in} - \dot{\underline{E}}_{out} = \Delta \dot{\underline{E}}_{system} = 0$$
Rate of net energy transfer
by heat, work, and mass
$$\dot{\overline{E}}_{in} = \dot{E}_{out}$$

$$\dot{\underline{Q}}_{in} + \dot{m}h_1 = \dot{m}h_2 \quad (\text{since } \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

$$\dot{\underline{Q}}_{in} = \dot{m}c_p(T_2 - T_1) = \rho \dot{\underline{V}}c_p(T_2 - T_1)$$

The reduction in the infiltration rate is 2.2 - 1.1 = 1.1 ACH. The reduction in the sensible infiltration heat load corresponding to it is

$$\dot{Q}_{\text{infiltration, saved}} = \rho_o c_p (ACH_{\text{saved}}) (\mathcal{V}_{\text{building}}) (T_i - T_o)$$

= (0.0734 lbm/ft³)(0.24 Btu/lbm.°F)(1.1/h)(40,500 ft³)(72 - 36.5)°F
= 27,860 Btu/h = 0.2786 therm/h

since 1 therm = 100,000 Btu. The number of hours during a six month period is $6 \times 30 \times 24 = 4320$ h. Noting that the furnace efficiency is 0.92 and the unit cost of natural gas is \$1.24/therm, the energy and money saved during the 6-month period are

Energy savings =
$$(\dot{Q}_{infiltration, saved})$$
(No. of hours per year)/Efficiency
= $(0.2786 \text{ therm/h})(4320 \text{ h/year})/0.92$
= 1308 therms/year
Cost savings = (Energy savings)(Unit cost of energy)

= (1308 therms/year)(\$1.24/therm)

= \$1622/year

Therefore, reducing the infiltration rate by one-half will reduce the heating costs of this homeowner by \$1622 per year.

5-175 Outdoors air at -5°C and 90 kPa enters the building at a rate of 35 L/s while the indoors is maintained at 20°C. The rate of sensible heat loss from the building due to infiltration is to be determined.

Assumptions 1 The house is maintained at 20°C at all times. 2 The latent heat load is negligible. 3 The infiltrating air is heated to 20°C before it exfiltrates. 4 Air is an ideal gas with constant specific heats at room temperature. 5 The changes in kinetic and potential energies are negligible. 6 Steady flow conditions exist.

Properties The gas constant of air is R = 0.287 kPa.m³/kg·K. The specific heat of air at room temperature is $c_p = 1.005$ kJ/kg·°C (Table A-2).

Analysis The density of air at the outdoor conditions is

$$\rho_o = \frac{P_o}{RT_o} = \frac{90 \text{ kPa}}{(0.287 \text{ kPa.m}^3/\text{kg.K})(-5 + 273 \text{ K})} = 1.17 \text{ kg/m}^3$$

We can view infiltration as a steady stream of air that is heated as it flows in an imaginary duct passing through the building. The energy balance for this imaginary steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{Bate of net energy transfer} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{Q}_{in} + \dot{m}h_1 = \dot{m}h_2} \quad (since \Delta ke \cong \Delta pe \cong 0)$$

$$\underbrace{\dot{Q}_{in} = \dot{m}c_p(T_2 - T_1)}_{\dot{Q}_{in} = \dot{m}c_p(T_2 - T_1)}$$



Then the sensible infiltration heat load corresponding to an infiltration rate of 35 L/s becomes

$$\dot{Q}_{\text{infiltration}} = \rho_o \dot{V}_{air} c_p (T_i - T_o)$$

= (1.17 kg/m³)(0.035 m³/s)(1.005 kJ/kg.°C)[20-(-5)]°C
= **1.029 kW**

Therefore, sensible heat will be lost at a rate of 1.029 kJ/s due to infiltration.

5-176 Chilled air is to cool a room by removing the heat generated in a large insulated classroom by lights and students. The required flow rate of air that needs to be supplied to the room is to be determined.

Assumptions 1 The moisture produced by the bodies leave the room as vapor without any condensing, and thus the classroom has no latent heat load. 2 Heat gain through the walls and the roof is negligible. 4 Air is an ideal gas with constant specific heats at room temperature. 5 Steady operating conditions exist.

Properties The specific heat of air at room temperature is 1.005 kJ/kg.°C (Table A-2). The average rate of sensible heat generation by a person is given to be 60 W.

Analysis The rate of sensible heat generation by the people in the room and the total rate of sensible internal heat generation are

$$\dot{Q}_{\text{gen, sensible}} = \dot{q}_{\text{gen, sensible}} \text{ (No. of people)} = (60 \text{ W/person})(150 \text{ persons}) = 9000 \text{ W}$$

 $\dot{Q}_{\text{total, sensible}} = \dot{Q}_{\text{gen, sensible}} + \dot{Q}_{\text{lighting}} = 9000 + 6000 = 15,000 \text{ W}$

Both of these effects can be viewed as heat gain for the chilled air stream, which can be viewed as a steady stream of cool air that is heated as it flows in an imaginary duct passing through the room. The energy balance for this imaginary steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{Bate of net energy transfer} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{Q}_{in} + \dot{m}h_{1} = \dot{m}h_{2}} \quad (since \Delta ke \cong \Delta pe \cong 0)$$

$$\underbrace{\dot{Q}_{in} = \dot{Q}_{total, sensible} = \dot{m}c_{p}(T_{2} - T_{1})$$

Then the required mass flow rate of chilled air becomes

$$\dot{m}_{air} = \frac{\dot{Q}_{total, sensible}}{c_n \Delta T} = \frac{15 \text{ kJ/s}}{(1.005 \text{ kJ/kg} \cdot ^\circ\text{C})(25 - 15)^\circ\text{C}} = 1.49 \text{ kg/s}$$



Discussion The latent heat will be removed by the air-conditioning system as the moisture condenses outside the cooling coils.

5-177 A fan is powered by a 0.5 hp motor, and delivers air at a rate of 85 m^3 /min. The highest possible air velocity at the fan exit is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time at any point and thus $\Delta m_{\rm CV} = 0$ and $\Delta E_{\rm CV} = 0$. 2 The inlet velocity and the change in potential energy are negligible, $V_1 \cong 0$ and $\Delta pe \cong 0$. 3 There are no heat and work interactions other than the electrical power consumed by the fan motor. 4 The efficiencies of the motor and the fan are 100% since best possible operation is assumed. 5 Air is an ideal gas with constant specific heats at room temperature.

Properties The density of air is given to be $\rho = 1.18 \text{ kg/m}^3$. The constant pressure specific heat of air at room temperature is $c_p = 1.005 \text{ kJ/kg.}^\circ\text{C}$ (Table A-2).

Analysis We take the *fan-motor assembly* as the system. This is a *control volume* since mass crosses the system boundary during the process. We note that there is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$.

The velocity of air leaving the fan will be highest when all of the entire electrical energy drawn by the motor is converted to kinetic energy, and the friction between the air layers is zero. In this best possible case, no energy will be converted to thermal energy, and thus the temperature change of air will be zero, $T_2 = T_1$. Then the energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{system}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{70 \text{ (steady)}} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{W}_{e,in} + \dot{m}h_1 = \dot{m}(h_2 + V_2^2/2)} \text{ (since } V_1 \cong 0 \text{ and } \Delta pe \cong 0)$$

Noting that the temperature and thus enthalpy remains constant, the relation above simplifies further to

$$\dot{W}_{e,in} = \dot{m}V_2^2/2$$

where

$$\dot{m} = \rho \dot{\boldsymbol{V}} = (1.18 \text{ kg/m}^3)(85 \text{ m}^3/\text{min}) = 100.3 \text{ kg/min} = 1.67 \text{ kg/s}$$

Solving for V_2 and substituting gives

$$V_2 = \sqrt{\frac{2\dot{W}_{e,in}}{\dot{m}}} = \sqrt{\frac{2(0.5 \text{ hp})}{1.67 \text{ kg/s}}} \left(\frac{745.7 \text{ W}}{1 \text{ hp}}\right) \left(\frac{1 \text{ m}^2/\text{s}^2}{1 \text{ W}}\right) = 21.1 \text{ m/s}$$

Discussion In reality, the velocity will be less because of the inefficiencies of the motor and the fan.


5-178 The average air velocity in the circular duct of an air-conditioning system is not to exceed 8 m/s. If the fan converts 80 percent of the electrical energy into kinetic energy, the size of the fan motor needed and the diameter of the main duct are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time at any point and thus $\Delta m_{\rm CV} = 0$ and $\Delta E_{\rm CV} = 0$. 2 The inlet velocity is negligible, $V_1 \cong 0$. 3 There are no heat and work interactions other than the electrical power consumed by the fan motor. 4 Air is an ideal gas with constant specific heats at room temperature.

Properties The density of air is given to be $\rho = 1.20 \text{ kg/m}^3$. The constant pressure specific heat of air at room temperature is $c_p = 1.005 \text{ kJ/kg.}^\circ\text{C}$ (Table A-2).

Analysis We take the *fan-motor assembly* as the system. This is a *control volume* since mass crosses the system boundary during the process. We note that there is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. The change in the kinetic energy of air as it is accelerated from zero to 8 m/s at a rate of 130 m³/min is

$$\dot{m} = \rho \dot{\boldsymbol{V}} = (1.20 \text{ kg/m}^3)(130 \text{ m}^3/\text{min}) = 156 \text{ kg/min} = 2.6 \text{ kg/s}$$
$$\Delta K \dot{E} = \dot{m} \frac{V_2^2 - V_1^2}{2} = (2.6 \text{ kg/s}) \frac{(8 \text{ m/s})^2 - 0}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2}\right) = 0.0832 \text{ kW}$$

It is stated that this represents 80% of the electrical energy consumed by the motor. Then the total electrical power consumed by the motor is determined to be

$$0.7\dot{W}_{motor} = \Delta K \dot{E} \rightarrow \dot{W}_{motor} = \frac{\Delta K \dot{E}}{0.8} = \frac{0.0832 \text{ kW}}{0.8} = 0.104 \text{ kW}$$

The diameter of the main duct is

$$\dot{\boldsymbol{V}} = VA = V(\pi D^2 / 4) \rightarrow D = \sqrt{\frac{4\dot{\boldsymbol{V}}}{\pi V}} = \sqrt{\frac{4(130 \text{ m}^3 / \text{min})}{\pi (8 \text{ m/s})}} \left(\frac{1 \text{ min}}{60 \text{ s}}\right) = 0.587 \text{ m}$$

Therefore, the motor should have a rated power of at least 0.104 kW, and the diameter of the duct should be at least 58.7 cm



5-179 An adiabatic air compressor is powered by a direct-coupled steam turbine, which is also driving a generator. The net power delivered to the generator is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 The devices are adiabatic and thus heat transfer is negligible. 4 Air is an ideal gas with variable specific heats.

Properties From the steam tables (Tables A-4 through 6)

$$\begin{array}{c} P_3 = 12.5 \text{ MPa} \\ T_3 = 500^{\circ}\text{C} \end{array} \right\} h_3 = 3343.6 \text{ kJ/kg} \end{array}$$

and

$$P_4 = 10 \text{ kPa} \\ x_4 = 0.92$$
 $h_4 = h_f + x_4 h_{fg} = 191.81 + (0.92)(2392.1) = 2392.5 \text{ kJ/kg}$

From the air table (Table A-17),

$$T_1 = 295 \text{ K} \longrightarrow h_1 = 295.17 \text{ kJ/kg}$$

 $T_2 = 620 \text{ K} \longrightarrow h_2 = 628.07 \text{ kJ/kg}$

Analysis There is only one inlet and one exit for either device, and thus $\dot{m}_{\rm in} = \dot{m}_{\rm out} = \dot{m}$. We take either the turbine or the compressor as the system, which is a control volume since mass crosses the boundary. The energy balance for either steady-flow system can be expressed in the rate form as

$$\underline{\dot{E}_{in} - \dot{E}_{out}}_{by heat, work, and mass} = \underbrace{\Delta \dot{E}_{system}}_{Rate of heat energy transfer} = 0$$
Rate of change in internal, kinetic, potential, etc. energies
$$\dot{E}_{in} = \dot{E}_{out}$$

For the turbine and the compressor it becomes

Compressor:

$$\dot{W}_{\text{comp, in}} + \dot{m}_{\text{air}}h_1 = \dot{m}_{\text{air}}h_2 \rightarrow \dot{W}_{\text{comp, in}} = \dot{m}_{\text{air}}(h_2 - h_1)$$

Turbine:

$$\dot{m}_{\text{steam}}h_3 = \dot{W}_{\text{turb, out}} + \dot{m}_{\text{steam}}h_4 \rightarrow \dot{W}_{\text{turb, out}} = \dot{m}_{\text{steam}}(h_3 - h_4)$$

Substituting,

$$\dot{W}_{\text{comp,in}} = (10 \text{ kg/s})(628.07 - 295.17) \text{ kJ/kg} = 3329 \text{ kW}$$

 $\dot{W}_{\text{nurb out}} = (25 \text{ kg/s})(3343.6 - 2392.5) \text{ kJ/kg} = 23,777 \text{ kW}$

Therefore,

$$\dot{W}_{\text{net out}} = \dot{W}_{\text{turb out}} - \dot{W}_{\text{comp in}} = 23,777 - 3329 = 20,448 \text{ kW}$$



5-180 Helium is compressed by a compressor. The power required is to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Helium is an ideal gas with constant specific heats. 4 The compressor is adiabatic.

Properties The constant pressure specific heat of helium is $c_p = 5.1926 \text{ kJ/kg} \cdot \text{K}$. The gas constant is $R = 2.0769 \text{ kJ/kg} \cdot \text{K}$ (Table A-2a).

Analysis There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. We take the compressor as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

 $\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{70 \text{ (steady)}} = 0$ $\underbrace{\text{Rate of change in internal, kinetic, potential, etc. energies}}_{\dot{E}_{in}} = \dot{E}_{out}$ $\underbrace{\dot{W}_{in} + \dot{m}h_1 = \dot{m}h_2}_{in} \text{ (since } \Delta \text{ke} \cong \Delta \text{pe} \cong 0)}_{\dot{W}_{in}} = \dot{m}(h_2 - h_1) = \dot{m}c_p(T_2 - T_1)$

The specific volume of air at the inlet and the mass flow rate are

$$\boldsymbol{\nu}_{1} = \frac{RT_{1}}{P_{1}} = \frac{(2.0769 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(20 + 273 \text{ K})}{150 \text{ kPa}} = 4.0569 \text{ m}^{3}/\text{kg}$$
$$\dot{\boldsymbol{m}} = \frac{A_{1}V_{1}}{\boldsymbol{\nu}_{1}} = \frac{(0.1 \text{ m}^{2})(15 \text{ m/s})}{4.0569 \text{ m}^{3}/\text{kg}} = 0.3697 \text{ kg/s}$$

Then the power input is determined from the energy balance equation to be

$$\dot{W}_{in} = \dot{m}c_n(T_2 - T_1) = (0.3697 \text{ kg/s})(5.1926 \text{ kJ/kg} \cdot \text{K})(200 - 20)\text{K} = 345.5 \text{ kW}$$



Assumptions 1 This is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible.

Properties From the R-134a tables (Table A-11)

$$T_1 = 10^{\circ}C$$

 $x_1 = 1$ $h_1 = 256.16 \text{ kJ/kg}$

Analysis We take the compressor as the system, which is a control volume since mass crosses the boundary. Noting that one fluid stream enters and leaves the compressor, the energy balance for this steady-flow system can be expressed in the rate form as



Substituting,

$$\dot{Q}_{\text{out}} = \dot{W}_{\text{in}} + \dot{m} \left(h_1 - h_2 - \frac{V_2^2}{2} \right)$$

= 132.4 kW + (5 kg/s) $\left[256.116 - 281.39 \right] \text{kJ/kg} - \frac{(50 \text{ m/s})^2}{2} \left(\frac{1 \text{ kJ/kg}}{1000 \text{ m}^2/\text{s}^2} \right) \right]$
= **0.02 kW**

which is practically zero and therefore the process is adiabatic.

5-182 A submarine that has an air-ballast tank originally partially filled with air is considered. Air is pumped into the ballast tank until it is entirely filled with air. The final temperature and mass of the air in the ballast tank are to be determined.

Assumptions 1 Air is an ideal gas with constant specific heats. 2 The process is adiabatic. 3 There are no work interactions.

Properties The gas constant of air is R = 0.287 kPa·m³/kg·K (Table A-1). The specific heat ratio of air at room temperature is k = 1.4 (Table A-2*a*). The specific volume of water is taken 0.001 m³/kg.

Analysis The conservation of mass principle applied to the air gives

$$\frac{dm_a}{dt} = \dot{m}_{\rm in}$$

and as applied to the water becomes

$$\frac{dm_w}{dt} = -\dot{m}_{\rm out}$$

The first law for the ballast tank produces

$$0 = \frac{d(mu)_a}{dt} + \frac{d(mu)_w}{dt} + h_w \dot{m}_w - h_a \dot{m}_a$$

Combining this with the conservation of mass expressions, rearranging and canceling the common dt term produces

$$d(mu)_a + d(mu)_w = h_a dm_a + h_w dm_w$$

Integrating this result from the beginning to the end of the process gives

$$[(mu)_2 - (mu)_1]_a + [(mu)_2 - (mu)_1]_w = h_a (m_2 - m_1)_a + h_w (m_2 - m_1)_w$$

Substituting the ideal gas equation of state and the specific heat models for the enthalpies and internal energies expands this to

$$\frac{PV_2}{RT_2}c_vT_2 - \frac{PV_1}{RT_1}c_vT_1 - m_{w,1}c_wT_w = c_pT_{in}\left(\frac{PV_2}{RT_2} - \frac{PV_1}{RT_1}\right) - m_{w,1}c_wT_w$$

When the common terms are cancelled, this result becomes

$$T_2 = \frac{\boldsymbol{V}_2}{\frac{\boldsymbol{V}_1}{T_1} + \frac{1}{kT_{\text{in}}}(\boldsymbol{V}_2 - \boldsymbol{V}_1)} = \frac{700}{\frac{100}{288} + \frac{1}{(1.4)(293)}(700 - 100)} = 386.8 \text{ K}$$

The final mass from the ideal gas relation is

$$m_2 = \frac{PV_2}{RT_2} = \frac{(1500 \text{ kPa})(700 \text{ m}^3)}{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(386.8 \text{ K})} = 9460 \text{ kg}$$

5-183 A submarine that has an air-ballast tank originally partially filled with air is considered. Air is pumped into the ballast tank in an isothermal manner until it is entirely filled with air. The final mass of the air in the ballast tank and the total heat transfer are to be determined.

Assumptions 1 Air is an ideal gas with constant specific heats. 2 There are no work interactions.

Properties The gas constant of air is R = 0.287 kPa·m³/kg·K (Table A-1). The specific heats of air at room temperature are $c_p = 1.005$ kJ/kg·K and $c_v = 0.718$ kJ/kg·K (Table A-2*a*). The specific volume of water is taken 0.001 m³/kg.

Analysis The initial air mass is

$$m_1 = \frac{P_1 V_1}{RT_1} = \frac{(1500 \text{ kPa})(100 \text{ m}^3)}{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(288.15 \text{ K})} = 1814 \text{ kg}$$

and the initial water mass is

$$m_w = \frac{\nu_1}{\nu_1} = \frac{600 \,\mathrm{m}^3}{0.001 \,\mathrm{m}^3/\mathrm{kg}} = 600,000 \,\mathrm{kg}$$

and the final mass of air in the tank is

$$m_2 = \frac{P_2 V_2}{RT_2} = \frac{(1500 \text{ kPa})(700 \text{ m}^3)}{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(288.15 \text{ K})} = 12,697 \text{ kg}$$

The first law when adapted to this system gives

$$\begin{aligned} Q_{\rm in} + m_i h_i - m_e h_e &= m_2 u_2 - m_1 u_1 \\ Q_{\rm in} &= m_2 u_2 - m_1 u_1 + m_e h_e - m_i h_i \\ Q_{\rm in} &= m_2 c_v T - (m_{a,1} c_v T + m_w u_w) + m_w h_w - (m_2 - m_1) c_p T \end{aligned}$$

Noting that

$$u_w \cong h_w = 62.98 \text{ kJ/kg}$$

Substituting,

$$\begin{aligned} \mathcal{Q}_{\text{in}} &= 12,697 \times 0.718 \times 288 - (1814 \times 0.718 \times 288 + 600,000 \times 62.98) \\ &+ 600,000 \times 62.98 - (12,697 - 1814) \times 1.005 \times 288 \\ &= \mathbf{0} \text{ kJ} \end{aligned}$$

The process is adiabatic.

5-184 A cylindrical tank is charged with nitrogen from a supply line. The final mass of nitrogen in the tank and final temperature are to be determined for two cases.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. 2 Air is an ideal gas with constant specific heats. 3 Kinetic and potential energies are negligible. 4 There are no work interactions involved. **Properties** The gas constant of nitrogen is 0.2968 kPa·m³/kg·K (Table A-1). The specific heats of nitrogen at room temperature are $c_p = 1.039$ kJ/kg·K and $c_v = 0.743$ kJ/kg·K (Table A-2a).

Analysis (*a*) We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_i = m_2 - m_1$$

Energy balance:

$$\underbrace{E_{in} - E_{out}}_{Net energy transfer} = \underbrace{\Delta E_{system}}_{Change in internal, kinetic, potential, etc. energies}$$
$$m_i h_i = m_2 u_2 - m_1 u_1$$
$$m_i c_p T_i = m_2 c_v T_2 - m_1 c_v T_1$$

Combining the two balances:

$$(m_2 - m_1)c_n T_i = m_2 c_v T_2 - m_1 c_v T_1$$

The initial and final masses are given by

$$m_1 = \frac{P_1 \mathbf{V}}{RT_1} = \frac{(200 \text{ kPa})(0.1 \text{ m}^3)}{(0.2968 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(298 \text{ K})} = 0.2261 \text{ kg}$$
$$m_2 = \frac{P_2 \mathbf{V}}{RT_2} = \frac{(800 \text{ kPa})(0.1 \text{ m}^3)}{(0.2968 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})T_2} = \frac{269.5}{T_2}$$

Substituting,

$$\left(\frac{269.5}{T_2} - 0.2261\right)(1.039)(298) = \frac{269.5}{T_2}(0.743)T_2 - (0.2261)(0.743)(298)$$

whose solution is

$$T_2 = 379.0 \text{ K}$$

The final mass is then

$$m_2 = \frac{269.5}{T_2} = \frac{269.5}{379.0} = 0.7112 \text{ kg}$$

(b) When there is rapid heat transfer between the nitrogen and tank such that the cylinder and nitrogen remain in thermal equilibrium during the process, the energy balance equation may be written as

$$(m_2 - m_1)c_pT_i = (m_{nit,2}c_vT_2 + m_tc_tT_2) - (m_{nit,1}c_vT_1 + m_tc_tT_1)$$

Substituting,

$$\left(\frac{269.5}{T_2} - 0.2261\right)(1.039)(298) = \left[\frac{269.5}{T_2}(0.743)T_2 + (50)(0.43)T_2\right] - \left[(0.2261)(0.743)(298) + (50)(0.43)(298)\right]$$

whose solution is

$$T_2 = 300.7 \text{ K}$$

The final mass is then

$$m_2 = \frac{269.5}{300.7} = \frac{269.5}{300.7} =$$
0.8962 kg

Air \rightarrow 800 kPa, 25°C 200 kPa 25°C 0.1 m³ **5-185** The air in a tank is released until the pressure in the tank reduces to a specified value. The mass withdrawn from the tank is to be determined for three methods of analysis.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process. **2** Air is an ideal gas with constant specific heats. **3** Kinetic and potential energies are negligible. **4** There are no work or heat interactions involved.

Properties The gas constant of air is 0.287 kPa·m³/kg·K (Table A-1). The specific heats of air at room temperature are $c_p = 1.005$ kJ/kg·K and $c_v = 0.718$ kJ/kg·K. Also k = 1.4 (Table A-2*a*).

Analysis (*a*) We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system}$$
$$- m_e = m_2 - m_1$$
$$m_e = m_1 - m_2$$

Energy balance:

$$\underbrace{E_{in} - E_{out}}_{Net energy transfer} = \underbrace{\Delta E_{system}}_{Change in internal, kinetic, potential, etc. energies}$$
$$-m_e h_e = m_2 u_2 - m_1 u_1$$
$$0 = m_2 u_2 - m_1 u_1 + m_e h_e$$
$$0 = m_2 c_v T_2 - m_1 c_v T_1 + m_e c_p T_e$$



Combining the two balances:

 $0 = m_2 c_v T_2 - m_1 c_v T_1 + (m_1 - m_2) c_p T_e$

The initial and final masses are given by

$$m_1 = \frac{P_1 \mathbf{V}}{RT_1} = \frac{(800 \text{ kPa})(1 \text{ m}^3)}{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(25 + 273 \text{ K})} = 9.354 \text{ kg}$$
$$m_2 = \frac{P_2 \mathbf{V}}{RT_2} = \frac{(150 \text{ kPa})(1 \text{ m}^3)}{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})T_2} = \frac{522.6}{T_2}$$

The temperature of air leaving the tank changes from the initial temperature in the tank to the final temperature during the discharging process. We assume that the temperature of the air leaving the tank is the average of initial and final temperatures in the tank. Substituting into the energy balance equation gives

$$0 = m_2 c_v T_2 - m_1 c_v T_1 + (m_1 - m_2) c_p T_e$$

$$0 = \frac{522.6}{T_2} (0.718) T_2 - (9.354)(0.718)(298) + \left(9.354 - \frac{522.6}{T_2}\right) (1.005) \left(\frac{298 + T_2}{2}\right)$$

whose solution is

$$T_2 = 191.0 \,\mathrm{K}$$

Substituting, the final mass is

$$m_2 = \frac{522.6}{191} = 2.736 \text{ kg}$$

and the mass withdrawn is

$$m_e = m_1 - m_2 = 9.354 - 2.736 =$$
6.618 kg

(b) Considering the process in two parts, first from 800 kPa to 400 kPa and from 400 kPa to 150 kPa, the solution will be as follows:

From 800 kPa to 400 kPa:

$$m_2 = \frac{P_2 \mathbf{V}}{RT_2} = \frac{(400 \text{ kPa})(1 \text{ m}^3)}{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})T_2} = \frac{1394}{T_2}$$

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$$0 = \frac{1394}{T_2} (0.718)T_2 - (9.354)(0.718)(298) + \left(9.354 - \frac{1394}{T_2}\right)(1.005)\left(\frac{298 + T_2}{2}\right)$$

$$T_2 = 245.1 \text{ K}$$

$$m_2 = \frac{1394}{245.1} = 5.687 \text{ kg}$$

$$m_{e,1} = m_1 - m_2 = 9.354 - 5.687 = 3.667 \text{ kg}$$

From 400 kPa to 150 kPa:

 $0 = \frac{522.6}{T_2} (0.718)T_2 - (5.687)(0.718)(245.1) + \left(5.687 - \frac{522.6}{T_2}\right)(1.005)\left(\frac{245.1 + T_2}{2}\right)$ $T_2 = 186.5 \text{ K}$ $m_2 = \frac{522.6}{186.5} = 2.803 \text{ kg}$ $m_{e,2} = m_1 - m_2 = 5.687 - 2.803 = 2.884 \text{ kg}$

The total mass withdrawn is

$$m_e = m_{e,1} + m_{e,2} = 3.667 + 2.884 = 6.551 \,\mathrm{kg}$$

(c) The mass balance may be written as

$$\frac{dm}{dt} = -\dot{m}_e$$

When this is combined with the ideal gas equation of state, it becomes

$$\frac{\mathbf{V}}{R}\frac{d(P/T)}{dt} = -\dot{m}_e$$

since the tank volume remains constant during the process. An energy balance on the tank gives

$$\frac{d(mu)}{dt} = -h_e \dot{m}_e$$

$$c_v \frac{d(mT)}{dt} = c_p T \frac{dm}{dt}$$

$$c_v \frac{V}{R} \frac{dP}{dt} = c_p T \frac{V}{R} \frac{d(P/T)}{dt}$$

$$c_v \frac{dP}{dt} = c_p \left(\frac{dP}{dt} - \frac{P}{T} \frac{dT}{dt}\right)$$

$$(c_p - c_v) \frac{dP}{P} = c_p \frac{dT}{dt}$$

When this result is integrated, it gives

$$T_2 = T_1 \left(\frac{P_2}{P_1}\right)^{(k-1)/k} = (298 \text{ K}) \left(\frac{150 \text{ kPa}}{800 \text{ kPa}}\right)^{0.4/1.4} = 184.7 \text{ K}$$

The final mass is

$$m_2 = \frac{P_2 V}{RT_2} = \frac{(150 \text{ kPa})(1 \text{ m}^3)}{(0.287 \text{ kPa} \cdot \text{m}^3/\text{kg} \cdot \text{K})(184.7 \text{ K})} = 2.830 \text{ kg}$$

and the mass withdrawn is

$$m_e = m_1 - m_2 = 9.354 - 2.830 =$$
6.524 kg

Discussion The result in first method is in error by 1.4% while that in the second method is in error by 0.4%.

5-186 A tank initially contains saturated mixture of R-134a. A valve is opened and R-134a vapor only is allowed to escape slowly such that temperature remains constant. The heat transfer necessary with the surroundings to maintain the temperature and pressure of the R-134a constant is to be determined.

Assumptions **1** This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the exit remains constant. **2** Kinetic and potential energies are negligible. **3** There are no work interactions involved.

Analysis We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system}$$
$$- m_e = m_2 - m_1$$
$$m_e = m_1 - m_2$$

Energy balance:

$$\underbrace{\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}}}_{\text{by heat, work, and mass}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}} \\ Q_{\text{in}} - m_e h_e = m_2 u_2 - m_1 u_1 \\ Q_{\text{in}} = m_2 u_2 - m_1 u_1 + m_e h_e$$

Combining the two balances:

$$Q_{\rm in} = m_2 u_2 - m_1 u_1 + (m_1 - m_2) h_e$$

The specific volume at the initial state is

$$v_1 = \frac{v}{m_1} = \frac{0.001 \,\mathrm{m}^3}{0.4 \,\mathrm{kg}} = 0.0025 \,\mathrm{m}^3/\mathrm{kg}$$

The initial state properties of R-134a in the tank are

$$T_{1} = 26^{\circ}\text{C}$$

$$\boldsymbol{v}_{1} = 0.0025 \text{ m}^{3}/\text{kg} \begin{cases} x_{1} = \frac{\boldsymbol{v}_{1} - \boldsymbol{v}_{f}}{\boldsymbol{v}_{fg}} = \frac{0.0025 - 0.0008313}{0.029976 - 0.0008313} = 0.05726 \\ u_{1} = u_{f} + x_{1}u_{fg} = 87.26 + (0.05726)(156.87) = 96.24 \text{ kJ/kg} \end{cases}$$
(Table A-11)

The enthalpy of saturated vapor refrigerant leaving the bottle is

$$h_e = h_{g(a),26^{\circ}C} = 264.68 \text{ kJ/kg}$$

The specific volume at the final state is

$$v_2 = \frac{v}{m_2} = \frac{0.001 \,\mathrm{m}^3}{0.1 \,\mathrm{kg}} = 0.01 \,\mathrm{m}^3/\mathrm{kg}$$

The internal energy at the final state is

$$T_{2} = 26^{\circ}\text{C}$$

$$v_{2} = 0.01 \text{ m}^{3}/\text{kg} \begin{cases} x_{2} = \frac{v_{2} - v_{f}}{v_{fg}} = \frac{0.01 - 0.0008313}{0.029976 - 0.0008313} = 0.3146 \\ u_{2} = u_{f} + x_{1}u_{fg} = 87.26 + (0.3146)(156.87) = 136.61 \text{ kJ/kg} \end{cases}$$
(Table A-11)

Substituting into the energy balance equation,

$$Q_{in} = m_2 u_2 - m_1 u_1 + (m_1 - m_2) h_e$$

= (0.1 kg)(136.61 kJ/kg) - (0.4 kg)(96.24 kJ/kg) + (0.4 - 0.1 kg)(264.68 kJ/kg)
= **54.6 kJ**

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5-187 Steam expands in a turbine steadily. The mass flow rate of the steam, the exit velocity, and the power output are to be determined.

Assumptions 1 This is a steady-flow process since there is no change with time. 2 Potential energy changes are negligible.

Properties From the steam tables (Tables A-4 through 6)

$$P_1 = 7 \text{ MPa}$$
 $v_1 = 0.05567 \text{ m}^3/\text{kg}$
 $T_1 = 600^{\circ}\text{C}$ $h_1 = 3650.6 \text{ kJ/kg}$

and

$$P_{2} = 25 \text{ kPa} \left\{ \boldsymbol{v}_{2} = \boldsymbol{v}_{f} + x_{2}\boldsymbol{v}_{fg} = 0.00102 + (0.95)(6.2034 - 0.00102) = 5.8933 \text{ m}^{3}/\text{kg} \\ x_{2} = 0.95 \right\} \quad h_{2} = h_{f} + x_{2}h_{fg} = 271.96 + (0.95)(2345.5) = 2500.2 \text{ kJ/kg}$$

Analysis (a) The mass flow rate of the steam is

$$\dot{m} = \frac{1}{\nu_1} V_1 A_1 = \frac{1}{0.05567 \text{ m}^3/\text{kg}} (60 \text{ m/s}) (0.015 \text{ m}^2) = 16.17 \text{ kg/s}$$

(b) There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. Then the exit velocity is determined from

$$\dot{m} = \frac{1}{v_2} V_2 A_2 \longrightarrow V_2 = \frac{\dot{m} v_2}{A_2} = \frac{(16.17 \text{ kg/s})(5.8933 \text{ m}^3/\text{kg})}{0.14 \text{ m}^2} = 680.6 \text{ m/s}$$

(c) We take the turbine as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0$$

$$\frac{\dot{E}_{\text{in}} - \dot{E}_{\text{out}}}{\dot{E}_{\text{in}} = \dot{E}_{\text{out}}}$$

$$\frac{\dot{m}(h_1 + V_1^2 / 2) = \dot{W}_{\text{out}} + \dot{Q}_{\text{out}} + \dot{m}(h_2 + V_2^2 / 2) \quad (\text{since } \Delta \text{pe} \cong 0)$$

$$\frac{\dot{W}_{\text{out}} = -\dot{Q}_{\text{out}} - \dot{m}\left(h_2 - h_1 + \frac{V_2^2 - V_1^2}{2}\right)$$

Then the power output of the turbine is determined by substituting to be

$$\dot{W}_{\text{out}} = -(16.17 \times 20) \,\text{kJ/s} - (16.17 \,\text{kg/s}) \left(2500.2 - 3650.6 + \frac{(680.6 \,\text{m/s})^2 - (60 \,\text{m/s})^2}{2} \left(\frac{1 \,\text{kJ/kg}}{1000 \,\text{m}^2/\text{s}^2}\right)\right) = 14,560 \,\text{kW}$$





5-188 Problem 5-187 is reconsidered. The effects of turbine exit area and turbine exit pressure on the exit velocity and power output of the turbine as the exit pressure varies from 10 kPa to 50 kPa (with the same quality), and the exit area to varies from 1000 cm^2 to 3000 cm^2 is to be investigated. The exit velocity and the power output are to be plotted against the exit pressure for the exit areas of 1000, 2000, and 3000 cm^2 .

Analysis The problem is solved using EES, and the results are tabulated and plotted below.

 $\begin{array}{l} \mbox{Fluid}='Steam_IAPWS' \\ A[1]=150 \ [cm^2] \\ T[1]=600 \ [C] \\ P[1]=7000 \ [kPa] \\ Vel[1]= 60 \ [m/s] \\ A[2]=1400 \ [cm^2] \\ P[2]=25 \ [kPa] \\ q_out = 20 \ [kJ/kg] \\ m_dot = A[1]*Vel[1]/v[1]*convert(cm^2,m^2) \\ v[1]=volume(Fluid$, T=T[1], P=P[1]) "specific volume of steam at state 1" \\ Vel[2]=m_dot*v[2]/(A[2]*convert(cm^2,m^2)) \\ v[2]=volume(Fluid$, x=0.95, P=P[2]) "specific volume of steam at state 2" \\ T[2]=temperature(Fluid$, P=P[2], v=v[2]) "[C]" "not required, but good to know" \\ \end{array}$

"[conservation of Energy for steady-flow:"
"Ein_dot - Eout_dot = DeltaE_dot" "For steady-flow, DeltaE_dot = 0"
DELTAE_dot=0
"For the turbine as the control volume, neglecting the PE of each flow steam:"
E_dot_in=E_dot_out
h[1]=enthalpy(Fluid\$,T=T[1], P=P[1])
E_dot_in=m_dot*(h[1]+ Vel[1]^2/2*Convert(m^2/s^2, kJ/kg))
h[2]=enthalpy(Fluid\$,x=0.95, P=P[2])
E_dot_out=m_dot*(h[2]+ Vel[2]^2/2*Convert(m^2/s^2, kJ/kg))+ m_dot *q_out+ W_dot_out
Power=W_dot_out
Q_dot_out=m_dot*q_out

P ₂	Power	Vel ₂
[kPa]	[kW]	[m/s]
10	-22158	2253
14.44	-1895	1595
18.89	6071	1239
23.33	9998	1017
27.78	12212	863.2
32.22	13573	751.1
36.67	14464	665.4
41.11	15075	597.8
45.56	15507	543
50	15821	497.7

Table values are for A[2]=1000 [cm^2]



5-189 Air is preheated by the exhaust gases of a gas turbine in a regenerator. For a specified heat transfer rate, the exit temperature of air and the mass flow rate of exhaust gases are to be determined.

Assumptions **1** This is a steady-flow process since there is no change with time. **2** Kinetic and potential energy changes are negligible. **3** There are no work interactions. **4** Heat loss from the regenerator to the surroundings is negligible and thus heat transfer from the hot fluid is equal to the heat transfer to the cold fluid. **5** Exhaust gases can be treated as air. **6** Air is an ideal gas with variable specific heats.

Properties The gas constant of air is 0.287 kPa.m³/kg.K (Table A-1). The enthalpies of air are (Table A-17)

$$T_1 = 550 \text{ K} \rightarrow h_1 = 555.74 \text{ kJ/kg}$$

 $T_3 = 800 \text{ K} \rightarrow h_3 = 821.95 \text{ kJ/kg}$
 $T_4 = 600 \text{ K} \rightarrow h_4 = 607.02 \text{ kJ/kg}$

Analysis (*a*) We take the *air side* of the heat exchanger as the system, which is a control volume since mass crosses the boundary. There is only one inlet and one exit, and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. The energy balance for this steady-flow system can be expressed in the rate form as





Substituting,

$$3200 \text{ kJ/s} = (800/60 \text{ kg/s})(h_2 - 554.71 \text{ kJ/kg}) \rightarrow h_2 = 794.71 \text{ kJ/kg}$$

Then from Table A-17 we read

 $T_2 = 775.1 \text{ K}$

(b) Treating the exhaust gases as an ideal gas, the mass flow rate of the exhaust gases is determined from the steady-flow energy relation applied only to the exhaust gases,

$$\dot{E}_{in} = \dot{E}_{out}$$

$$\dot{m}_{exhaust}h_3 = \dot{Q}_{out} + \dot{m}_{exhaust}h_4 \quad (since \ \dot{W} = \Delta ke \cong \Delta pe \cong 0)$$

$$\dot{Q}_{out} = \dot{m}_{exhaust}(h_3 - h_4)$$
3200 kJ/s = $\dot{m}_{exhaust}(821.95 - 607.02)$ kJ/kg

It yields

$$\dot{m}_{\rm exhaust} =$$
 14.9 kg/s

Assumptions 1 This is a steady-flow process since there is no change with time at any point within the system and thus $\Delta m_{\rm CV} = 0$ and $\Delta E_{\rm CV} = 0$. 2 Water is an incompressible substance with constant specific heats. 3 The kinetic and potential energy changes are negligible, $\Delta ke \cong \Delta pe \cong 0$. 4 The pipe is insulated and thus the heat losses are negligible.

Properties The density and specific heat of water at room temperature are $\rho = 1000 \text{ kg/m}^3$ and $c = 4.18 \text{ kJ/kg} \cdot \text{°C}$ (Table A-3).

Analysis (*a*) We take the pipe as the system. This is a *control volume* since mass crosses the system boundary during the process. Also, there is only one inlet and one exit and thus $\dot{m}_1 = \dot{m}_2 = \dot{m}$. The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{Rate of net energy transfer} = \underbrace{\Delta \dot{E}_{system}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{Rate of change in internal, kinetic, potential, etc. energies} = 0$$

$$\underbrace{\dot{W}_{e,in} + \dot{m}h_1 = \dot{m}h_2}_{W_{e,in}} (since \ \dot{Q}_{out} \cong \Delta ke \cong \Delta pe \cong 0)$$

$$\underbrace{\dot{W}_{e,in} = \dot{m}(h_2 - h_1) = \dot{m}[c(T_2 - T_1) + \upsilon \Delta P^{\phi^{\phi_0}}] = \dot{m}c(T_2 - T_1)$$

$$WATER \longrightarrow D = 5 \text{ cm}$$

$$\underbrace{Watter}_{V_{e,in}} = \dot{m}(h_2 - h_1) = \dot{m}[c(T_2 - T_1) + \upsilon \Delta P^{\phi^{\phi_0}}] = \dot{m}c(T_2 - T_1)$$

The mass flow rate of water through the pipe is

$$\dot{m} = \rho \dot{V}_1 = (1000 \text{ kg/m}^3)(0.030 \text{ m}^3/\text{min}) = 30 \text{ kg/min}$$

Therefore,

$$\dot{W}_{e,in} = \dot{m}c(T_2 - T_1) = (30/60 \text{ kg/s})(4.18 \text{ kJ/kg} \circ \text{C})(55 - 20) \circ \text{C} = 73.2 \text{ kW}$$

(b) The average velocity of water through the pipe is determined from

$$V = \frac{\dot{V}}{A} = \frac{\dot{V}}{\pi r^2} = \frac{0.030 \text{ m}^3/\text{min}}{\pi (0.025 \text{ m})^2} = 15.3 \text{ m/min}$$

5-191 An insulated cylinder equipped with an external spring initially contains air. The tank is connected to a supply line, and air is allowed to enter the cylinder until its volume doubles. The mass of the air that entered and the final temperature in the cylinder are to be determined.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. 2 The expansion process is quasi-equilibrium. 3 Kinetic and potential energies are negligible. 4 The spring is a linear spring. 5 The device is insulated and thus heat transfer is negligible. 6 Air is an ideal gas with constant specific heats.

Properties The gas constant of air is R = 0.287 kJ/kg·K (Table A-1). The specific heats of air at room temperature are $c_v = 0.718$ and $c_p = 1.005$ kJ/kg·K (Table A-2a). Also, $u = c_v T$ and $h = c_p T$.

Analysis We take the cylinder as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_i = m_2 - m_1$$

Energy balance:

or,

 $\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}}$

$$m_i h_i = W_{b,out} + m_2 u_2 - m_1 u_1$$
 (since $Q \cong ke \cong pe \cong 0$)

Combining the two relations, $(m_2 - m_1)h_i = W_{h out} + m_2u_2 - m_1u_1$

$$(m_2 - m_1)c_pT_i = W_{b.out} + m_2c_pT_2 - m_1c_pT_1$$

The initial and the final masses in the tank are

$$m_{1} = \frac{P_{1}\boldsymbol{V}_{1}}{RT_{1}} = \frac{(150 \text{ kPa})(0.11 \text{ m}^{3})}{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})(295 \text{ K})} = 0.1949 \text{ kg}$$
$$m_{2} = \frac{P_{2}\boldsymbol{V}_{2}}{RT_{2}} = \frac{(600 \text{ kPa})(0.22 \text{ m}^{3})}{(0.287 \text{ kPa} \cdot \text{m}^{3}/\text{kg} \cdot \text{K})T_{2}} = \frac{459.9}{T_{2}}$$

Then from the mass balance becomes $m_i = m_2 - m_1 = \frac{459.9}{T_2} - 0.1949$

The spring is a linear spring, and thus the boundary work for this process can be determined from

$$W_b = Area = \frac{P_1 + P_2}{2} (\mathbf{V}_2 - \mathbf{V}_1) = \frac{(150 + 600) \text{kPa}}{2} (0.22 - 0.11) \text{m}^3 = 41.25 \text{ kJ}$$

Substituting into the energy balance, the final temperature of air T_2 is determined to be

$$-41.25 = -\left(\frac{459.9}{T_2} - 0.1949\right)(1.005)(295) + \left(\frac{459.9}{T_2}\right)(0.718)(T_2) - (0.1949)(0.718)(295)$$

It yields

Thus.

$$m_2 = \frac{459.9}{T_2} = \frac{459.9}{351.4} = 1.309 \text{ kg}$$

 $m_{\rm i} = m_2 - m_1 = 1.309 - 0.1949 = 1.11 \, \rm kg$

 $T_2 = 351 \, \mathrm{K}$

and

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5-192 R-134a is allowed to leave a piston-cylinder device with a pair of stops. The work done and the heat transfer are to be determined.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid leaving the device is assumed to be constant. 2 Kinetic and potential energies are negligible.

Properties The properties of R-134a at various states are (Tables A-11 through A-13)

$$P_{1} = 800 \text{ kPa} \begin{cases} \boldsymbol{v}_{1} = 0.032659 \text{ m}^{3}/\text{kg} \\ u_{1} = 290.84 \text{ kJ/kg} \\ h_{1} = 316.97 \text{ kJ/kg} \end{cases}$$
$$P_{2} = 0.042115 \text{ m}^{3}/\text{kg}$$

 $P_{2} = 500 \text{ kPa} \begin{cases} v_{2} = 0.0 \text{ kJ/kg} \\ u_{2} = 242.40 \text{ kJ/kg} \\ h_{2} = 263.46 \text{ kJ/kg} \end{cases}$



Analysis (*a*) We take the tank as the system, which is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_e = m_1 - m_2$$

Energy balance:

$$\underbrace{E_{in} - E_{out}}_{\substack{\text{Net energy transfer}\\ \text{by heat, work, and mass}}} = \underbrace{\Delta E_{\text{system}}}_{\substack{\text{Change in internal, kinetic, potential, etc. energies}}}$$
$$W_{b,in} - Q_{out} - m_e h_e = m_2 u_2 - m_1 u_1 \quad (\text{since } \text{ke} \cong \text{pe} \cong 0)$$

The volumes at the initial and final states and the mass that has left the cylinder are

$$V_1 = m_1 v_1 = (2 \text{ kg})(0.032659 \text{ m}^3/\text{kg}) = 0.06532 \text{ m}^3$$

 $V_2 = m_2 v_2 = (1/2)m_1 v_2 = (1/2)(2 \text{ kg})(0.042115 \text{ m}^3/\text{kg}) = 0.04212 \text{ m}^3$
 $m_e = m_1 - m_2 = 2 - 1 = 1 \text{ kg}$

The enthalpy of the refrigerant withdrawn from the cylinder is assumed to be the average of initial and final enthalpies of the refrigerant in the cylinder

$$h_e = (1/2)(h_1 + h_2) = (1/2)(316.97 + 263.46) = 290.21 \text{ kJ/kg}$$

Noting that the pressure remains constant after the piston starts moving, the boundary work is determined from

$$W_{\rm h in} = P_2(V_1 - V_2) = (500 \text{ kPa})(0.06532 - 0.04212)\text{m}^3 = 11.6 \text{ kJ}$$

(b) Substituting,

11.6 kJ –
$$Q_{out}$$
 – (1 kg)(290.21 kJ/kg) = (1 kg)(242.40 kJ/kg) – (2 kg)(290.84 kJ/kg)
 Q_{out} = **60.7 kJ**

5-193 The pressures across a pump are measured. The mechanical efficiency of the pump and the temperature rise of water are to be determined.

Assumptions 1 The flow is steady and incompressible. 2 The pump is driven by an external motor so that the heat generated by the motor is dissipated to the atmosphere. 3 The elevation difference between the inlet and outlet of the pump is negligible, $z_1 = z_2$. 4 The inlet and outlet diameters are the same and thus the inlet and exit velocities are equal, $V_1 = V_2$.

Properties We take the density of water to be $1 \text{ kg/L} = 1000 \text{ kg/m}^3$ and its specific heat to be $4.18 \text{ kJ/kg} \cdot ^{\circ}\text{C}$ (Table A-3).

Analysis (a) The mass flow rate of water through the pump is

$$\dot{m} = \rho V = (1 \text{ kg/L})(18 \text{ L/s}) = 18 \text{ kg/s}$$

The motor draws 6 kW of power and is 95 percent efficient. Thus the mechanical (shaft) power it delivers to the pump is

$$\dot{W}_{\text{pump,shaft}} = \eta_{\text{motor}} \dot{W}_{\text{electric}} = (0.95)(6 \text{ kW}) = 5.7 \text{ kW}$$

To determine the mechanical efficiency of the pump, we need to know the increase in the mechanical energy of the fluid as it flows through the pump, which is

$$\Delta \dot{E}_{\text{mech,fluid}} = \dot{E}_{\text{mech,out}} - \dot{E}_{\text{mech,in}} = \dot{m} \left(\frac{P_2}{\rho} + \frac{V_2^2}{2} + gz_2 \right) - \dot{m} \left(\frac{P_1}{\rho} + \frac{V_1^2}{2} + gz_1 \right)$$

Simplifying it for this case and substituting the given values,

$$\Delta \dot{E}_{\text{mech,fluid}} = \dot{m} \left(\frac{P_2 - P_1}{\rho} \right) = (18 \text{ kg/s}) \left(\frac{(300 - 100) \text{ kPa}}{1000 \text{ kg/m}^3} \right) \left(\frac{1 \text{ kJ}}{1 \text{ kPa} \cdot \text{m}^3} \right) = 3.6 \text{ kW}$$

Then the mechanical efficiency of the pump becomes

$$\eta_{pump} = \frac{\Delta E_{\text{mech,fluid}}}{\dot{W}_{\text{pump shaft}}} = \frac{3.6 \text{ kW}}{5.7 \text{ kW}} = 0.632 = 63.2\%$$

(*b*) Of the 5.7-kW mechanical power supplied by the pump, only 3.6 kW is imparted to the fluid as mechanical energy. The remaining 2.1 kW is converted to thermal energy due to frictional effects, and this "lost" mechanical energy manifests itself as a heating effect in the fluid,

$$\dot{E}_{\text{mech,loss}} = \dot{W}_{\text{pump,shaft}} - \Delta \dot{E}_{\text{mech,fluid}} = 5.7 - 3.6 = 2.1 \,\text{kW}$$

The temperature rise of water due to this mechanical inefficiency is determined from the thermal energy balance,

$$\dot{E}_{\text{mech,loss}} = \dot{m}(u_2 - u_1) = \dot{m}c\Delta T$$

Solving for ΔT ,

$$\Delta T = \frac{E_{\text{mech,loss}}}{\dot{m}c} = \frac{2.1 \,\text{kW}}{(18 \,\text{kg/s})(4.18 \,\text{kJ/kg} \cdot \text{K})} = 0.028 \,^{\circ}\text{C}$$

Therefore, the water will experience a temperature rise of 0.028°C, which is very small, as it flows through the pump.

Discussion In an actual application, the temperature rise of water will probably be less since part of the heat generated will be transferred to the casing of the pump and from the casing to the surrounding air. If the entire pump motor were submerged in water, then the 2.1 kW dissipated to the air due to motor inefficiency would also be transferred to the surrounding water as heat. This would cause the water temperature to rise more.



5-194 A water tank is heated by electricity. The water withdrawn from the tank is mixed with cold water in a chamber. The mass flow rate of hot water withdrawn from the tank and the average temperature of mixed water are to be determined.

Assumptions 1 The process in the mixing chamber is a steady-flow process since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 There are no work interactions. 4 The device is adiabatic and thus heat transfer is negligible.

Properties The specific heat and density of water are taken to be $c_p = 4.18 \text{ kJ/kg.K}, \rho = 1000 \text{ kg/m}^3$ (Table A-3).

Analysis We take the mixing chamber as the system, which is a control volume since mass crosses the boundary. The energy balance for this steady-flow system can be expressed in the rate form as

Energy balance:

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer}} = \underbrace{\Delta \dot{E}_{\text{system}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}}^{\pi 0 \text{ (steady)}} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{m}_1 h_1 + \dot{m}_2 h_2} = \dot{m}_3 h_3 \quad (\text{since } \dot{Q} \cong \dot{W} \cong \Delta \text{ke} \cong \Delta \text{pe} \cong 0)$$

C
Tank

$$T_1 = 80^{\circ}C$$

 $T_2 = 60^{\circ}C$
 \dot{m}
Mixing
chamber
 $20^{\circ}C$
 0.06 kg/s

or

Similarly, an energy balance may be written on the water tank as

 $\dot{m}_{\rm hot}c_p T_{\rm tank,ave} + \dot{m}_{\rm cold}c_p T_{\rm cold} = (\dot{m}_{\rm hot} + \dot{m}_{\rm cold})c_p T_{\rm mixture}$

$$\dot{W}_{e,in} + \dot{m}_{hot} c_p (T_{cold} - T_{tank,ave}) \Delta t = m_{tank} c_p (T_{tank,2} - T_{tank,1})$$
(2)

where

$$T_{\text{tank,ave}} = \frac{T_{\text{tank},1} + T_{\text{tank},2}}{2} = \frac{80 + 60}{2} = 70^{\circ}\text{C}$$

and

$$m_{\text{tank}} = \rho \mathbf{V} = (1000 \,\text{kg/m}^3)(0.060 \,\text{m}^3) = 60 \,\text{kg}$$

Substituting into Eq. (2),

$$[1.6 \text{ kJ/s} + \dot{m}_{\text{hot}} (4.18 \text{ kJ/kg.}^{\circ}\text{C})(20 - 70)^{\circ}\text{C}](8 \times 60 \text{ s}) = (60 \text{ kg})(4.18 \text{ kJ/kg.}^{\circ}\text{C})(60 - 80)^{\circ}\text{C}$$
$$\longrightarrow \dot{m}_{\text{hot}} = \mathbf{0.0577 \text{ kg/s}}$$

Substituting into Eq. (1),

$$(0.0577 \text{ kg/s})(4.18 \text{ kJ/kg.}^{\circ}\text{C})(70^{\circ}\text{C}) + (0.06 \text{ kg/s})(4.18 \text{ kJ/kg.}^{\circ}\text{C})(20^{\circ}\text{C})$$

= $[(0.0577 + 0.06)kg/s](4.18 kJ/kg.^{\circ}C)T_{mixture}$

20°

(1)

$$\rightarrow T_{\text{mixture}} = 44.5^{\circ}\text{C}$$

5-195 The turbocharger of an internal combustion engine consisting of a turbine, a compressor, and an aftercooler is considered. The temperature of the air at the compressor outlet and the minimum flow rate of ambient air are to be determined.

Assumptions 1 All processes are steady since there is no change with time. 2 Kinetic and potential energy changes are negligible. 3 Air properties are used for exhaust gases. 4 Air is an ideal gas with constant specific heats. 5 The mechanical efficiency between the turbine and the compressor is 100%. 6 All devices are adiabatic. 7 The local atmospheric pressure is 100 kPa.

Properties The constant pressure specific heats of exhaust gases, warm air, and cold ambient air are taken to be $c_p = 1.063$, 1.008, and 1.005 kJ/kg·K, respectively (Table A-2b).



Analysis (a) An energy balance on turbine gives

$$\dot{W}_{\rm T} = \dot{m}_{\rm exh} c_{p,\rm exh} \left(T_{\rm exh,1} - T_{\rm exh,2} \right) = (0.02 \text{ kg/s})(1.063 \text{ kJ/kg} \cdot \text{K})(400 - 350)\text{K} = 1.063 \text{ kW}$$

This is also the power input to the compressor since the mechanical efficiency between the turbine and the compressor is assumed to be 100%. An energy balance on the compressor gives the air temperature at the compressor outlet

$$\dot{W}_{\rm C} = \dot{m}_{\rm a} c_{p,\rm a} (T_{\rm a,2} - T_{\rm a,1})$$

1.063 kW = (0.018 kg/s)(1.008 kJ/kg·K)($T_{\rm a,2} - 50$)K $\longrightarrow T_{\rm a,2} =$ **108.6** °C

(b) An energy balance on the aftercooler gives the mass flow rate of cold ambient air

$$\dot{m}_{a}c_{p,a}(T_{a,2} - T_{a,3}) = \dot{m}_{ca}c_{p,ca}(T_{ca,2} - T_{ca,1})$$
(0.018 kg/s)(1.008 kJ/kg · °C)(108.6 - 80)°C = $\dot{m}_{ca}(1.005 \text{ kJ/kg} \cdot °C)(40 - 30)$ °C
 $\dot{m}_{ca} = 0.05161 \text{ kg/s}$

The volume flow rate may be determined if we first calculate specific volume of cold ambient air at the inlet of aftercooler. That is,

$$v_{ca} = \frac{RT}{P} = \frac{(0.287 \text{ kJ/kg} \cdot \text{K})(30 + 273 \text{ K})}{100 \text{ kPa}} = 0.8696 \text{ m}^3/\text{kg}$$

 $\dot{V}_{ca} = \dot{m}v_{ca} = (0.05161 \text{ kg/s})(0.8696 \text{ m}^3/\text{kg}) = 0.0449 \text{ m}^3/\text{s} = 44.9 \text{ L/s}$

5-196 Heat is transferred to a pressure cooker at a specified rate for a specified time period. The cooking temperature and the water remaining in the cooker are to be determined.

Assumptions 1 This process can be analyzed as a *uniform-flow process* since the properties of the steam leaving the control volume remain constant during the entire cooking process. 2 The kinetic and potential energies of the streams

are negligible, ke \cong pe \cong 0. **3** The pressure cooker is stationary and thus its kinetic and potential energy changes are zero; that is, $\Delta KE = \Delta PE = 0$ and $\Delta E_{system} = \Delta U_{system}$. **4** The pressure (and thus temperature) in the pressure

cooker remains constant. **5** Steam leaves as a saturated vapor at the cooker pressure. **6** There are no boundary, electrical, or shaft work interactions involved. **7** Heat is transferred to the cooker at a constant rate.

Analysis We take the *pressure cooker* as the system. This is a *control volume* since mass crosses the system boundary during the process. We observe that this is an unsteady-flow process since changes occur within the control volume. Also, there is one exit and no inlets for mass flow.

(a) The absolute pressure within the cooker is

$$P_{\rm abs} = P_{\rm gage} + P_{\rm atm} = 75 + 100 = 175 \, \rm kPa$$

Since saturation conditions exist in the cooker at all times, the cooking temperature must be the saturation temperature corresponding to this pressure. From Table A–5, it is

 $T_2 = T_{\text{sat } @ 175 \text{ kPa}} = 116.04^{\circ}\text{C}$

which is about 16°C higher than the ordinary cooking temperature.

(b) Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow - m_e = (m_2 - m_1)_{\rm CV}$$
 or $m_e = (m_1 - m_2)_{\rm CV}$

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{potential, etc. energies}}$$
$$\underbrace{\Delta E_{\text{system}}}_{p_{\text{potential, etc. energies}}}$$
$$\underbrace{Q_{\text{in}} - m_e h_e}_{p_e} = m_2 u_2 - m_1 u_1 \text{ (since } W \cong ke \cong pe \cong 0)$$

Combining the mass and energy balances gives

$$Q_{\rm in} = (m_1 - m_2)h_e + m_2u_2 - m_1u_1$$

The amount of heat transfer during this process is found from

$$Q_{\rm in} = \dot{Q}_{\rm in} \Delta T = (0.5 \, \rm kJ/s)(30 \times 60 \, \rm s) = 900 \, \rm kJ$$

Steam leaves the pressure cooker as saturated vapor at 175 kPa at all times. Thus,

$$h_e = h_{g@175 \text{ kPa}} = 2700.2 \text{ kJ/kg}$$

The initial internal energy is found after the quality is determined:

$$\mathbf{v}_1 = \frac{\mathbf{V}}{m_1} = \frac{0.006 \text{ m}^3}{1 \text{ kg}} = 0.006 \text{ m}^3/\text{kg}$$

 $x_1 = \frac{\mathbf{v}_1 - \mathbf{v}_f}{\mathbf{v}_{f_1}} = \frac{0.006 - 0.001}{1.004 - 0.001} = 0.00499$

Thus,

$$u_1 = u_f + x_1 u_{fg} = 486.82 + (0.00499)(2037.7) \text{ kJ/kg} = 497 \text{ kJ/kg}$$

and





The mass of the system at the final state is $m_2 = V/v_2$. Substituting this into the energy equation yields

$$Q_{\rm in} = \left(m_1 - \frac{\boldsymbol{\nu}}{\boldsymbol{\nu}_2}\right)h_e + \left(\frac{\boldsymbol{\nu}}{\boldsymbol{\nu}_2}u_2 - m_1u_1\right)$$

There are two unknowns in this equation, u_2 and v_2 . Thus we need to relate them to a single unknown before we can determine these unknowns. Assuming there is still some liquid water left in the cooker at the final state (i.e., saturation conditions exist), v_2 and u_2 can be expressed as

$$\mathbf{v}_2 = \mathbf{v}_f + x_2 \mathbf{v}_{fg} = 0.001 + x_2 (1.004 - 0.001) \text{ m}^3/\text{kg}$$

 $u_2 = u_f + x_2 u_{fg} = 486.82 + x_2 (2037.7) \text{ kJ/kg}$

Recall that during a boiling process at constant pressure, the properties of each phase remain constant (only the amounts change). When these expressions are substituted into the above energy equation, x_2 becomes the only unknown, and it is determined to be

$$x_2 = 0.009$$

Thus,

$$v_2 = 0.001 + (0.009)(1.004 - 0.001) \text{ m}^3/\text{kg} = 0.010 \text{ m}^3/\text{kg}$$

and

$$m_2 = \frac{\mathbf{v}}{\mathbf{v}_2} = \frac{0.006 \text{ m}^3}{0.01 \text{ m}^3/\text{kg}} = 0.6 \text{ kg}$$

Therefore, after 30 min there is 0.6 kg water (liquid + vapor) left in the pressure cooker.

Discussion Note that almost half of the water in the pressure cooker has evaporated during cooking.

5-197 A water tank open to the atmosphere is initially filled with water. The tank discharges to the atmosphere through a long pipe connected to a valve. The initial discharge velocity from the tank and the time required to empty the tank are to be determined.

Assumptions 1 The flow is incompressible. 2 The draining pipe is horizontal. 3 The tank is considered to be empty when the water level drops to the center of the valve.

Analysis (a) Substituting the known quantities, the discharge velocity can be expressed as

$$V = \sqrt{\frac{2gz}{1.5 + fL/D}} = \sqrt{\frac{2gz}{1.5 + 0.015(100 \text{ m})/(0.10 \text{ m})}} = \sqrt{0.1212gz}$$

Then the initial discharge velocity becomes

$$V_1 = \sqrt{0.1212gz_1} = \sqrt{0.1212(9.81 \text{ m/s}^2)(2 \text{ m})} = 1.54 \text{ m/s}^2$$

where z is the water height relative to the center of the orifice at that time.

(*b*) The flow rate of water from the tank can be obtained by multiplying the discharge velocity by the pipe cross-sectional area,

$$\dot{\boldsymbol{\nu}} = A_{\text{pipe}} V_2 = \frac{\pi D^2}{4} \sqrt{0.1212gz}$$

Then the amount of water that flows through the pipe during a differential time interval dt is

$$d\boldsymbol{V} = \dot{\boldsymbol{V}}dt = \frac{\pi D^2}{4}\sqrt{0.1212\,gz}dt \tag{1}$$

which, from conservation of mass, must be equal to the decrease in the volume of water in the tank,

$$d\boldsymbol{V} = A_{\text{tank}}(-dz) = -\frac{\pi D_0^2}{4} dz$$
(2)

where dz is the change in the water level in the tank during dt. (Note that dz is a negative quantity since the positive direction of z is upwards. Therefore, we used -dz to get a positive quantity for the amount of water discharged). Setting Eqs. (1) and (2) equal to each other and rearranging,

$$\frac{\pi D^2}{4} \sqrt{0.1212 gz} dt = -\frac{\pi D_0^2}{4} dz \rightarrow dt = -\frac{D_0^2}{D^2} \frac{dz}{\sqrt{0.1212 gz}} = -\frac{D_0^2}{D^2 \sqrt{0.1212 gz}} z^{-\frac{1}{2}} dz$$

The last relation can be integrated easily since the variables are separated. Letting t_f be the discharge time and integrating it from t = 0 when $z = z_1$ to $t = t_f$ when z = 0 (completely drained tank) gives

$$\int_{t=0}^{t_f} dt = -\frac{D_0^2}{D^2 \sqrt{0.1212g}} \int_{z=z_1}^0 z^{-1/2} dz \rightarrow t_f = -\frac{D_0^2}{D^2 \sqrt{0.1212g}} \left| \frac{z^{\frac{1}{2}}}{\frac{1}{2}} \right|_{z_1}^0 = \frac{2D_0^2}{D^2 \sqrt{0.1212g}} z_1^{\frac{1}{2}}$$

Simplifying and substituting the values given, the draining time is determined to be

$$t_f = \frac{2D_0^2}{D^2} \sqrt{\frac{z_1}{0.1212g}} = \frac{2(10 \text{ m})^2}{(0.1 \text{ m})^2} \sqrt{\frac{2 \text{ m}}{0.1212(9.81 \text{ m/s}^2)}} = 25,940 \text{ s} = 7.21 \text{ h}$$

Discussion The draining time can be shortened considerably by installing a pump in the pipe.





5-198 A fluid is flowing in a circular pipe. A relation is to be obtained for the average fluid velocity in therms of V(r), R, and r.

Analysis Choosing a circular ring of area $dA = 2\pi r dr$ as our differential area, the mass flow rate through a cross-sectional area can be expressed as

$$\dot{m} = \int_{A} \rho V(r) dA = \int_{0}^{R} \rho V(r) 2\pi r dr$$

Solving for V_{avg} ,

$$V_{\rm avg} = \frac{2}{R^2} \int_0^R V(r) r \, dr$$



5-199 Two streams of same ideal gas at different states are mixed in a mixing chamber. The simplest expression for the mixture temperature in a specified format is to be obtained.

Analysis The energy balance for this steady-flow system can be expressed in the rate form as

$$\underbrace{\dot{E}_{in} - \dot{E}_{out}}_{\text{Rate of net energy transfer by heat, work, and mass}} = \underbrace{\Delta \dot{E}_{system}^{70 \text{ (steady)}}}_{\text{Rate of change in internal, kinetic, potential, etc. energies}} = 0$$

$$\underbrace{\dot{E}_{in} = \dot{E}_{out}}_{\dot{m}_1 h_1 + \dot{m}_2 h_2} = \dot{m}_3 h_3 \quad (\text{since } \dot{Q} = \dot{W} = 0)$$

$$\dot{m}_1 c_p T_1 + \dot{m}_2 c_p T_2 = \dot{m}_3 c_p T_3$$



and,

$$\dot{m}_3 = \dot{m}_1 + \dot{m}_2$$

.

Solving for final temperature, we find

$$T_3 = \frac{m_1}{\dot{m}_3} T_1 + \frac{m_2}{\dot{m}_3} T_2$$

5-200 A rigid container filled with an ideal gas is heated while gas is released so that the temperature of the gas remaining in the container stays constant. An expression for the mass flow rate at the outlet as a function of the rate of pressure change in the container is to be derived.

Analysis At any instant, the mass in the control volume may be expressed as

$$m_{\rm CV} = \frac{V}{V} = \frac{V}{RT} P$$

Since there are no inlets to this control volume, the conservation of mass principle becomes

$$\dot{m}_{\rm in} - \dot{m}_{\rm out} = \frac{dm_{\rm CV}}{dt}$$
$$\dot{m}_{\rm out} = -\frac{dm_{\rm CV}}{dt} = -\frac{V}{RT}\frac{dP}{dt}$$

5-201 An evacuated bottle is surrounded by atmospheric air. A valve is opened, and air is allowed to fill the bottle. The amount of heat transfer through the wall of the bottle when thermal and mechanical equilibrium is established is to be determined.

Assumptions 1 This is an unsteady process since the conditions within the device are changing during the process, but it can be analyzed as a uniform-flow process since the state of fluid at the inlet remains constant. 2 Air is an ideal gas. 3 Kinetic and potential energies are negligible. 4 There are no work interactions involved. 5 The direction of heat transfer is to the air in the bottle (will be verified).

Analysis We take the bottle as the system. It is a control volume since mass crosses the boundary. Noting that the microscopic energies of flowing and nonflowing fluids are represented by enthalpy h and internal energy u, respectively, the mass and energy balances for this uniform-flow system can be expressed as

Mass balance:

$$m_{\rm in} - m_{\rm out} = \Delta m_{\rm system} \rightarrow m_i = m_2$$
 (since $m_{\rm out} = m_{\rm initial} = 0$)

Energy balance:

$$\underbrace{E_{\text{in}} - E_{\text{out}}}_{\text{Net energy transfer}} = \underbrace{\Delta E_{\text{system}}}_{\text{Change in internal, kinetic, potential, etc. energies}}$$
$$Q_{\text{in}} + m_i h_i = m_2 u_2 \quad (\text{since } W \cong E_{\text{out}} = E_{\text{initial}} = \text{ke} \cong \text{pe} \cong 0)$$

Combining the two balances:

$$Q_{\rm in} = m_2(u_2 - h_i) = m_2(c_v T_2 - c_p T_i)$$

but

$$T_{\rm i} = T_2 = T_0$$

and

$$c_p - c_v = R$$

Substituting,

$$Q_{\rm in} = m_2 (c_v - c_p) T_0 = -m_2 R T_0 = -\frac{P_0 V}{R T_0} R T_0 = -P_0 V$$

Therefore,

 $Q_{\text{out}} = P_0 V$ (Heat is lost from the tank)



Fundamentals of Engineering (FE) Exam Problems

5-202 Steam is accelerated by a nozzle steadily from a low velocity to a velocity of 280 m/s at a rate of 2.5 kg/s. If the temperature and pressure of the steam at the nozzle exit are 400°C and 2 MPa, the exit area of the nozzle is

(a) 8.4 cm^2 (b) 10.7 cm^2 (c) 13.5 cm^2 (d) 19.6 cm^2 (e) 23.0 cm^2 Answer (c) 13.5 cm^2

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

Vel_1=0 "m/s" Vel_2=280 "m/s" m=2.5 "kg/s" T2=400 "C" P2=2000 "kPa" "The rate form of energy balance is E_dot_in - E_dot_out = DELTAE_dot_cv" v2=VOLUME(Steam_IAPWS,T=T2,P=P2) m=(1/v2)*A2*Vel_2 "A2 in m^2"

"Some Wrong Solutions with Common Mistakes:" R=0.4615 "kJ/kg.K" P2*v2ideal=R*(T2+273) m=(1/v2ideal)*W1_A2*Vel_2 "assuming ideal gas" P1*v2ideal=R*T2 m=(1/v2ideal)*W2_A2*Vel_2 "assuming ideal gas and using C" m=W3_A2*Vel_2 "not using specific volume" (a) 15 cm^2 (b) 50 cm^2 (c) 105 cm^2 (d) 150 cm^2 (e) 190 cm^2

Answer (d) 150 cm^2

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

Vel_1=122 "m/s" m=3.5 "kg/s" T1=300 "C" P1=500 "kPa" "The rate form of energy balance is E_dot_in - E_dot_out = DELTAE_dot_cv" v1=VOLUME(Steam_IAPWS,T=T1,P=P1) m=(1/v1)*A*Vel_1 "A in m^2"

"Some Wrong Solutions with Common Mistakes:" R=0.4615 "kJ/kg.K" P1*v1ideal=R*(T1+273) m=(1/v1ideal)*W1_A*Vel_1 "assuming ideal gas" P1*v2ideal=R*T1 m=(1/v2ideal)*W2_A*Vel_1 "assuming ideal gas and using C" m=W3_A*Vel_1 "not using specific volume"

5-204 An adiabatic heat exchanger is used to heat cold water at 15° C entering at a rate of 5 kg/s by hot air at 90°C entering also at rate of 5 kg/s. If the exit temperature of hot air is 20°C, the exit temperature of cold water is

(a) 27° C (b) 32° C (c) 52° C (d) 85° C (e) 90° C Answer (b) 32° C

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

C_w=4.18 "kJ/kg-C" Cp_air=1.005 "kJ/kg-C" Tw1=15 "C" m_dot_w=5 "kg/s" Tair1=90 "C" Tair2=20 "C" m_dot_air=5 "kg/s" "The rate form of energy balance for a steady-flow system is E_dot_in = E_dot_out" m_dot_air*Cp_air*(Tair1-Tair2)=m_dot_w*C_w*(Tw2-Tw1)

"Some Wrong Solutions with Common Mistakes:" (Tair1-Tair2)=(W1_Tw2-Tw1) "Equating temperature changes of fluids" Cv_air=0.718 "kJ/kg.K" m_dot_air*Cv_air*(Tair1-Tair2)=m_dot_w*C_w*(W2_Tw2-Tw1) "Using Cv for air" W3_Tw2=Tair1 "Setting inlet temperature of hot fluid = exit temperature of cold fluid" W4_Tw2=Tair2 "Setting exit temperature of hot fluid = exit temperature of cold fluid" **5-205** A heat exchanger is used to heat cold water at 15°C entering at a rate of 2 kg/s by hot air at 85°C entering at rate of 3 kg/s. The heat exchanger is not insulated, and is loosing heat at a rate of 25 kJ/s. If the exit temperature of hot air is 20°C, the exit temperature of cold water is

(a) 28° C (b) 35° C (c) 38° C (d) 41° C (e) 80° C

Answer (b) 35°C

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

"Some Wrong Solutions with Common Mistakes:"

m_dot_air*Cp_air*(Tair1-Tair2)=m_dot_w*C_w*(W1_Tw2-Tw1) "Not considering Q_loss" m_dot_air*Cp_air*(Tair1-Tair2)=m_dot_w*C_w*(W2_Tw2-Tw1)-Q_loss "Taking heat loss as heat gain" (Tair1-Tair2)=(W3_Tw2-Tw1) "Equating temperature changes of fluids" Cv_air=0.718 "kJ/kg.K" m_dot_air*Cv_air*(Tair1-Tair2)=m_dot_w*C_w*(W4_Tw2-Tw1)+Q_loss "Using Cv for air"

5-206 An adiabatic heat exchanger is used to heat cold water at 15°C entering at a rate of 5 kg/s by hot water at 90°C entering at rate of 4 kg/s. If the exit temperature of hot water is 50°C, the exit temperature of cold water is

(a) 42° C (b) 47° C (c) 55° C (d) 78° C (e) 90° C

Answer (b) 47°C

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

C_w=4.18 "kJ/kg-C" Tcold_1=15 "C" m_dot_cold=5 "kg/s" Thot_1=90 "C" Thot_2=50 "C" m_dot_hot=4 "kg/s" Q_loss=0 "kJ/s" "The rate form of energy balance for a steady-flow system is E_dot_in = E_dot_out" m_dot_hot*C_w*(Thot_1-Thot_2)=m_dot_cold*C_w*(Tcold_2-Tcold_1)+Q_loss

"Some Wrong Solutions with Common Mistakes:"

Thot_1-Thot_2=W1_Tcold_2-Tcold_1 "Equating temperature changes of fluids" W2_Tcold_2=90 "Taking exit temp of cold fluid=inlet temp of hot fluid"

5-207 In a shower, cold water at 10°C flowing at a rate of 5 kg/min is mixed with hot water at 60°C flowing at a rate of 2 kg/min. The exit temperature of the mixture will be

(a) 24.3° C (b) 35.0° C (c) 40.0° C (d) 44.3° C (e) 55.2° C Answer (a) 24.3° C

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

C_w=4.18 "kJ/kg-C" Tcold_1=10 "C" m_dot_cold=5 "kg/min" Thot_1=60 "C" m_dot_hot=2 "kg/min" "The rate form of energy balance for a steady-flow system is E_dot_in = E_dot_out" m_dot_hot*C_w*Thot_1+m_dot_cold*C_w*Tcold_1=(m_dot_hot+m_dot_cold)*C_w*Tmix "Some Wrong Solutions with Common Mistakes:" W1_Tmix=(Tcold_1+Thot_1)/2 "Taking the average temperature of inlet fluids"

5-208 In a heating system, cold outdoor air at 7°C flowing at a rate of 4 kg/min is mixed adiabatically with heated air at 70°C flowing at a rate of 3 kg/min. The exit temperature of the mixture is

(a) 34° C (b) 39° C (c) 45° C (d) 63° C (e) 77° C Answer (a) 34° C

Answer (a) 54 C

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

C_air=1.005 "kJ/kg-C" Tcold_1=7 "C" m_dot_cold=4 "kg/min" Thot_1=70 "C" m_dot_hot=3 "kg/min" "The rate form of energy balance for a steady-flow system is E_dot_in = E_dot_out" m_dot_hot*C_air*Thot_1+m_dot_cold*C_air*Tcold_1=(m_dot_hot+m_dot_cold)*C_air*Tmix "Some Wrong Solutions with Common Mistakes:" W1_Tmix=(Tcold_1+Thot_1)/2 "Taking the average temperature of inlet fluids" **5-209** Hot combustion gases (assumed to have the properties of air at room temperature) enter a gas turbine at 1 MPa and 1500 K at a rate of 0.1 kg/s, and exit at 0.2 MPa and 900 K. If heat is lost from the turbine to the surroundings at a rate of 15 kJ/s, the power output of the gas turbine is

(a) 15 kW (b) 30 kW (c) 45 kW (d) 60 kW (e) 75 kW Answer (c) 45 kW

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

Cp_air=1.005 "kJ/kg-C" T1=1500 "K" T2=900 "K" m_dot=0.1 "kg/s" Q_dot_loss=15 "kJ/s" "The rate form of energy balance for a steady-flow system is E_dot_in = E_dot_out" W_dot_out+Q_dot_loss=m_dot*Cp_air*(T1-T2) "Alternative: Variable specific heats - using EES data" W_dot_outvariable+Q_dot_loss=m_dot*(ENTHALPY(Air,T=T1)-ENTHALPY(Air,T=T2))

"Some Wrong Solutions with Common Mistakes:" W1_Wout=m_dot*Cp_air*(T1-T2) "Disregarding heat loss" W2_Wout-Q_dot_loss=m_dot*Cp_air*(T1-T2) "Assuming heat gain instead of loss"

5-210 Steam expands in a turbine from 4 MPa and 500°C to 0.5 MPa and 250°C at a rate of 1350 kg/h. Heat is lost from the turbine at a rate of 25 kJ/s during the process. The power output of the turbine is

(a) 157 kW (b) 207 kW (c) 182 kW (d) 287 kW (e) 246 kW

Answer (a) 157 kW

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

 $\begin{array}{l} T1=\!500 \ "C" \\ P1=\!4000 \ "kPa" \\ T2=\!250 \ "C" \\ P2=\!500 \ "kPa" \\ m_dot=\!1350/3600 \ "kg/s" \\ Q_dot_loss=\!25 \ "kJ/s" \\ h1=\!ENTHALPY(Steam_IAPWS,T=\!T1,P=\!P1) \\ h2=\!ENTHALPY(Steam_IAPWS,T=\!T2,P=\!P2) \\ \ "The rate form of energy balance for a steady-flow system is E_dot_in = E_dot_out" \\ W_dot_out+Q_dot_loss=m_dot^*(h1-h2) \end{array}$

"Some Wrong Solutions with Common Mistakes:" W1_Wout=m_dot*(h1-h2) "Disregarding heat loss" W2_Wout-Q_dot_loss=m_dot*(h1-h2) "Assuming heat gain instead of loss" u1=INTENERGY(Steam_IAPWS,T=T1,P=P1) u2=INTENERGY(Steam_IAPWS,T=T2,P=P2) W3_Wout+Q_dot_loss=m_dot*(u1-u2) "Using internal energy instead of enthalpy" W4_Wout-Q_dot_loss=m_dot*(u1-u2) "Using internal energy and wrong direction for heat"

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5-211 Steam is compressed by an adiabatic compressor from 0.2 MPa and 150°C to 0.8 MPa and 350°C at a rate of 1.30 kg/s. The power input to the compressor is

(a) 511 kW (b) 393 kW (c) 302 kW (d) 717 kW (e) 901 kW Answer (a) 511 kW

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

P1=200 "kPa" T1=150 "C" P2=800 "kPa" T2=350 "C" m_dot=1.30 "kg/s" Q_dot_loss=0 "kJ/s" h1=ENTHALPY(Steam_IAPWS,T=T1,P=P1) h2=ENTHALPY(Steam_IAPWS,T=T2,P=P2) "The rate form of energy balance for a steady-flow system is E_dot_in = E_dot_out" W_dot_in-Q_dot_loss=m_dot*(h2-h1)

"Some Wrong Solutions with Common Mistakes:" W1_Win-Q_dot_loss=(h2-h1)/m_dot "Dividing by mass flow rate instead of multiplying" W2_Win-Q_dot_loss=h2-h1 "Not considering mass flow rate" u1=INTENERGY(Steam_IAPWS,T=T1,P=P1) u2=INTENERGY(Steam_IAPWS,T=T2,P=P2) W3_Win-Q_dot_loss=m_dot*(u2-u1) "Using internal energy instead of enthalpy" W4_Win-Q_dot_loss=u2-u1 "Using internal energy and ignoring mass flow rate" (a) 4.94 kW (b) 6.04 kW (c) 7.14 kW (d) 7.50 kW (e) 8.13 kW

Answer (c) 7.14 kW

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

 $\begin{array}{l} \mathsf{P1=}140 \ \texttt{kPa}\texttt{"} \\ \texttt{x1=}1 \\ \mathsf{P2=}900 \ \texttt{"kPa}\texttt{"} \\ \mathsf{T2=}60 \ \texttt{"C"} \\ \texttt{m_dot=}0.108 \ \texttt{"kg/s"} \\ \mathsf{Q_dot_loss=}1.10 \ \texttt{"kJ/s"} \\ \texttt{h1=}\mathsf{ENTHALPY}(\mathsf{R134a}, \texttt{x=}\texttt{x1}, \mathsf{P=}\mathsf{P1}) \\ \texttt{h2=}\mathsf{ENTHALPY}(\mathsf{R134a}, \texttt{x=}\texttt{r12}, \mathsf{P=}\mathsf{P2}) \\ \texttt{"The rate form of energy balance for a steady-flow system is } \mathsf{E_dot_in} = \mathsf{E_dot_out"} \\ \texttt{W_dot_in-Q_dot_loss=}m_dot^*(\mathsf{h2-h1}) \end{array}$

"Some Wrong Solutions with Common Mistakes:" W1_Win+Q_dot_loss=m_dot*(h2-h1) "Wrong direction for heat transfer" W2_Win =m_dot*(h2-h1) "Not considering heat loss" u1=INTENERGY(R134a,x=x1,P=P1) u2=INTENERGY(R134a,T=T2,P=P2) W3_Win-Q_dot_loss=m_dot*(u2-u1) "Using internal energy instead of enthalpy" W4_Win+Q_dot_loss=u2-u1 "Using internal energy and wrong direction for heat transfer" **5-213** Refrigerant-134a expands in an adiabatic turbine from 1.2 MPa and 100°C to 0.18 MPa and 50°C at a rate of 1.25 kg/s. The power output of the turbine is

(a) 46.3 kW (b) 66.4 kW (c) 72.7 kW (d) 89.2 kW (e) 112.0 kW Answer (a) 46.3 kW

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

 $\begin{array}{l} P1=1200 \ "kPa" \\ T1=100 \ "C" \\ P2=180 \ "kPa" \\ T2=50 \ "C" \\ m_dot=1.25 \ "kg/s" \\ Q_dot_loss=0 \ "kJ/s" \\ h1=ENTHALPY(R134a,T=T1,P=P1) \\ h2=ENTHALPY(R134a,T=T2,P=P2) \\ \ "The rate form of energy balance for a steady-flow system is E_dot_in = E_dot_out" \\ -W_dot_out-Q_dot_loss=m_dot*(h2-h1) \end{array}$

"Some Wrong Solutions with Common Mistakes:" -W1_Wout-Q_dot_loss=(h2-h1)/m_dot "Dividing by mass flow rate instead of multiplying" -W2_Wout-Q_dot_loss=h2-h1 "Not considering mass flow rate" u1=INTENERGY(R134a,T=T1,P=P1) u2=INTENERGY(R134a,T=T2,P=P2) -W3_Wout-Q_dot_loss=m_dot*(u2-u1) "Using internal energy instead of enthalpy" -W4_Wout-Q_dot_loss=u2-u1 "Using internal energy and ignoring mass flow rate"

5-214 Refrigerant-134a at 1.4 MPa and 90°C is throttled to a pressure of 0.6 MPa. The temperature of the refrigerant after throttling is

(a) 22° C (b) 56° C (c) 82° C (d) 80° C (e) 90.0° C

Answer (d) 80°C

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

P1=1400 "kPa" T1=90 "C" P2=600 "kPa" h1=ENTHALPY(R134a,T=T1,P=P1) T2=TEMPERATURE(R134a,h=h1,P=P2)

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"Some Wrong Solutions with Common Mistakes:"
W1_T2=T1 "Assuming the temperature to remain constant"
W2_T2=TEMPERATURE(R134a,x=0,P=P2) "Taking the temperature to be the saturation temperature at P2"
u1=INTENERGY(R134a,T=T1,P=P1)
W3_T2=TEMPERATURE(R134a,u=u1,P=P2) "Assuming u=constant"
v1=VOLUME(R134a,T=T1,P=P1)
W4_T2=TEMPERATURE(R134a,v=v1,P=P2) "Assuming v=constant"
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5-215 Air at 27°C and 5 atm is throttled by a valve to 1 atm. If the valve is adiabatic and the change in kinetic energy is negligible, the exit temperature of air will be

(a) 10° C (b) 15° C (c) 20° C (d) 23° C (e) 27° C Answer (e) 27° C

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

"The temperature of an ideal gas remains constant during throttling, and thus T2=T1" T1=27 "C" P1=5 "atm" P2=1 "atm" T2=T1 "C"

"Some Wrong Solutions with Common Mistakes:" W1_T2=T1*P1/P2 "Assuming v=constant and using C" W2_T2=(T1+273)*P1/P2-273 "Assuming v=constant and using K" W3_T2=T1*P2/P1 "Assuming v=constant and pressures backwards and using C" W4_T2=(T1+273)*P2/P1 "Assuming v=constant and pressures backwards and using K"

5-216 Steam at 1 MPa and 300°C is throttled adiabatically to a pressure of 0.4 MPa. If the change in kinetic energy is negligible, the specific volume of the steam after throttling will be

(a) $0.358 \text{ m}^3/\text{kg}$ (b) $0.233 \text{ m}^3/\text{kg}$ (c) $0.375 \text{ m}^3/\text{kg}$ (d) $0.646 \text{ m}^3/\text{kg}$ (e) $0.655 \text{ m}^3/\text{kg}$ Answer (d) $0.646 \text{ m}^3/\text{kg}$

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

P1=1000 "kPa" T1=300 "C" P2=400 "kPa" h1=ENTHALPY(Steam_IAPWS,T=T1,P=P1) v2=VOLUME(Steam_IAPWS,h=h1,P=P2)

"Some Wrong Solutions with Common Mistakes:" W1_v2=VOLUME(Steam_IAPWS,T=T1,P=P2) "Assuming the volume to remain constant" u1=INTENERGY(Steam,T=T1,P=P1) W2_v2=VOLUME(Steam_IAPWS,u=u1,P=P2) "Assuming u=constant" W3_v2=VOLUME(Steam_IAPWS,T=T1,P=P2) "Assuming T=constant" **5-217** Air is to be heated steadily by an 8-kW electric resistance heater as it flows through an insulated duct. If the air enters at 50° C at a rate of 2 kg/s, the exit temperature of air will be

(a) 46.0° C (b) 50.0° C (c) 54.0° C (d) 55.4° C (e) 58.0° C Answer (c) 54.0° C

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

Cp=1.005 "kJ/kg-C" T1=50 "C" m_dot=2 "kg/s" W_dot_e=8 "kJ/s" W_dot_e=m_dot*Cp*(T2-T1) "Checking using data from EES table" W_dot_e=m_dot*(ENTHALPY(Air,T=T_2table)-ENTHALPY(Air,T=T1))

"Some Wrong Solutions with Common Mistakes:" Cv=0.718 "kJ/kg.K" W_dot_e=Cp*(W1_T2-T1) "Not using mass flow rate" W_dot_e=m_dot*Cv*(W2_T2-T1) "Using Cv" W_dot_e=m_dot*Cp*W3_T2 "Ignoring T1"

5-218 Saturated water vapor at 40°C is to be condensed as it flows through a tube at a rate of 0.20 kg/s. The condensate leaves the tube as a saturated liquid at 40°C. The rate of heat transfer from the tube is

(a) 34 kJ/s (b) 481 kJ/s (c) 2406 kJ/s (d) 514 kJ/s (e) 548 kJ/s Answer (b) 481 kJ/s

Solution Solved by EES Software. Solutions can be verified by copying-and-pasting the following lines on a blank EES

screen. (Similar problems and their solutions can be obtained easily by modifying numerical values).

T1=40 "C" m_dot=0.20 "kg/s" h_f=ENTHALPY(Steam_IAPWS,T=T1,x=0) h_g=ENTHALPY(Steam_IAPWS,T=T1,x=1) h_fg=h_g-h_f Q_dot=m_dot*h_fg

"Some Wrong Solutions with Common Mistakes:" W1_Q=m_dot*h_f "Using hf" W2_Q=m_dot*h_g "Using hg" W3_Q=h_fg "not using mass flow rate" W4_Q=m_dot*(h_f+h_g) "Adding hf and hg"

5-219 ... 5-223 Design and Essay Problems

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